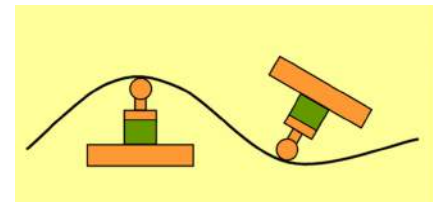
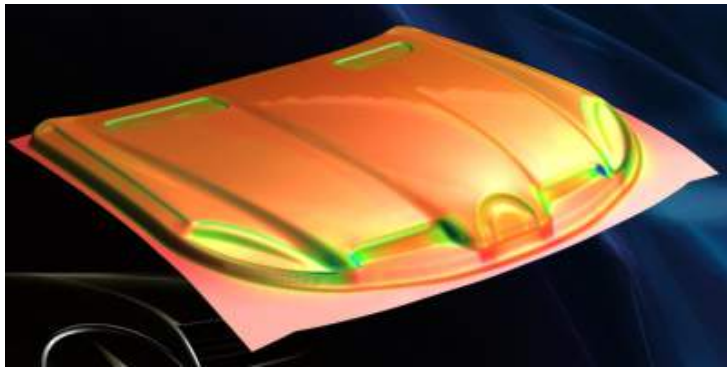
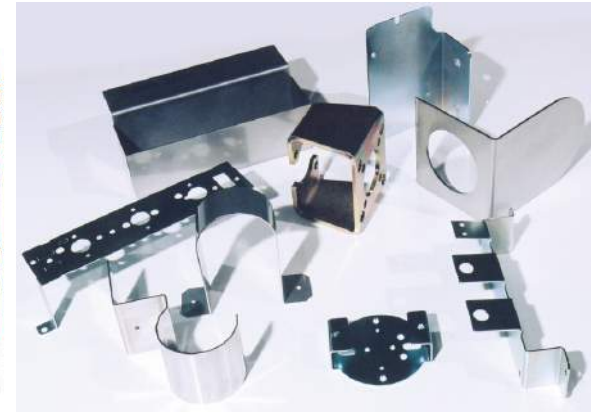


# *Sheet Metal Forming*

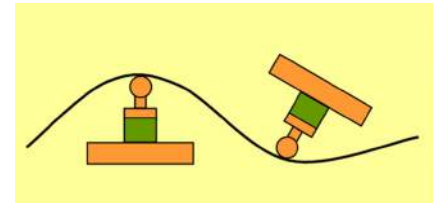
- ◆ “Sheet Metal Forming” Ch. 16 Kalpakjian
- ◆ “Design for Sheetmetal Working”,  
Ch. 9 Boothroyd, Dewhurst and Knight



# Examples-sheet metal formed



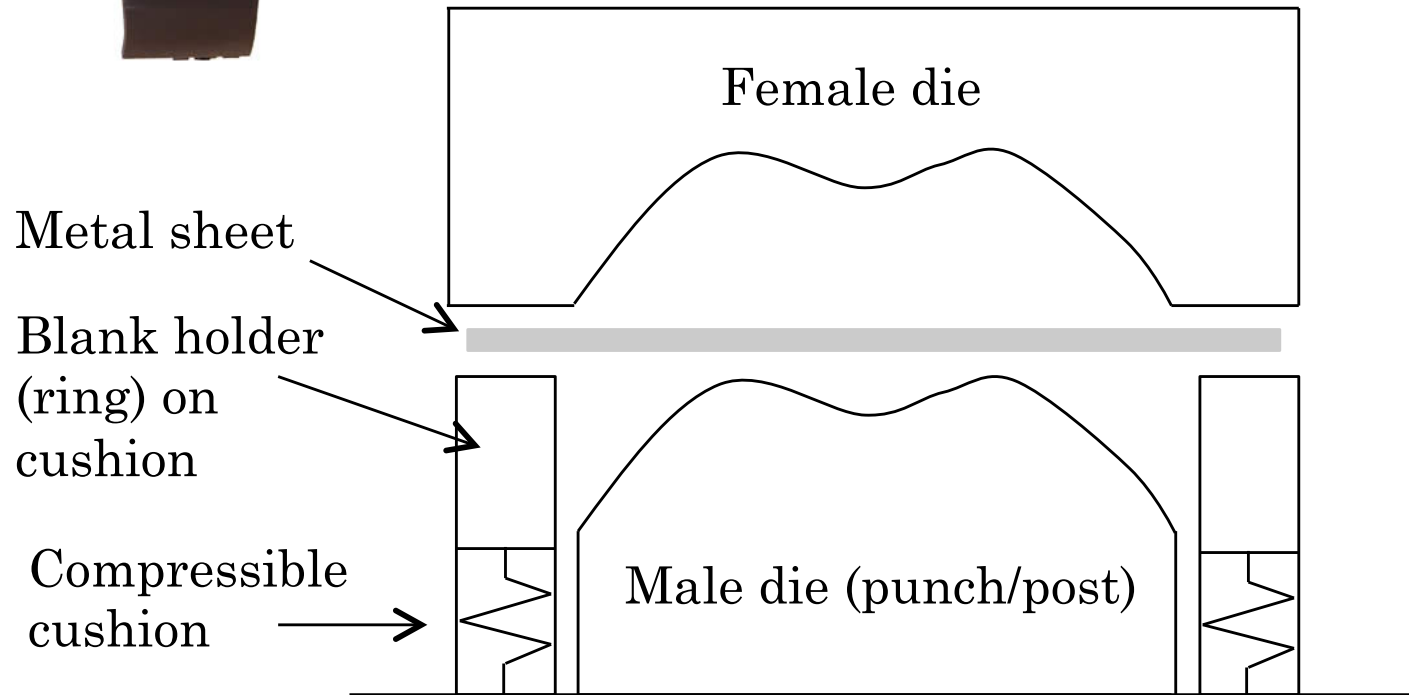
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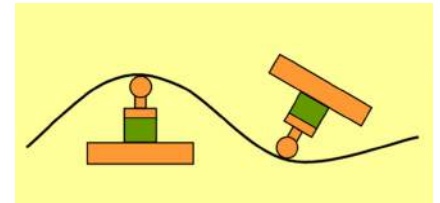
# Sheet metal stamping/drawing – car industry



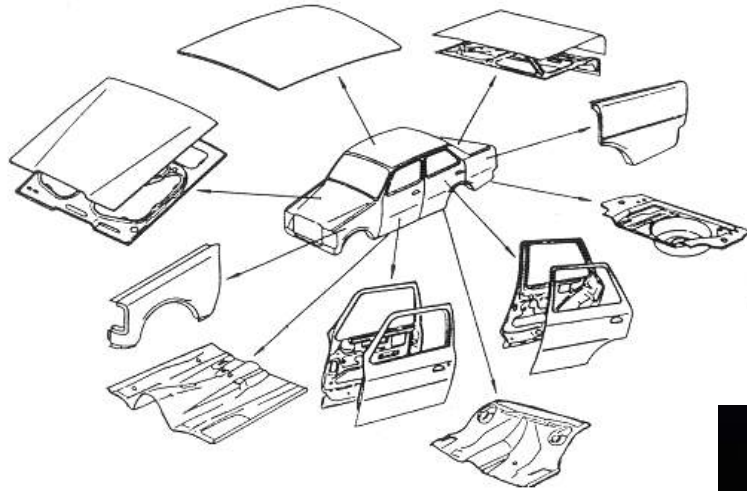
• **90million** cars and commercial vehicles produced worldwide in 2014



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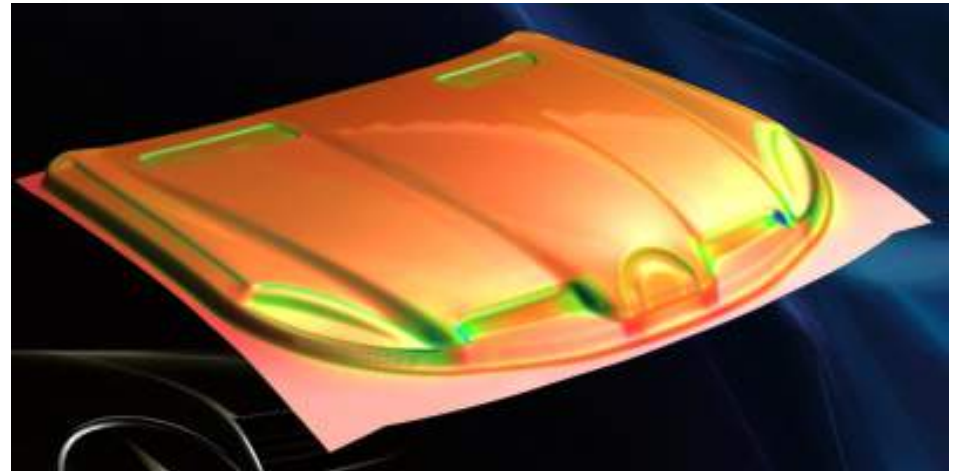


# Stamping Auto body panels

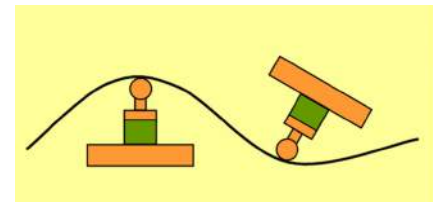


- 3 to 5 dies each
- Prototype dies ~ \$50,000
- Production dies ~ \$0.75-1

- Forming dies
- Trimming station
- Flanging station



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# Objectives

By the end of today you should be able to...

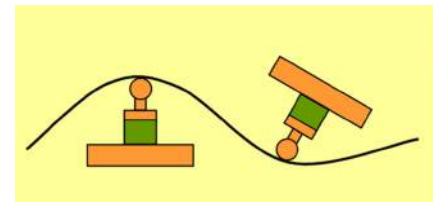
...**describe** different forming processes, when they might be used, and **compare** their production rates, costs and environmental impacts

...**calculate** forming forces, **predict** part defects (tearing, wrinkling, dimensional inaccuracy), and **propose** solutions

...**explain** current developments: opportunities and challenges



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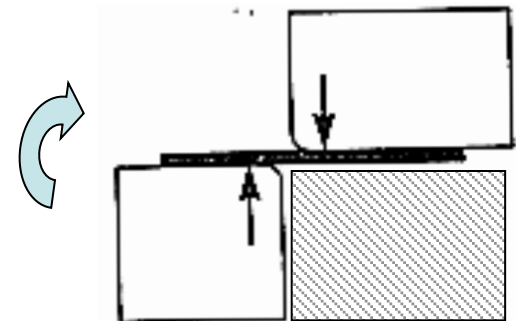
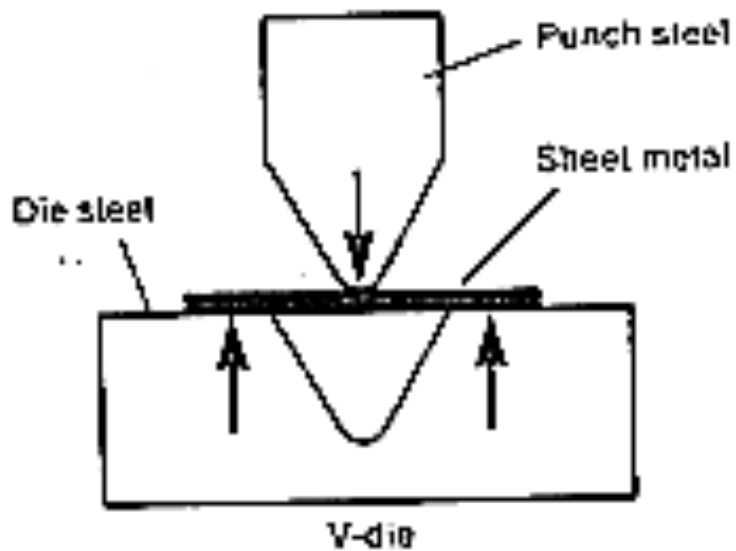


# *LMP Shop*

## Brake press



## Finger brake



# Technology – a brief review

Forming  
Speed

## Material drawn into shape

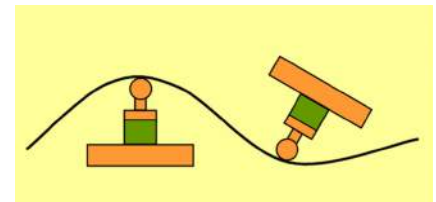
- Conventional drawing/stamping – expensive tooling, no net thinning, quick **20-1000pts/hr**
- Hydro-forming – cheap tooling, no net thinning, slow, high formability **7-13cycles/hr**

## Material stretched into shape

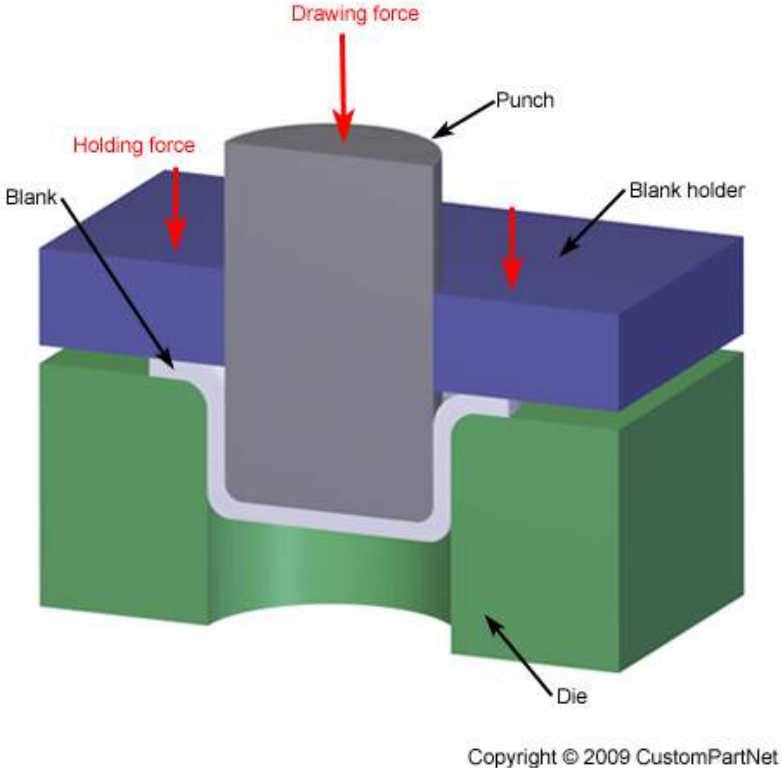
- Stretch forming – very cheap tooling, net thinning, slow, low formability **3-8pts/hr**
- Super-plastic forming – cheap tooling, net thinning, expensive sheet metal, slow, very high formability **0.3-4pts/hr**



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# Drawing – expensive tooling, no net thinning, quick



Deep-drawing

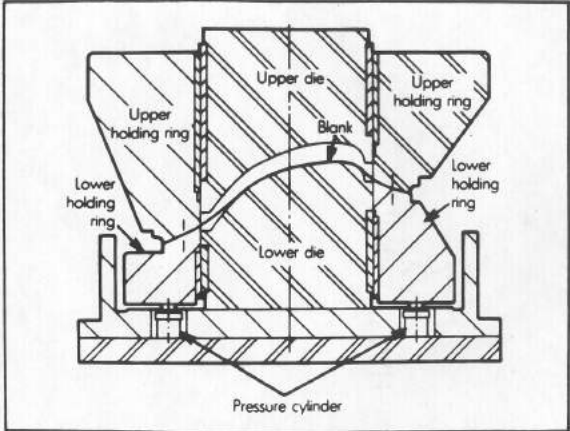
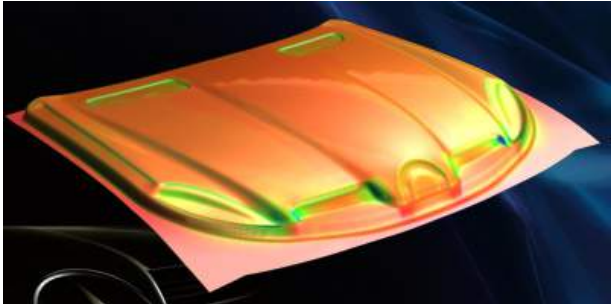


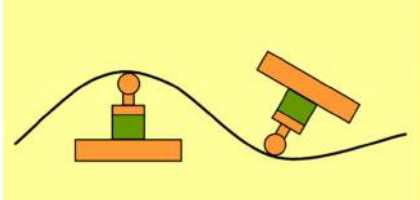
Fig. 7-23 Tooling for stretch-draw forming fenders from steel blanks. (Oldsmobile Div., General Motors Corp.)



Shallow-drawing (stamping)

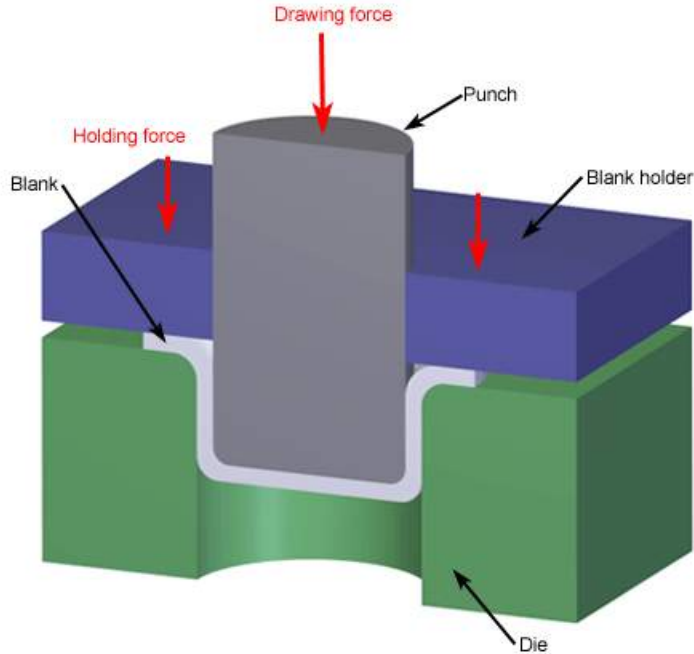


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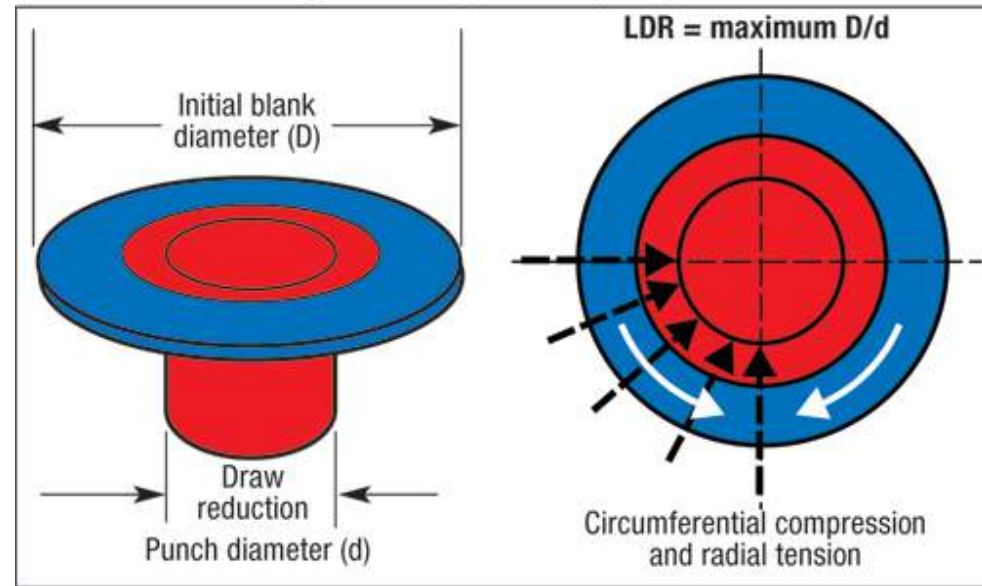


# Deep-drawing



Copyright © 2009 CustomPartNet

## Limiting Drawing Ratio (LDR) Defined

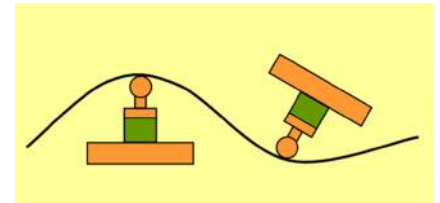


Blank holder helps prevent wrinkling and reduces springback

Blank holder not necessary if blank diameter / blank thickness is less than 25-40. Smaller values for deeper forming.



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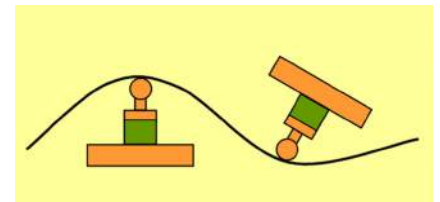




<http://www.thomasnet.com/articles/custom-manufacturing-fabricating/wrinkling-during-deep-drawing>

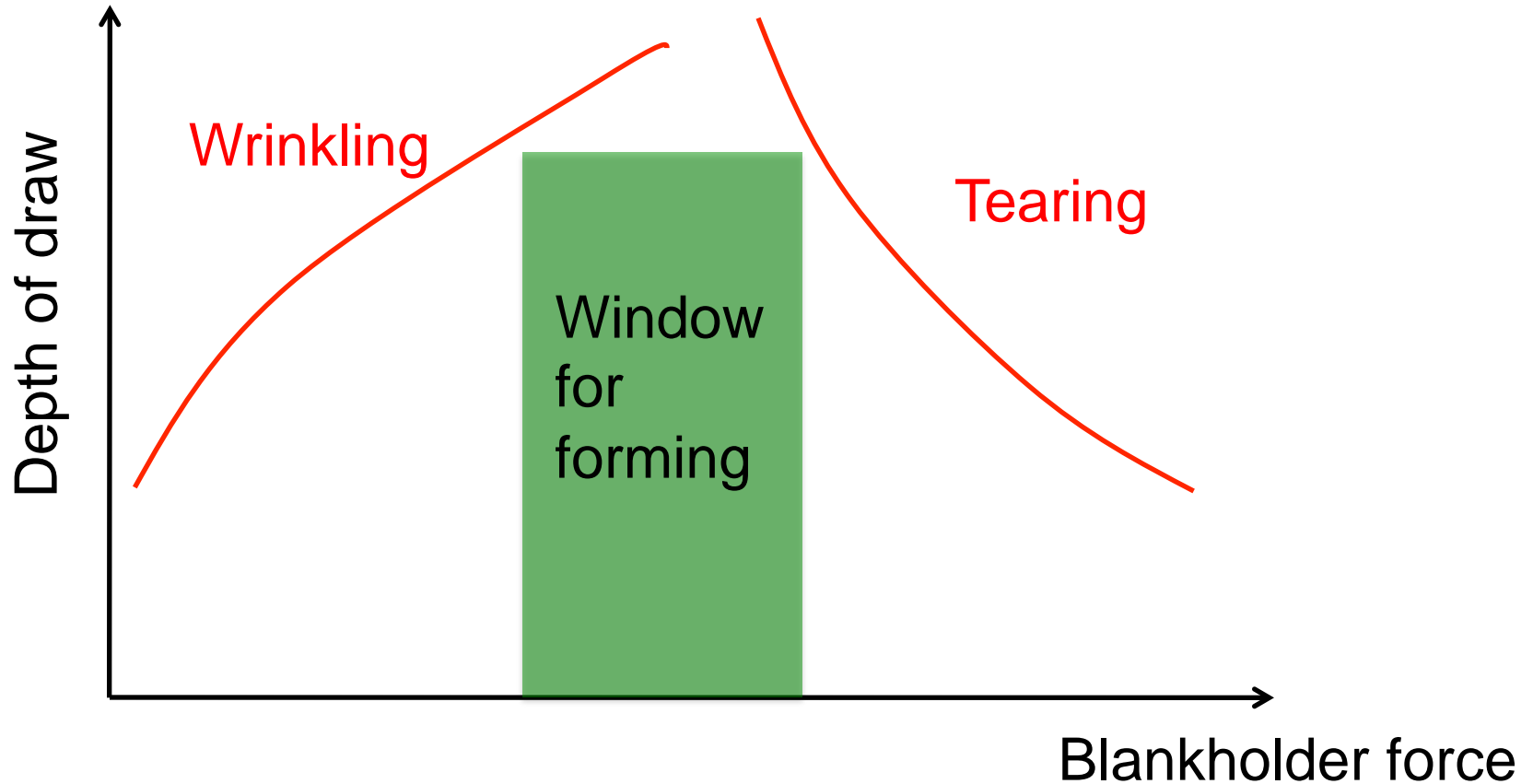


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# Blank holder force: forming window

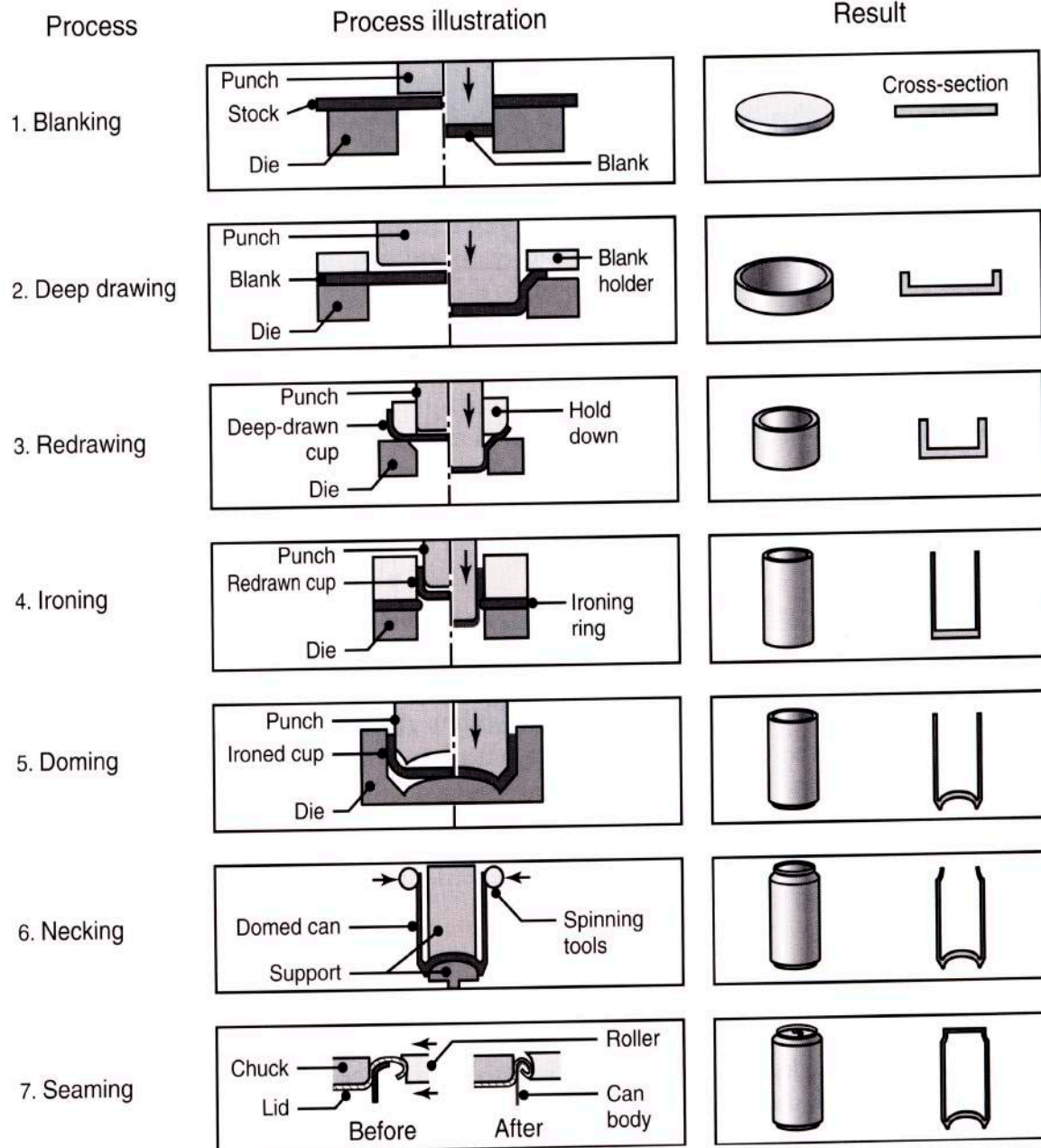
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# Deep Drawing of drinks cans



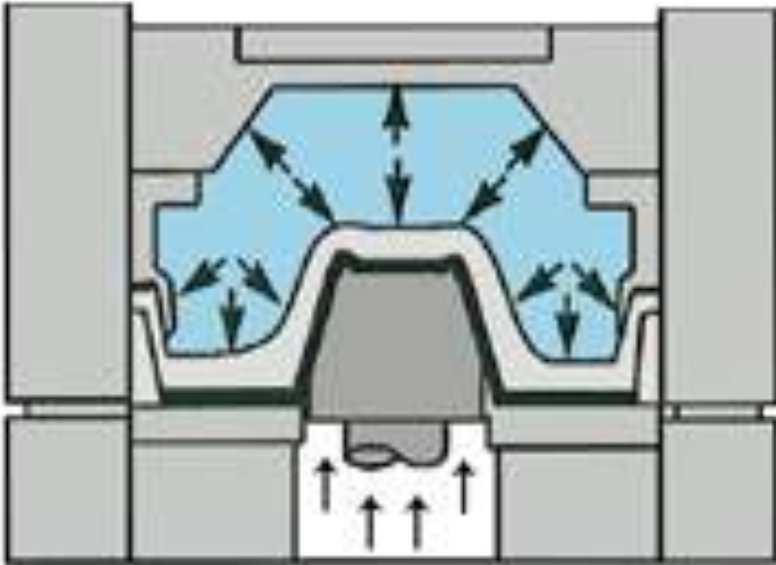
Hosford and Duncan  
(can making): <http://www.chymist.com/Aluminum%20can.pdf>



**FIGURE 16.31** The metal-forming processes involved in manufacturing a two-piece aluminum beverage can.

The metal-forming processes involved in manufacturing a two-piece aluminum beverage can.

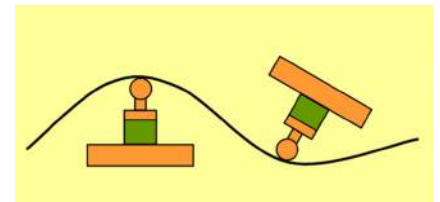
# Hydro-forming – cheap tooling, no net thinning, slow(ish), high formability



Low volume batches



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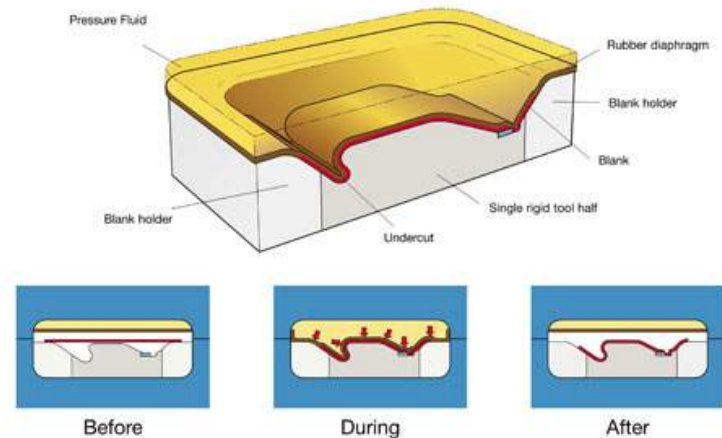




# Hydro-forming – cheap tooling, no net thinning, slow(ish), high formability



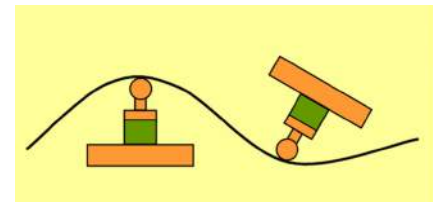
## Flexform – Principle



Low volume batches



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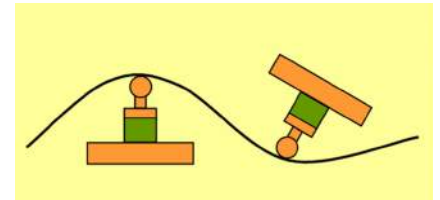
# Hydro-forming – cheap tooling, no net thinning, slow, high formability



Small flexforming tool made by additive manufacturing

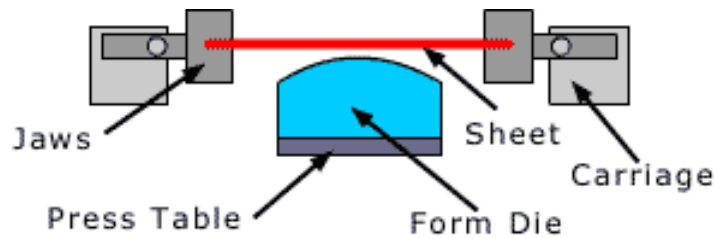


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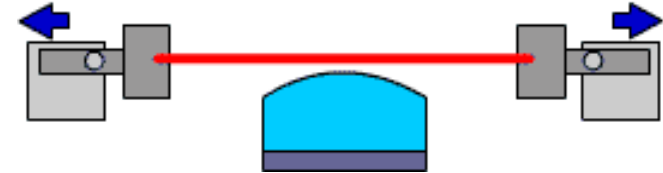


# Stretch forming

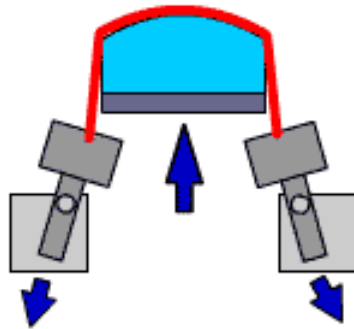
– very cheap tooling, net thinning, slow, low formability, sheet metal up to 15mx9m



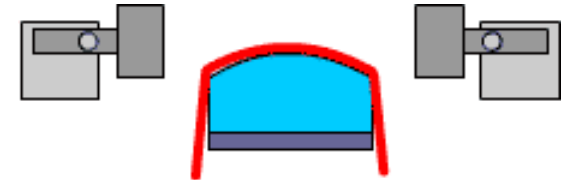
Loading



Pre-stretching



Wrapping

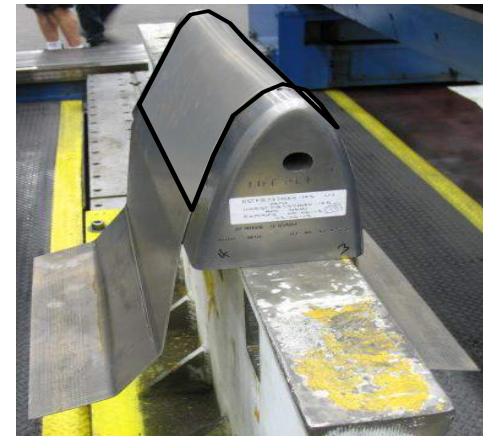


Release

\* source: [http://www.cyrilbath.com/sheet\\_process.html](http://www.cyrilbath.com/sheet_process.html)

## Low volume batches

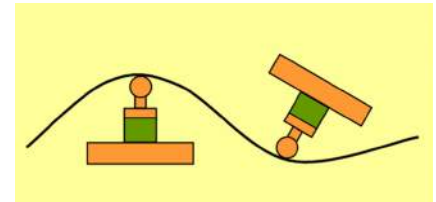
# Stretch forming: Example parts



**Higher aspect ratio, deeper parts**

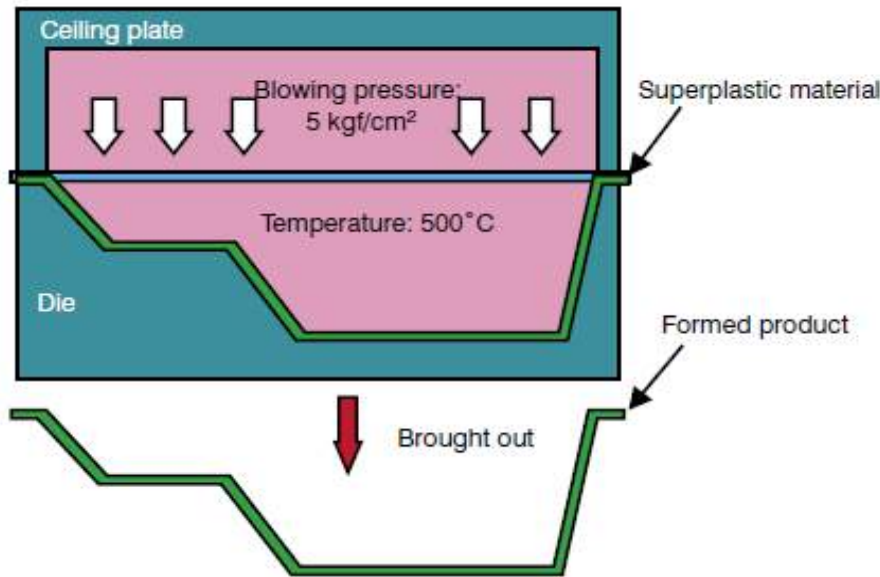
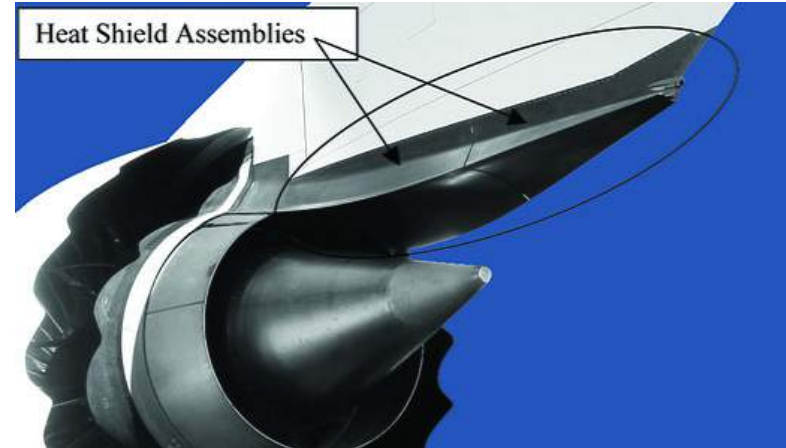
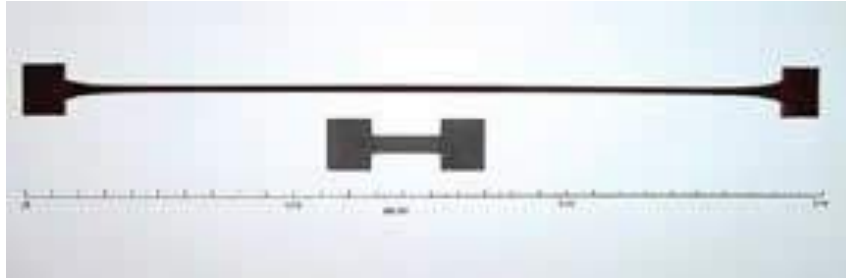


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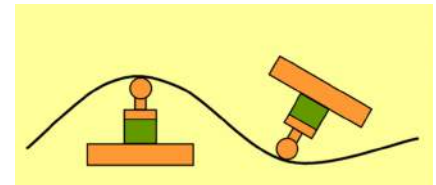
# Super-plastic forming – cheap tooling, net thinning, slow, expensive sheet metal, very high formability



Low volume batches, 0.5-0.75 melting temp



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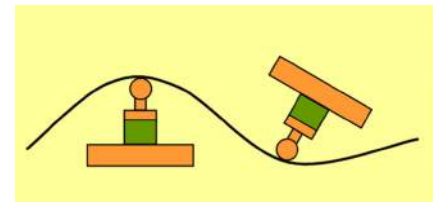




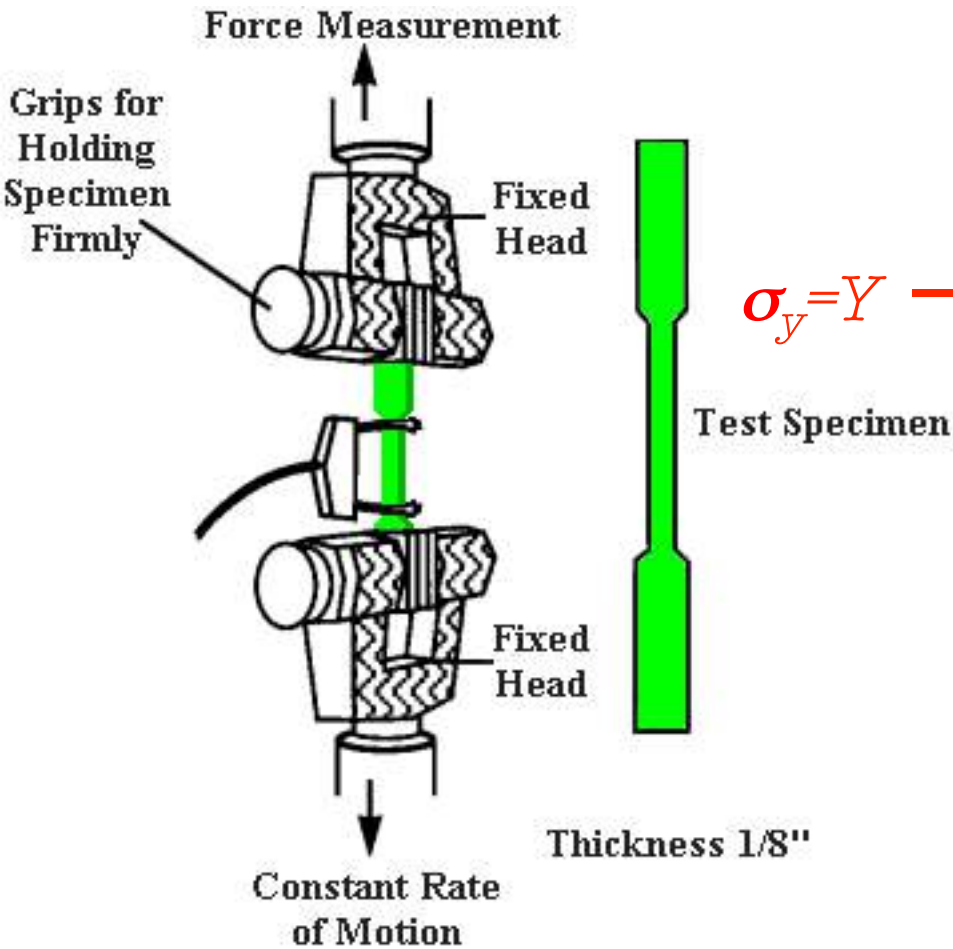
# Forming forces and part geometry



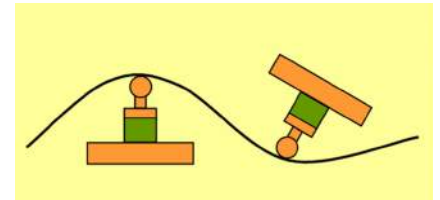
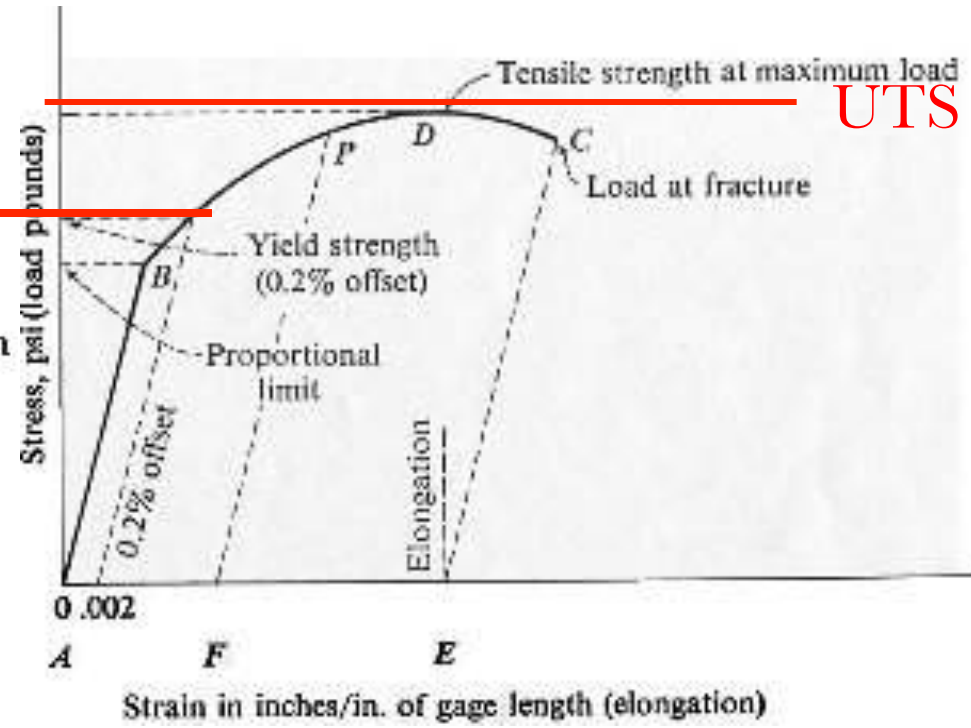
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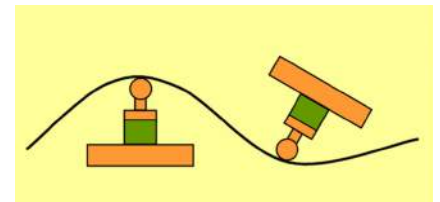
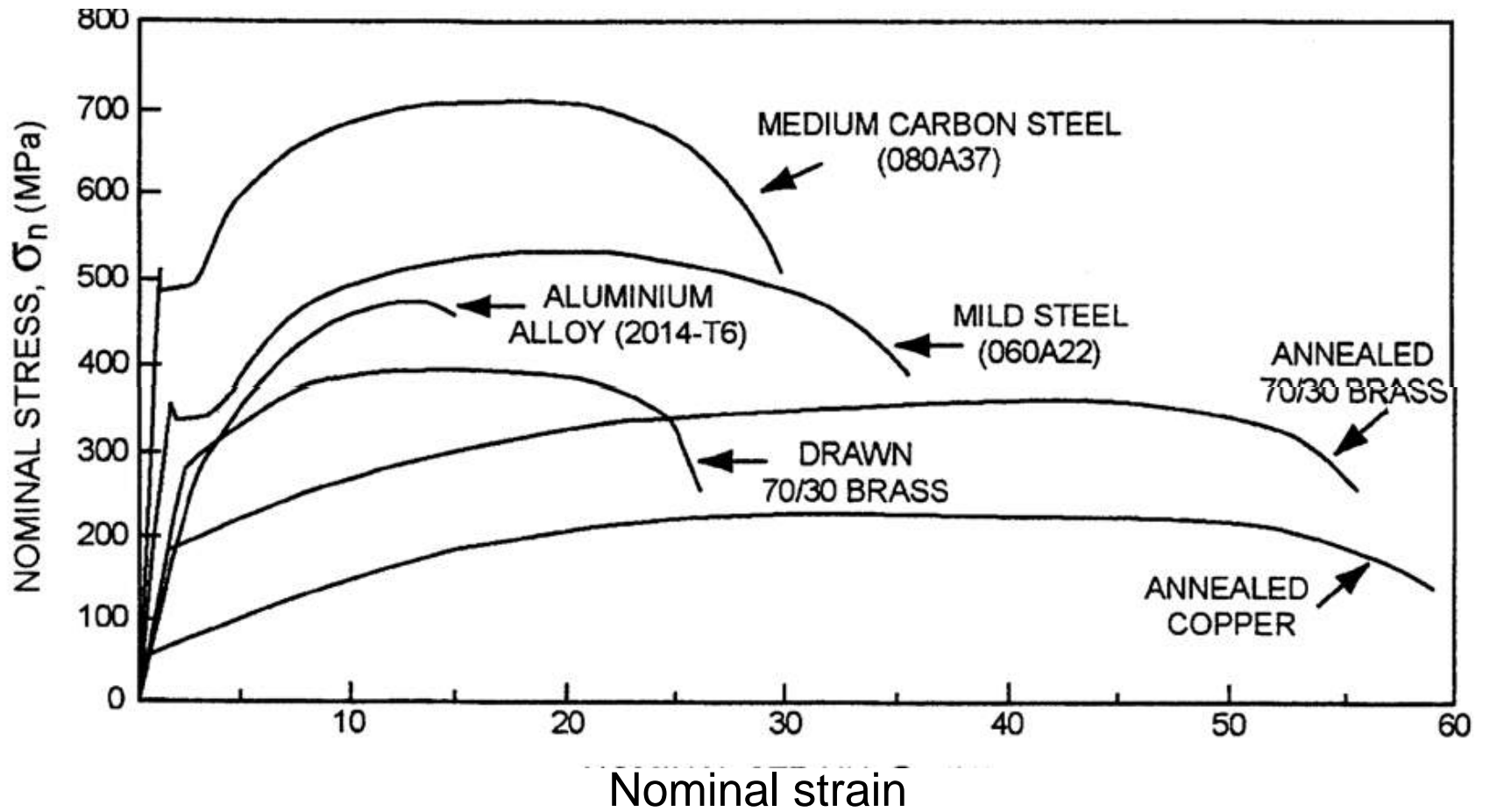


# Tensile test – the Stress-strain diagram



$$\sigma_y = Y$$





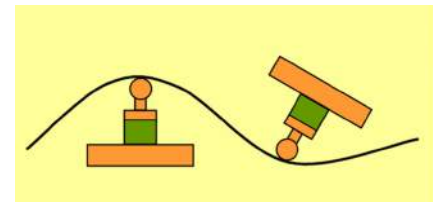
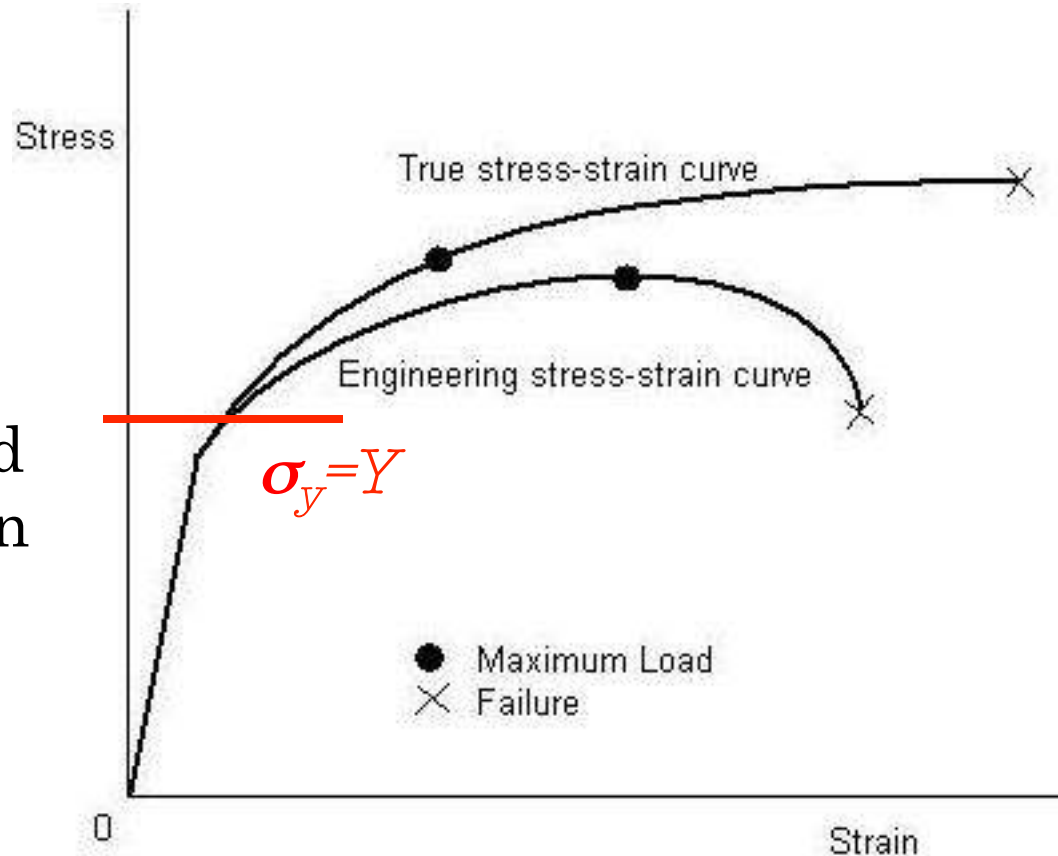
# True stress & strain

$$\epsilon_{tr} = \ln(1 + \epsilon_{en})$$

$$\sigma_{tr} = \sigma_{en} (1 + \epsilon_{en})$$

True stress can be expressed using a power law (Hollomon equation):

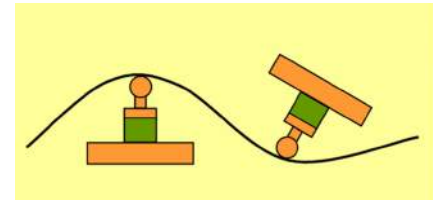
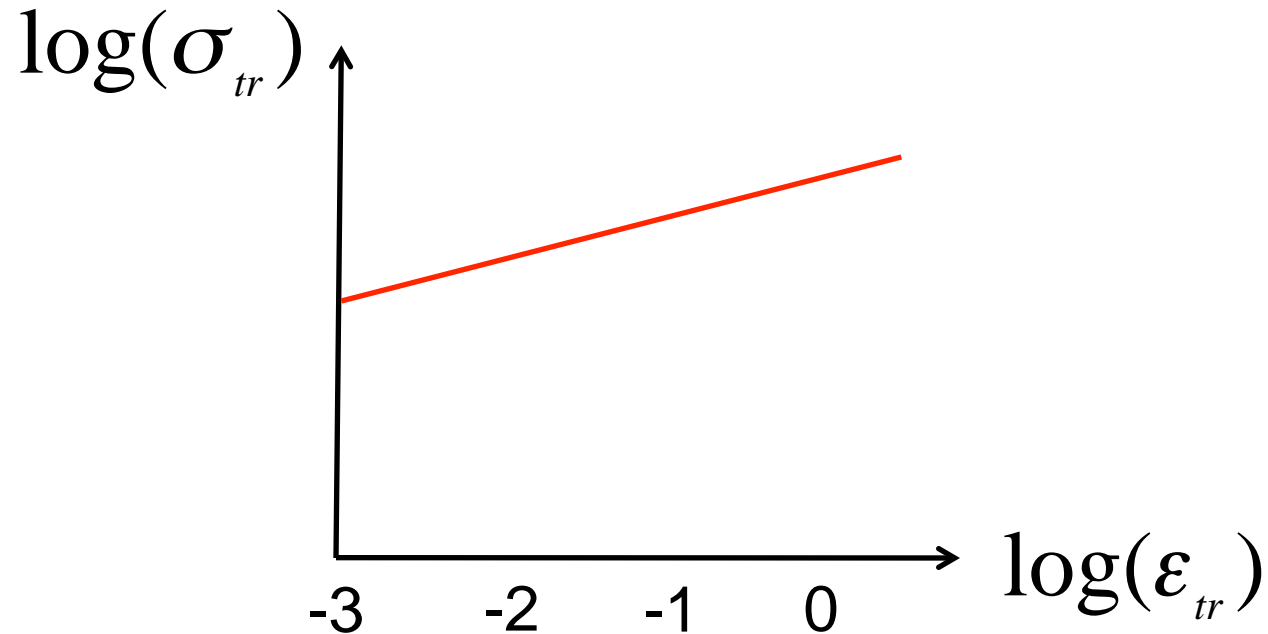
$$\sigma_{tr} = K \epsilon_{tr}^n$$



# Power-Law Expression (Hollomon equation)

$$\sigma_{tr} = K \varepsilon_{tr}^n$$

Can be re-written:  $\log(\sigma_{tr}) = n \log(\varepsilon_{tr}) + \log K$

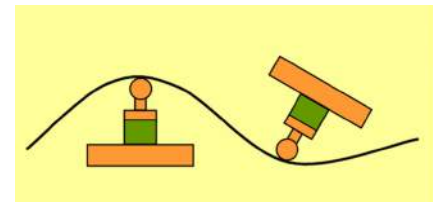
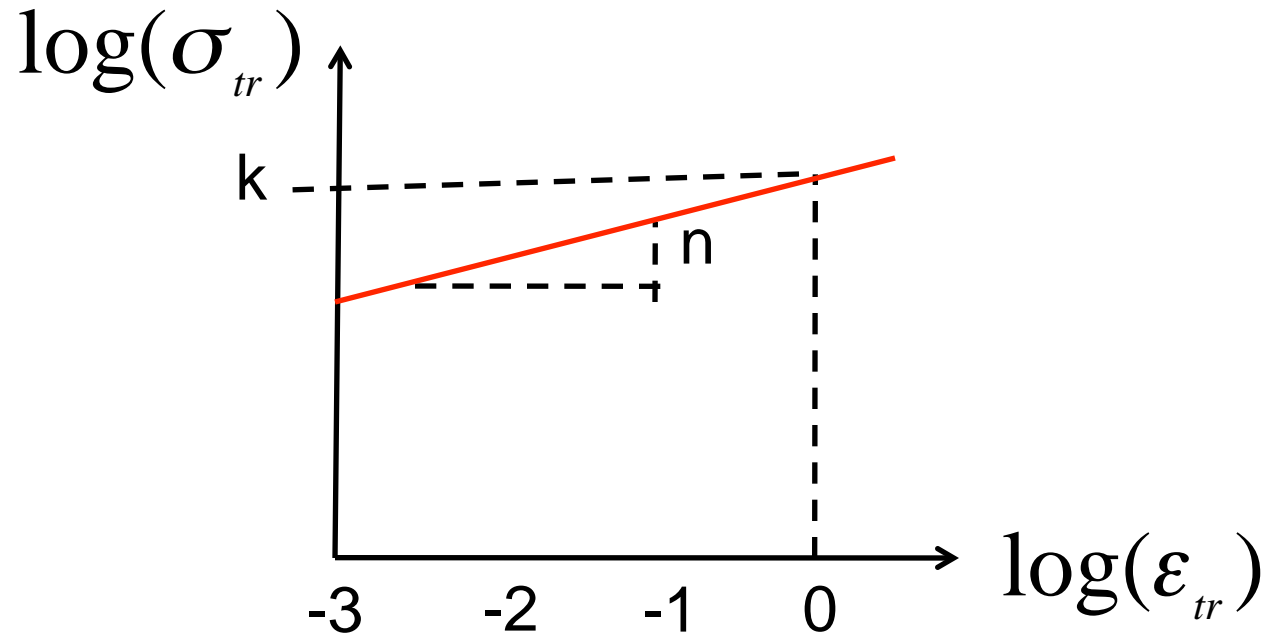




# Power-Law Expression (Hollomon equation)

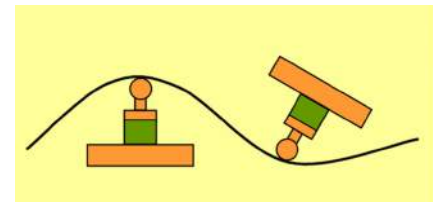
$$\sigma_{tr} = K \varepsilon_{tr}^n$$

Can be re-written:  $\log(\sigma_{tr}) = n \log(\varepsilon_{tr}) + \log K$

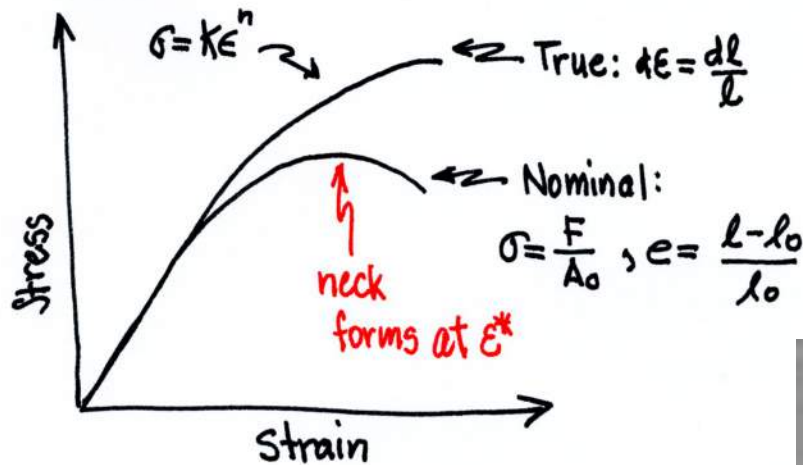


**TABLE 2.3****Typical Values for  $K$  and  $n$  for Selected Metals**

Material	$K$ (MPa)	$n$
Aluminum		
1100-O	180	0.20
2024-T4	690	0.16
5052-O	202	0.13
6061-O	205	0.20
6061-T6	410	0.05
7075-O	400	0.17
Brass		
70-30, annealed	900	0.49
85-15, cold rolled	580	0.34
Cobalt-based alloy, heat treated	2070	0.50
Copper, annealed	315	0.54
Steel		
Low-C, annealed	530	0.26
1020, annealed	745	0.20
4135, annealed	1015	0.17
4135, cold rolled	1100	0.14
4340, annealed	640	0.15
304 stainless, annealed	1275	0.45
410 stainless, annealed	960	0.10
Titanium		
Ti-6Al-4V, annealed, 20°C	1400	0.015
Ti-6Al-4V, annealed, 200°C	1040	0.026
Ti-6Al-4V, annealed, 600°C	650	0.064
Ti-6Al-4V, annealed, 800°C	350	0.146



# Tensile instability - necking



Tensile instability (1-D)

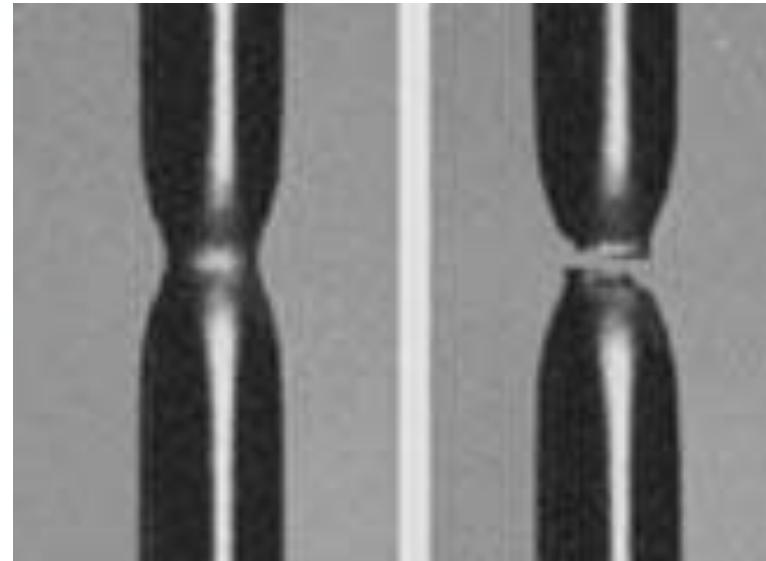
$$F = \sigma A; \text{ so } dF = \sigma dA + A d\sigma = 0 \text{ at max load}$$

$$\frac{d\sigma}{\sigma} = -\frac{dA}{A} = d\epsilon$$

$$\frac{d\sigma}{d\epsilon} = \sigma$$

$$\text{With } \sigma = K\epsilon^n: \quad \frac{d\sigma}{d\epsilon} = nK\epsilon^{n-1} = \sigma = K\epsilon^n$$

$$\Rightarrow \boxed{\epsilon^* = n}$$

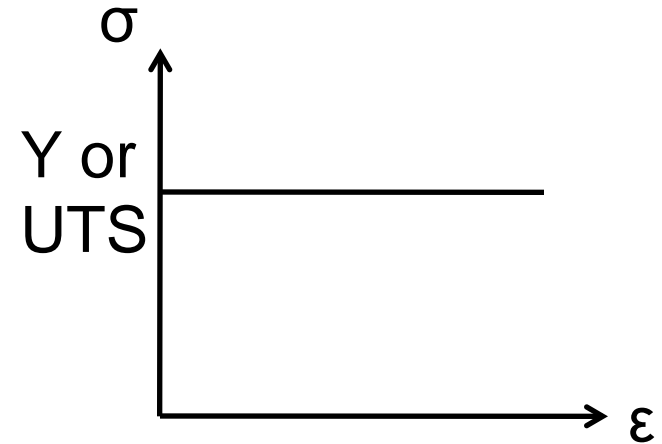


# Useful assumptions

Only interested in plastic effects:

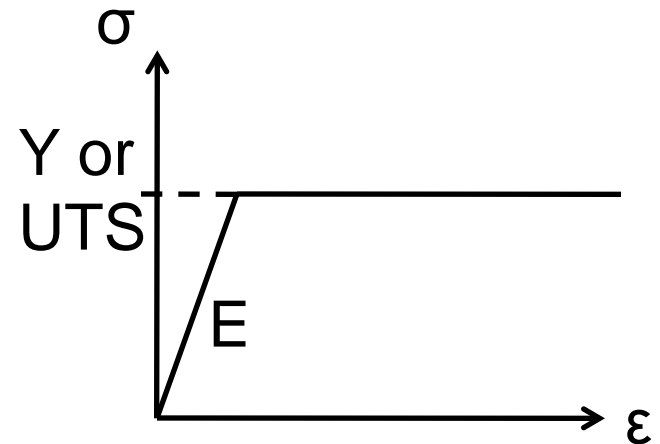
## **Perfectly plastic material**

At  $Y$ , material deforms (‘flows’) in compression and fails in tension

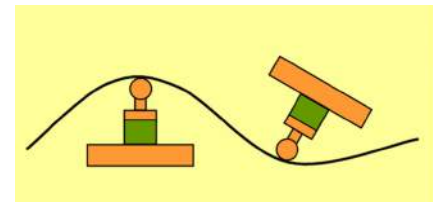


Interested in elastic and plastic effects:

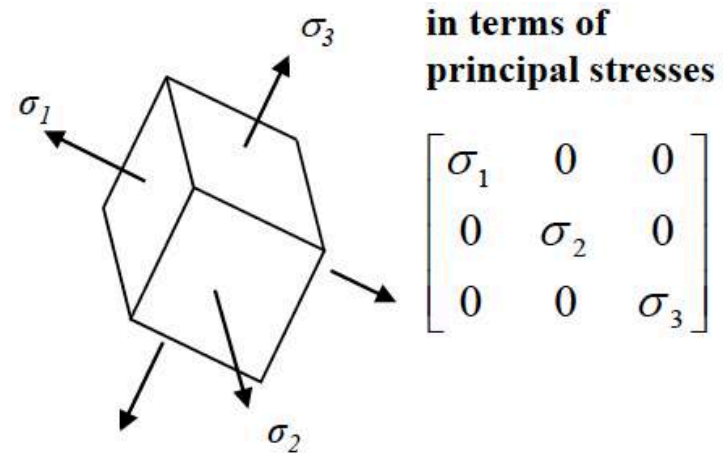
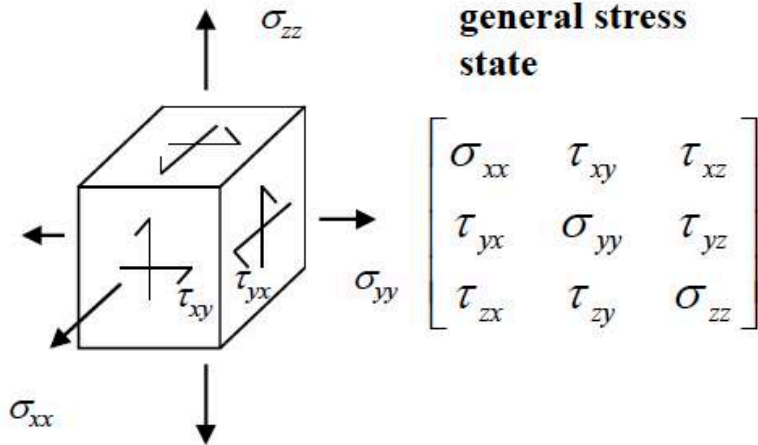
## **Elastic-perfectly plastic material**



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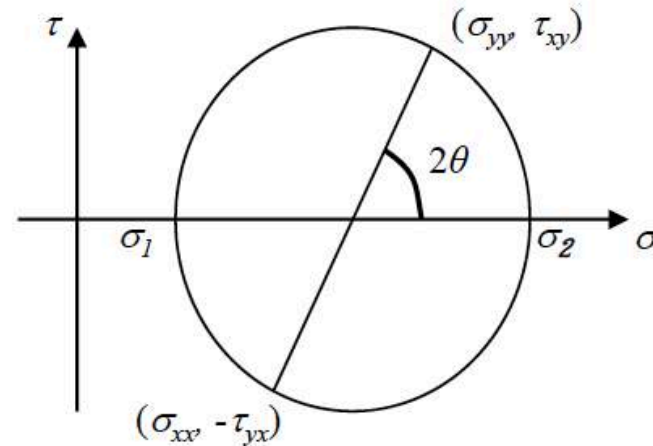


# 3D Problems



For any general stress state we can find a set of *principal axes*. The stress tensor for these axes contains no off-diagonal (shear) terms – only three principal stresses along the three axes.

Mohr's circle allows rotation of axes in two dimensions about one principal axis



In 1-D,  $\sigma = K\varepsilon^n$  assuming perfectly plastic, yielding at:  $\sigma = Y$

In 3-D,  $\sigma_{eff} = K\varepsilon_{eff}^n$  assuming perfectly plastic, yielding at:

$$\sigma_{eff} = Y$$



# 3D Yield Criteria

**Tresca:** Yielding occurs at a maximum shear stress

Effective stress (in principal directions):

$$\sigma_{eff} = \max_{i \neq j} [\sigma_i - \sigma_j]$$

Yield criterion:

$$\sigma_{eff} = Y$$

$$\tau_{max} = k = \frac{Y}{2}$$

Effective strain:

$$\epsilon_{eff} = (\epsilon_i)_{max}$$

**Von Mises:** Yielding at maximum distortion strain energy

Effective stress (in principal directions):

$$\sigma_{eff} = \sqrt{\frac{1}{2} \times \left[ (\sigma_2 - \sigma_3)^2 + (\sigma_3 - \sigma_1)^2 + (\sigma_1 - \sigma_2)^2 \right]}$$

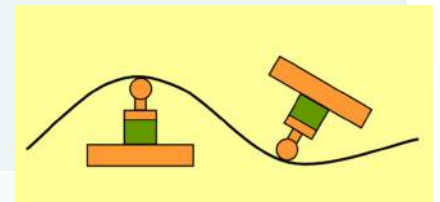
Yield criterion:

$$\sigma_{eff} = Y$$

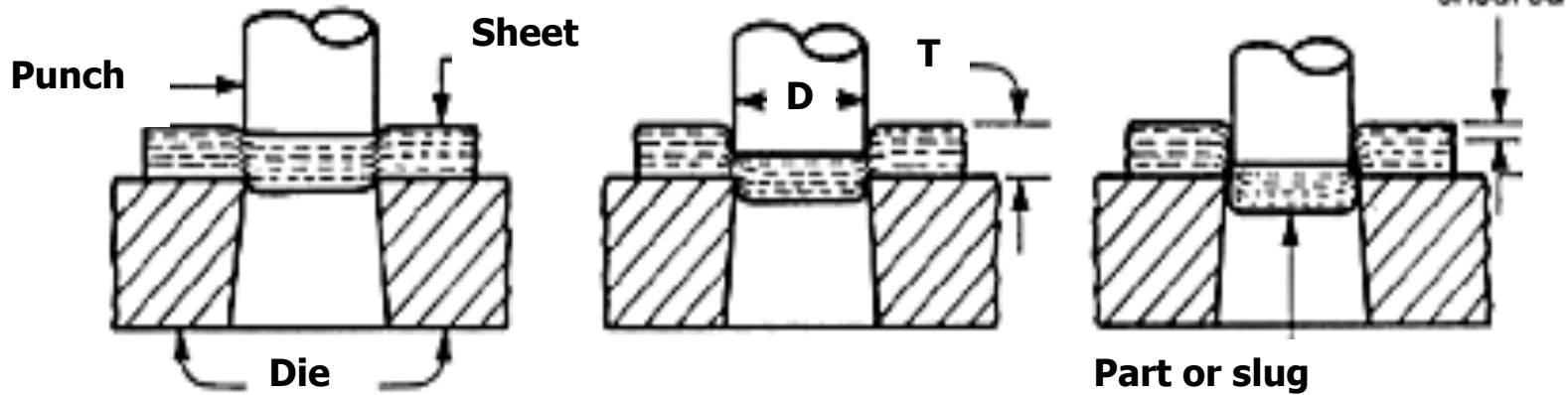
$$Y = \sqrt{3}k$$

Effective strain:

$$\epsilon_{eff} = \sqrt{\frac{2}{3} (\epsilon_1^2 + \epsilon_2^2 + \epsilon_3^2)}$$



# Shearing



$$F = 0.7 T L (\text{UTS})$$

T = Sheet Thickness

L = Total length Sheared

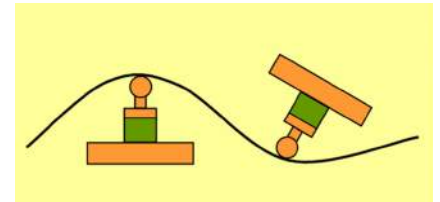
UTS = Ultimate Tensile Strength  
of material



*Shear press - LMP Shop*



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# Side Note: For a general state of stress use “effective stress”

## 2-6 EFFECTIVE STRESS

With either yield criterion, it is useful to define an effective stress denoted as  $\bar{\sigma}$  which is a function of the applied stresses. If the *magnitude* of  $\bar{\sigma}$  reaches a critical value, then the applied stress state will cause yielding; in essence, it has reached an effective level. For the von Mises criterion,

$$\bar{\sigma} = \frac{1}{\sqrt{2}} [(\sigma_1 - \sigma_2)^2 + (\sigma_2 - \sigma_3)^2 + (\sigma_3 - \sigma_1)^2]^{1/2} \quad (2-16)$$

while for the Tresca criterion,

$$\bar{\sigma} = \sigma_1 - \sigma_3 \quad \text{where} \quad \sigma_1 > \sigma_2 > \sigma_3 \quad (2-17)$$

Yielding occurs when  $\sigma_{\text{effective}} = Y$

# Origin of effective strain

## 2-7 EFFECTIVE STRAIN

Effective strain is *defined* such that the incremental work per unit volume is

$$dw = \bar{\sigma} d\bar{\epsilon} = \sigma_1 d\epsilon_1 + \sigma_2 d\epsilon_2 + \sigma_3 d\epsilon_3 \quad (2-18)$$

For the von Mises criterion, the effective strain is given by

$$d\bar{\epsilon} = \frac{\sqrt{2}}{3} [(d\epsilon_1 - d\epsilon_2)^2 + (d\epsilon_2 - d\epsilon_3)^2 + (d\epsilon_3 - d\epsilon_1)^2]^{1/2} \quad (2-19)$$

which may be expressed in a simpler form as

$$d\bar{\epsilon} = \left[ \frac{2}{3} (d\epsilon_1^2 + d\epsilon_2^2 + d\epsilon_3^2) \right]^{1/2} \quad (2-20)$$

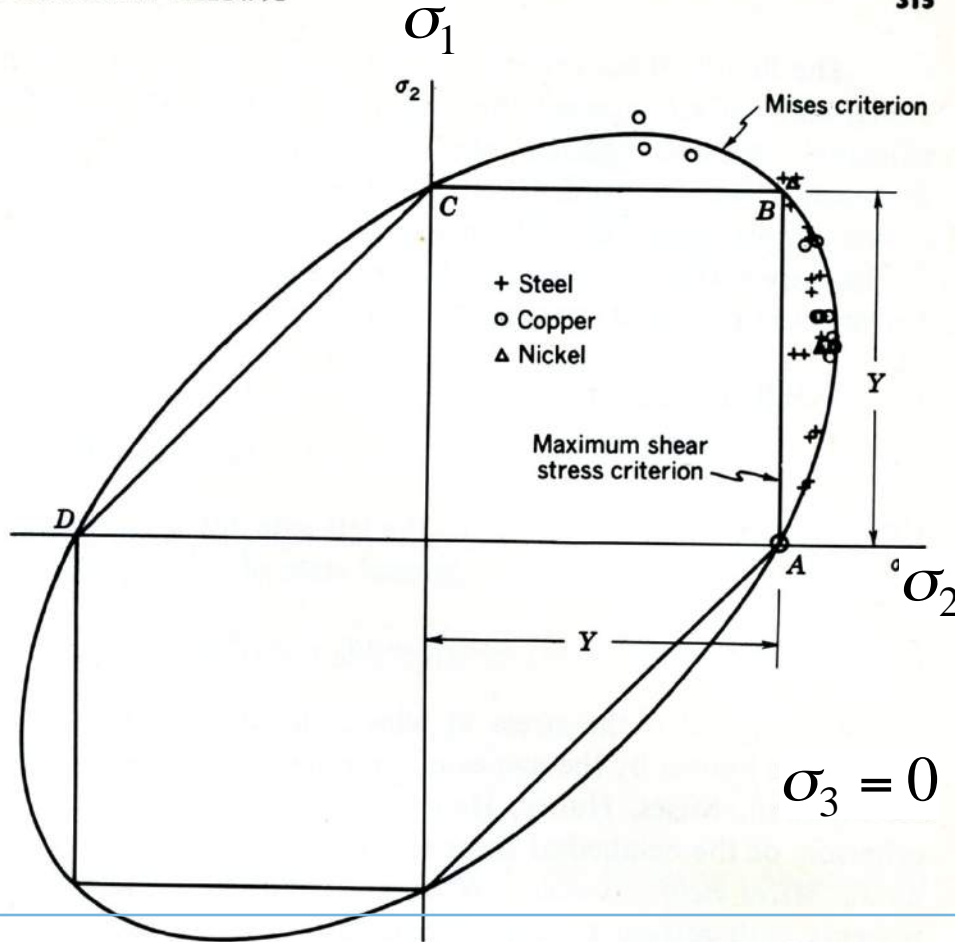
If the straining is proportional (with a constant ratio of  $d\epsilon_1 : d\epsilon_2 : d\epsilon_3$ ), the total effective strain may be expressed in terms of the total strains as

$$\bar{\epsilon} = \left[ \frac{2}{3} (\epsilon_1^2 + \epsilon_2^2 + \epsilon_3^2) \right]^{1/2} \quad (2-21)$$

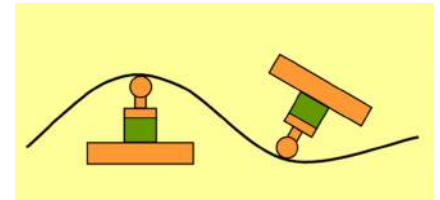
If the strain path is not constant,  $\bar{\epsilon}$  must be found from a path integral of  $d\bar{\epsilon}$ . In

$$\bar{\sigma} = K \bar{\epsilon}^n$$

# 3D Yield Effective stress



Tresca predicts 'flow' for lower stresses than von Mises





# Forming Limit Diagrams

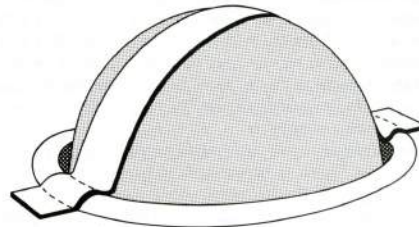
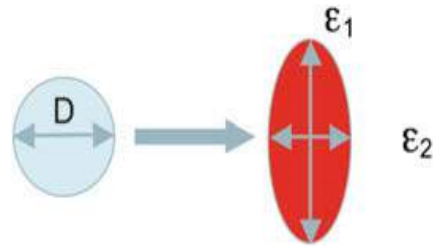
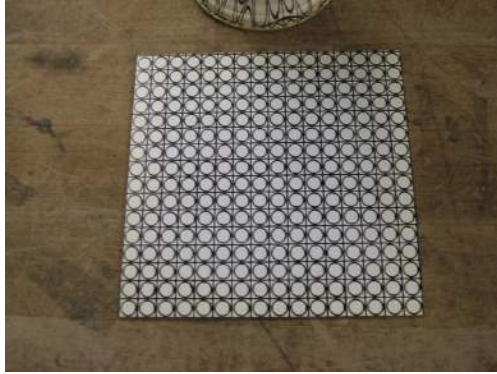
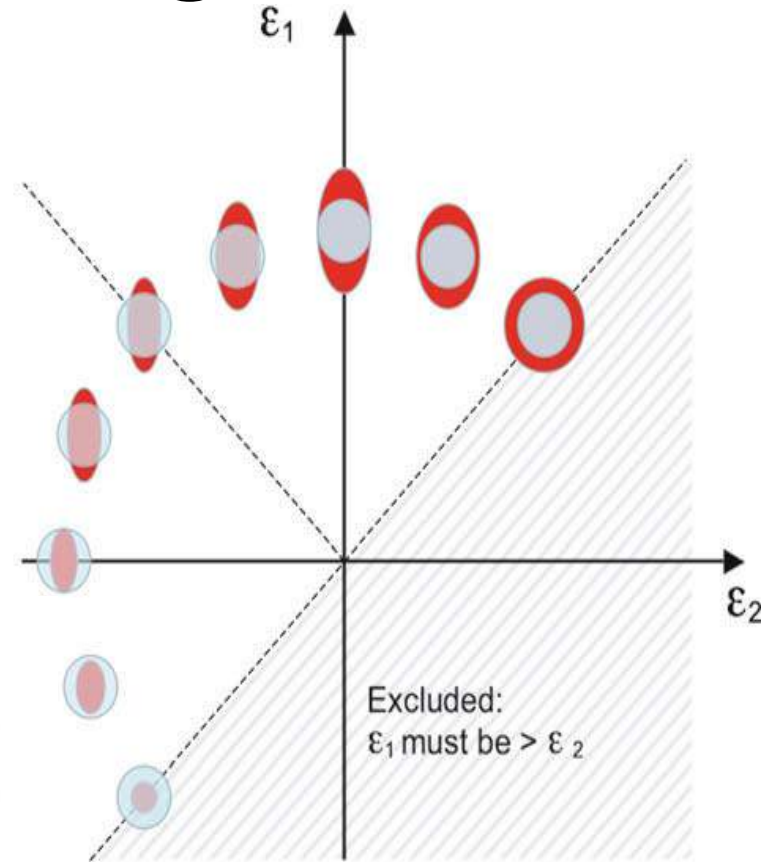
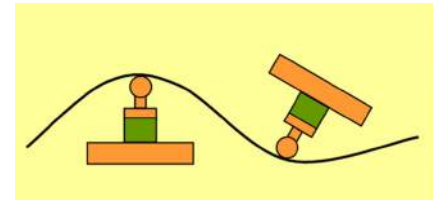
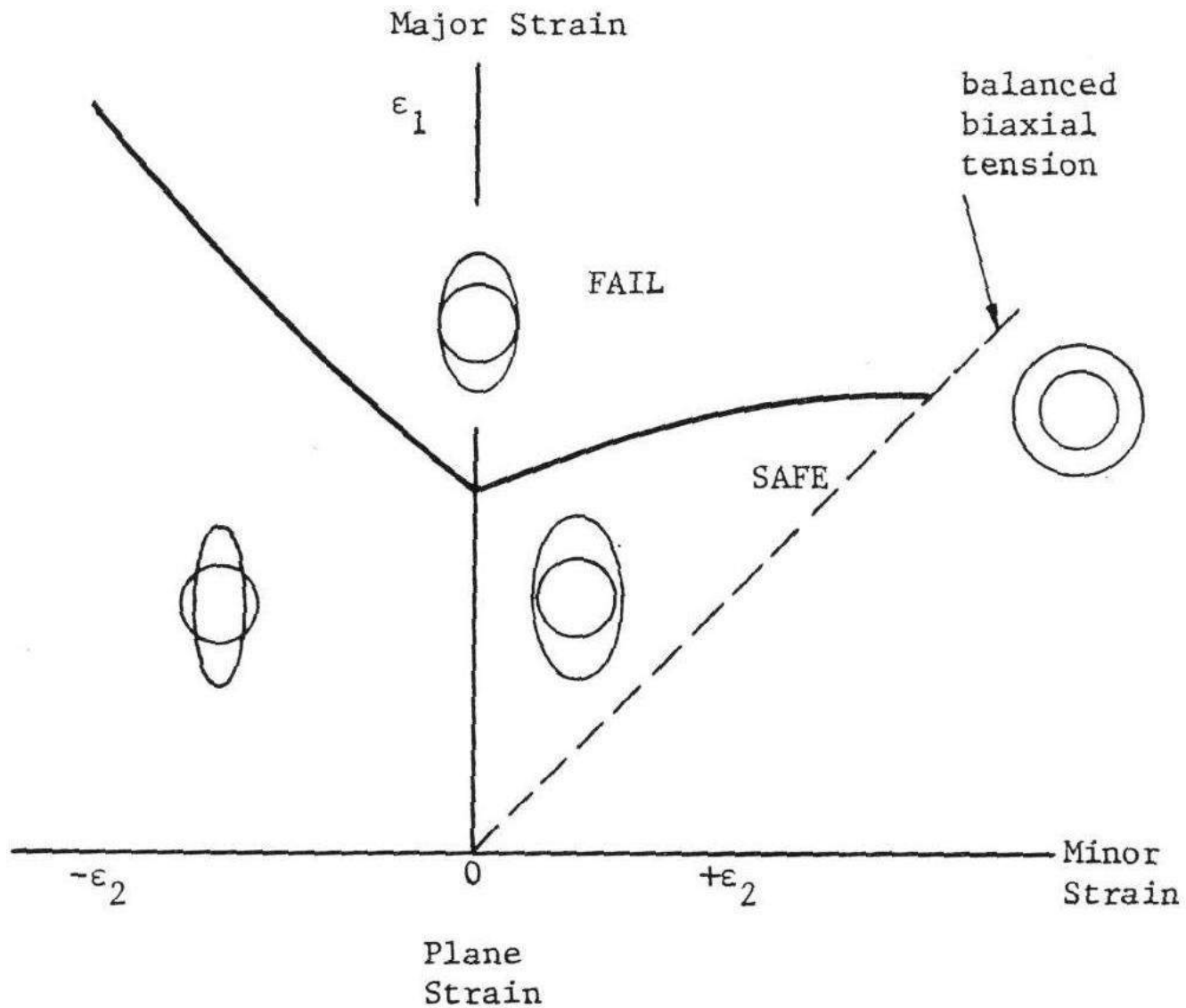


Figure 15-8 Strips of varying width are stretched to obtain different  $e_2/e_1$  ratios.

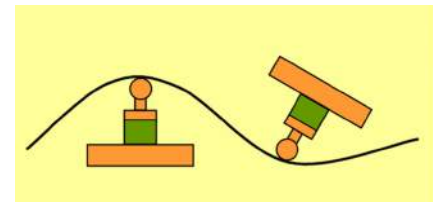


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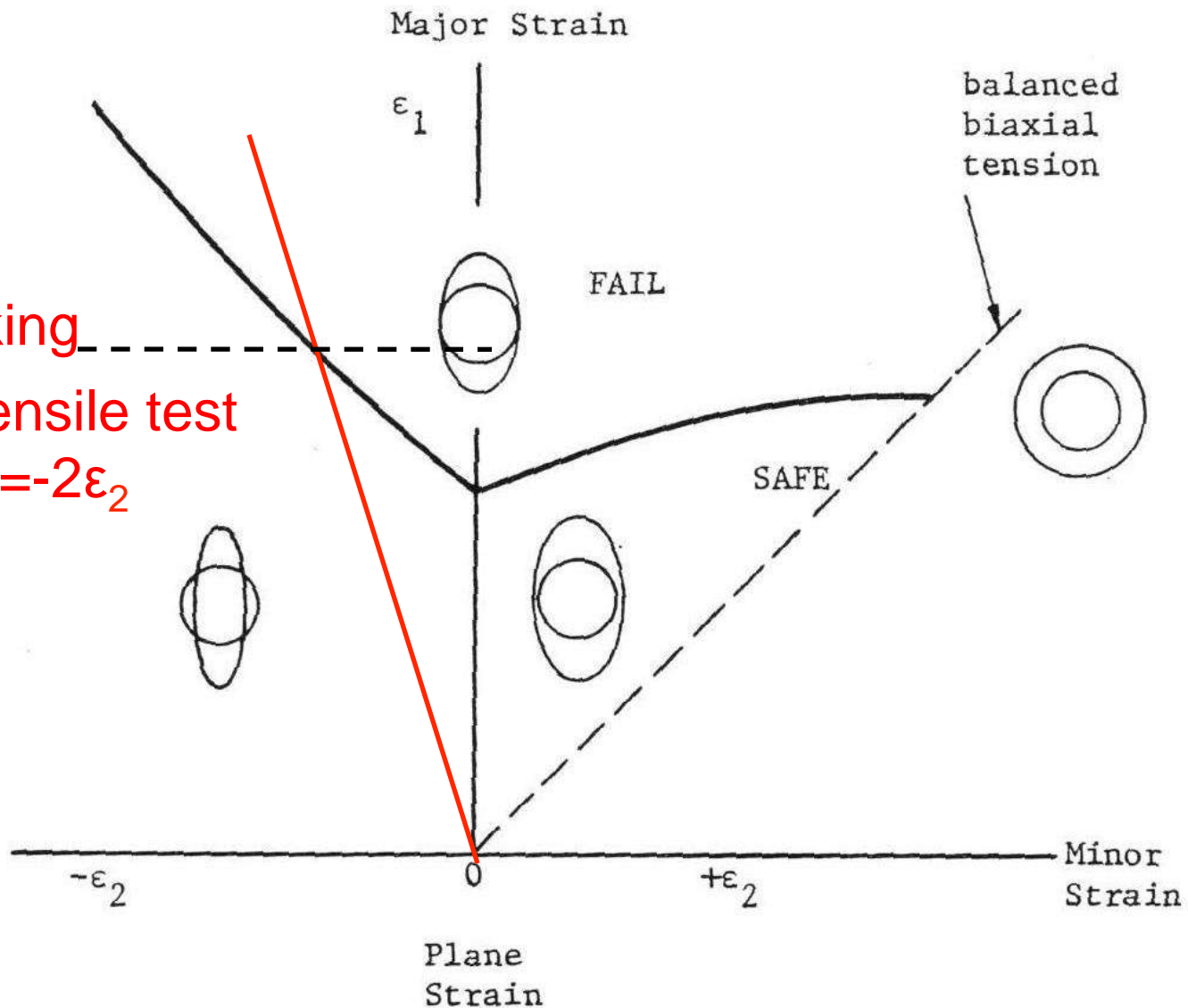
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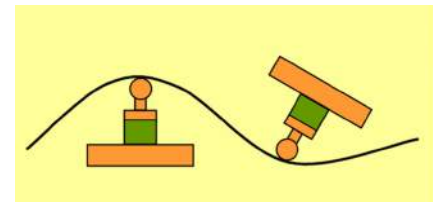
$\epsilon_1 = n = \text{necking}$

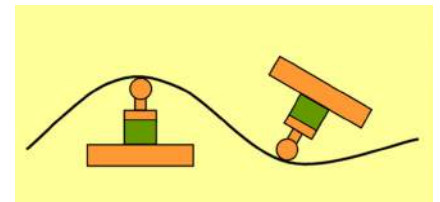
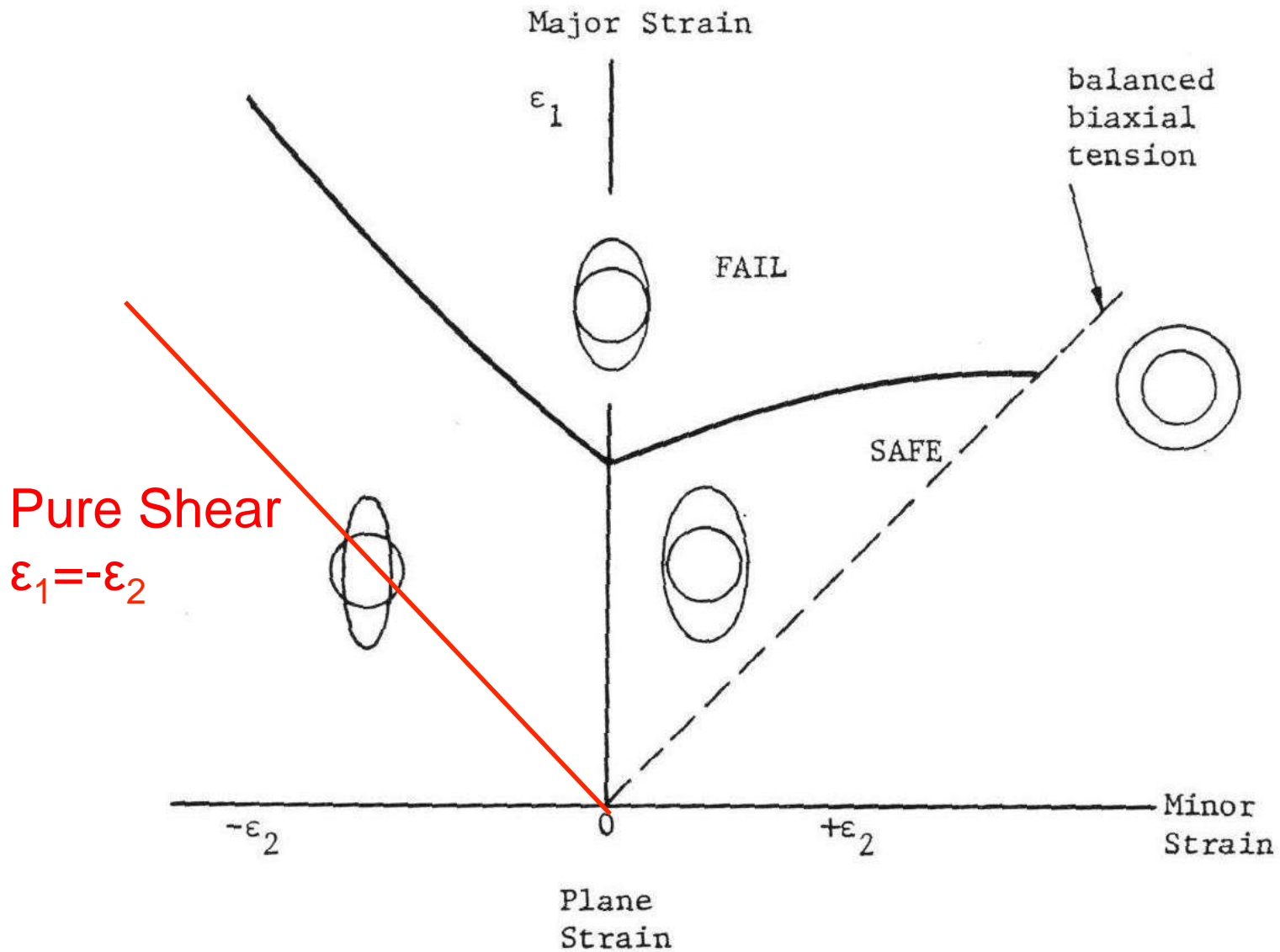
Tensile test

$\epsilon_1 = -2\epsilon_2$

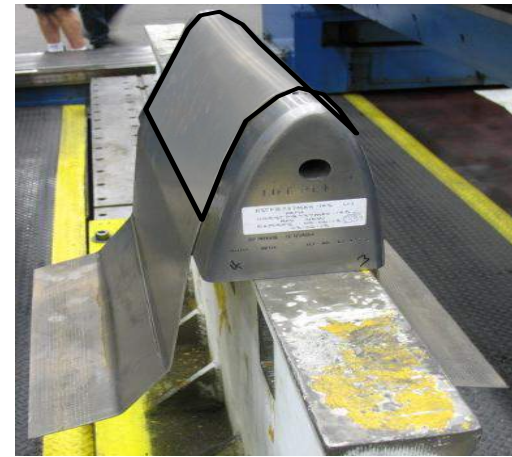


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# Stretch forming: **Forming force**



$$F = (Y_s + UTS)/2 * A$$

F = stretch forming force (lbs)

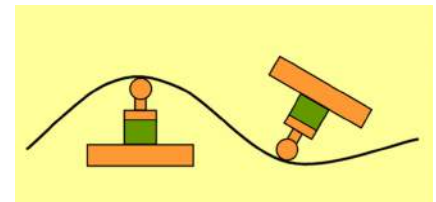
$Y_s$  = material yield strength (psi)

UTS = ultimate tensile strength of the material (psi)

A = Cross-sectional area of the workpiece (in<sup>2</sup>)

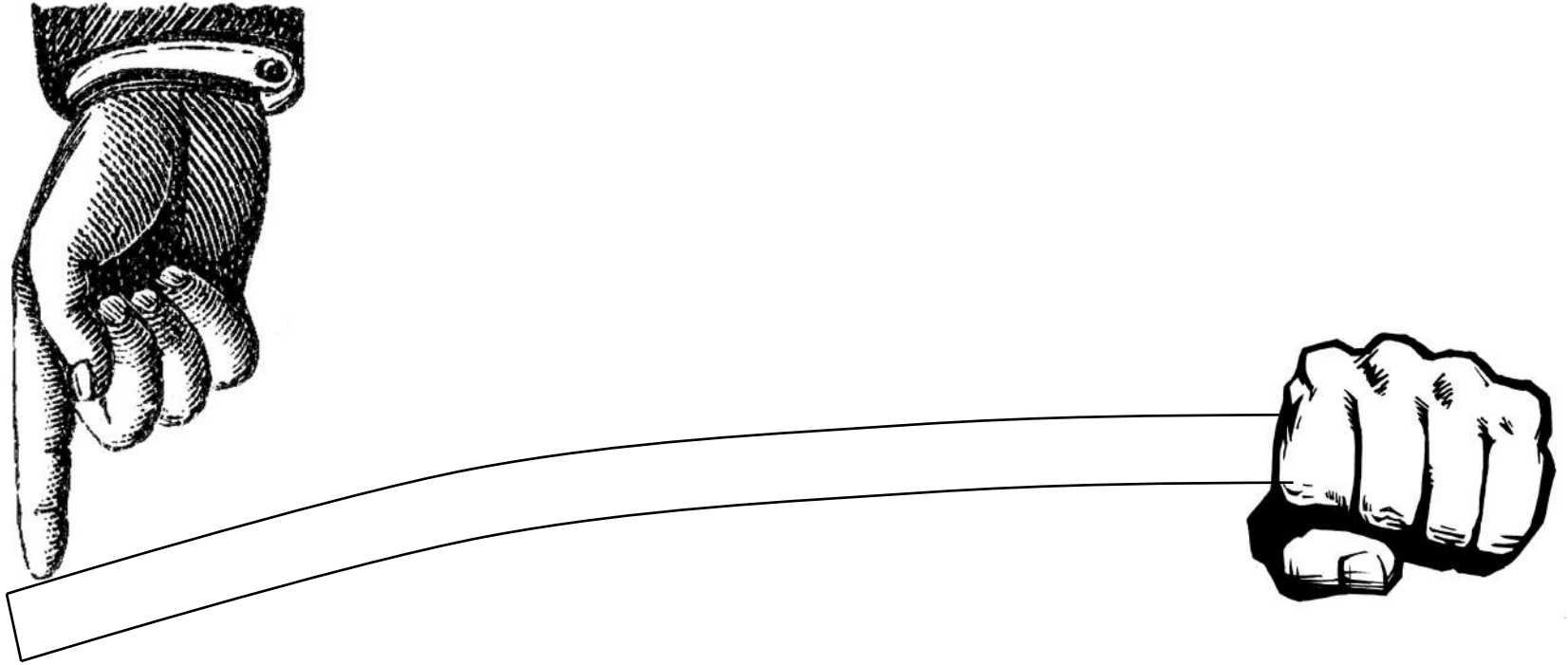


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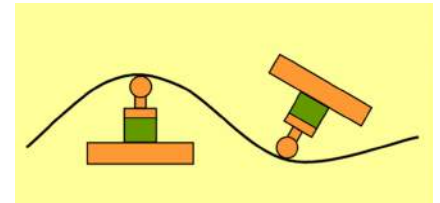




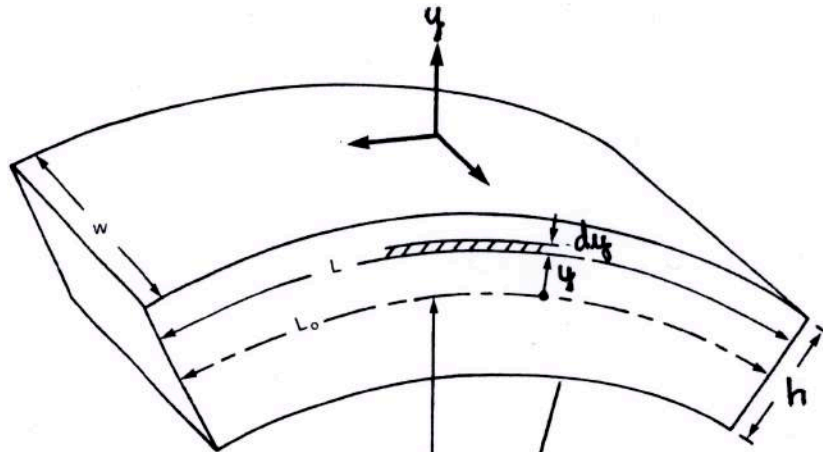
# Forces needed to bend sheet metal



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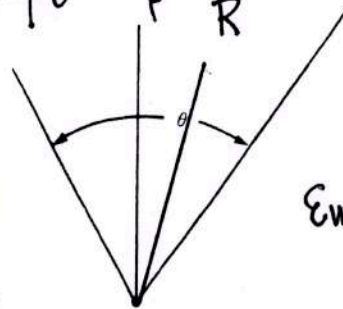
# Bending



note:  $L > L_0$

$$\Delta L = (L - L_0) = (R + y)\theta - R\theta = y\theta$$

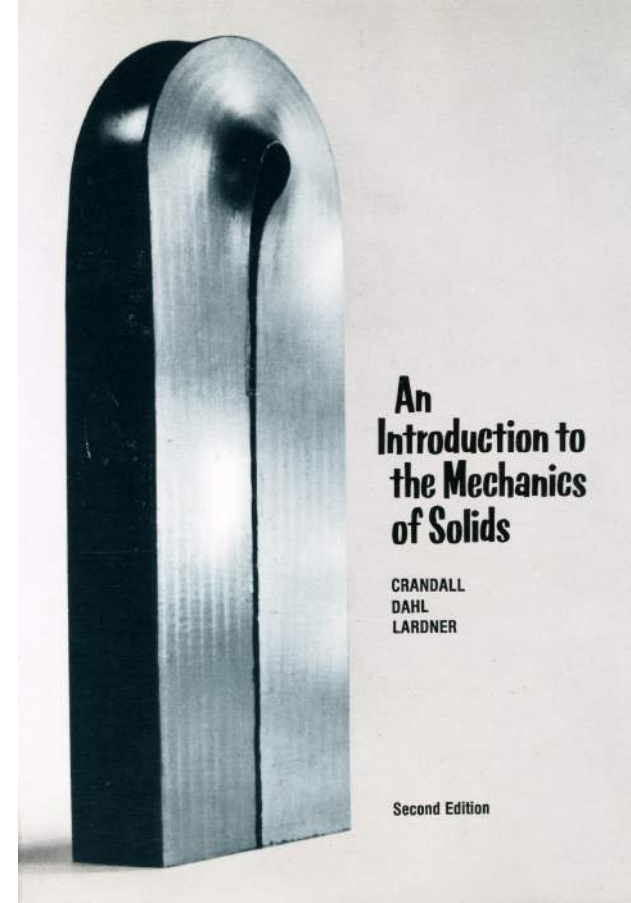
$$\epsilon = \frac{\Delta L}{L_0} = \frac{y\theta}{R\theta} = \frac{y}{R}$$



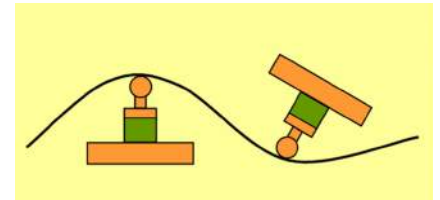
if  $\rho = R + \frac{h}{2}$

$$\epsilon_{max} = \frac{h/2}{R + h/2} = \frac{1}{\frac{2R}{h} + 1}$$

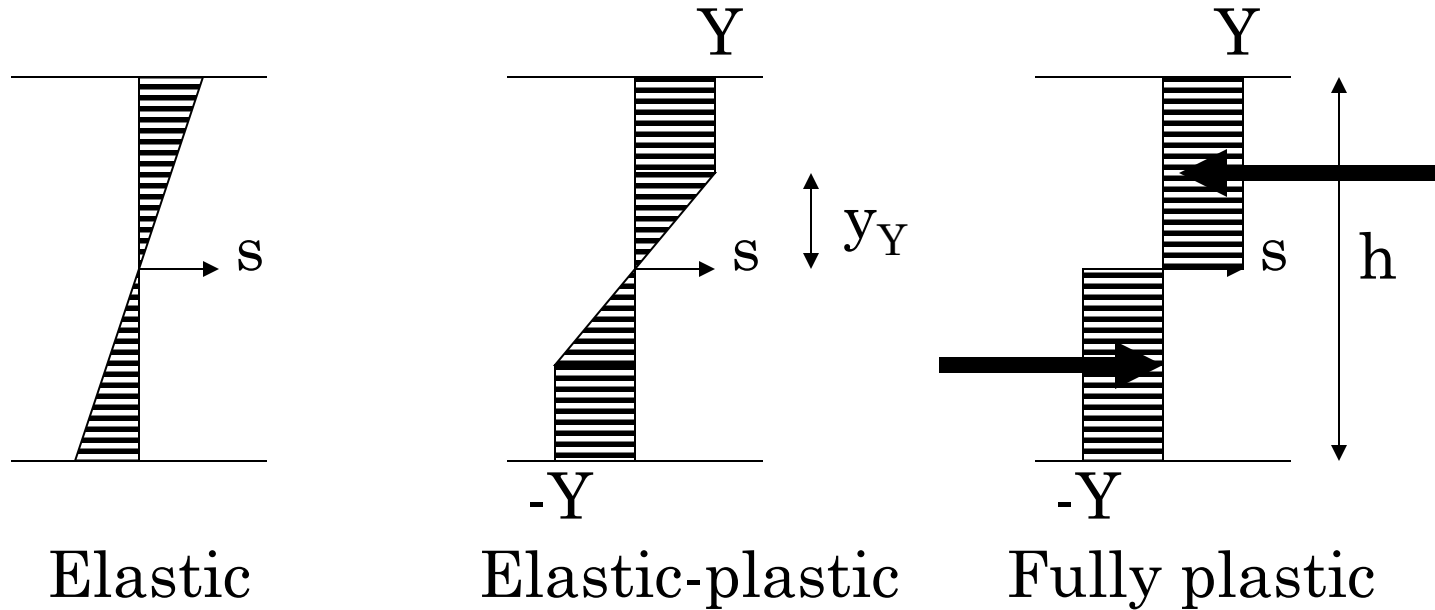
Figure Coordinate system for analysis of bending.



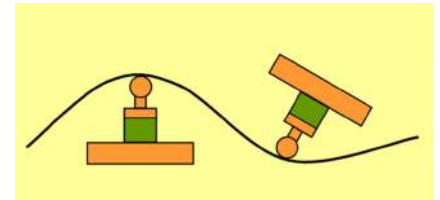
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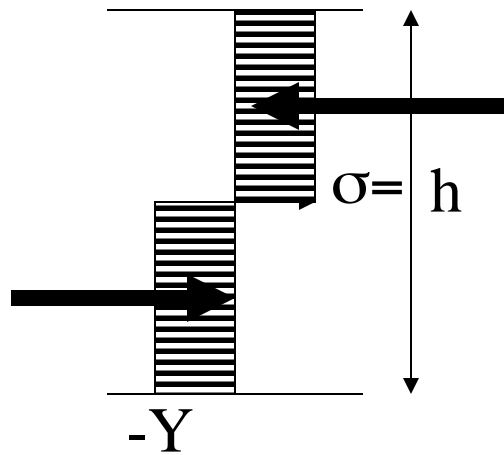
# Stress distribution through the thickness of the part



Fully Plastic Moment,  $M = Y (b h/2) h/2 = Ybh^2/4$



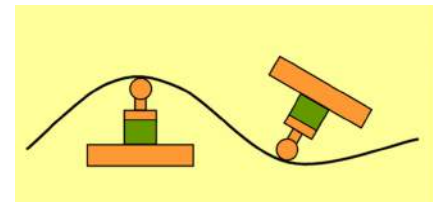
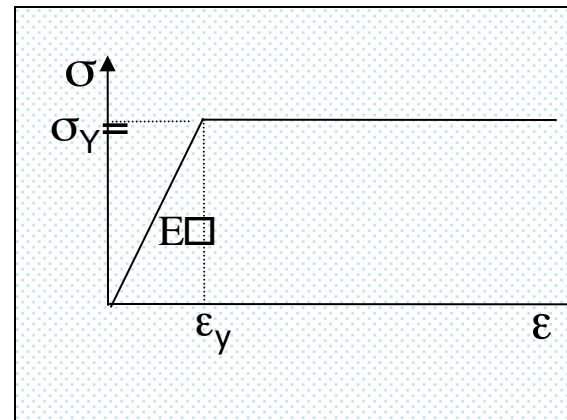
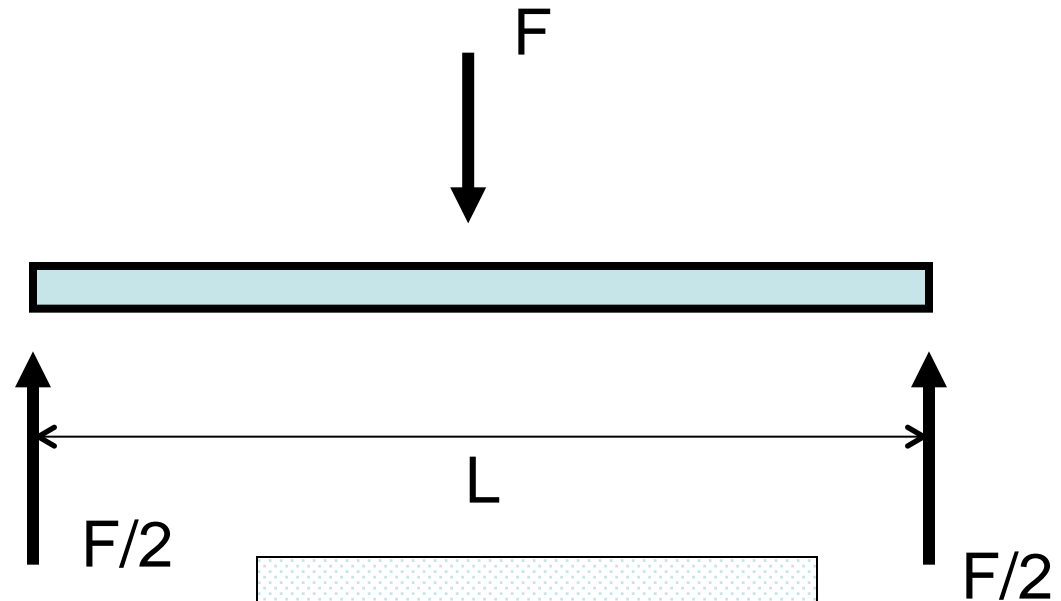
# Balance external and internal moments



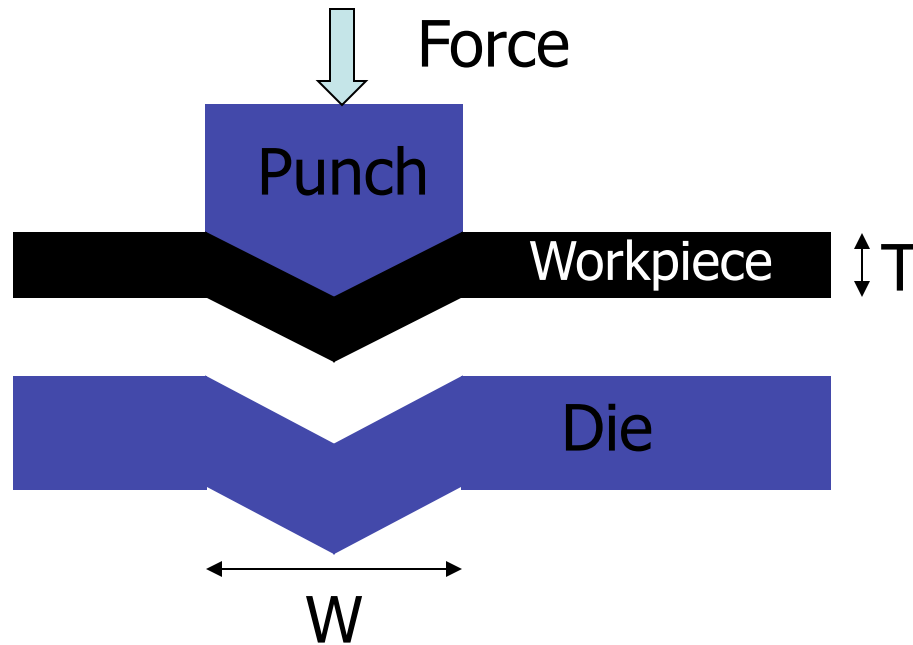
Fully plastic

$$Ybh^2/4 = FL/4 = M_{\max}$$

$$F = bh^2Y/L$$



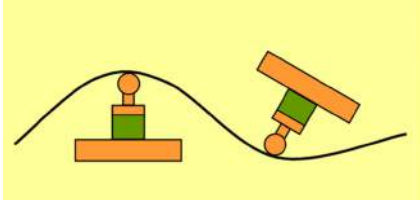
# Bending Force Requirement



$$F = \frac{LT^2}{W}(UTS)$$

T = Sheet Thickness  
W = Width of Die Opening  
L = Total length of bend  
(into the page)  
UTS = Ultimate Tensile  
Strength of material

Note: the notation used in the text (L, W) differs from that used in the previous development (b, L).



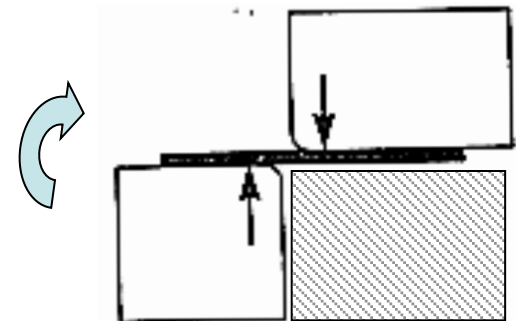
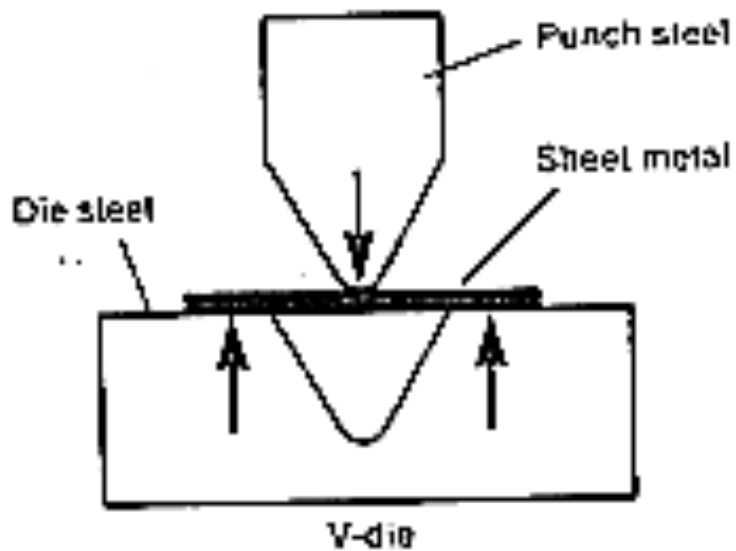


# *LMP Shop*

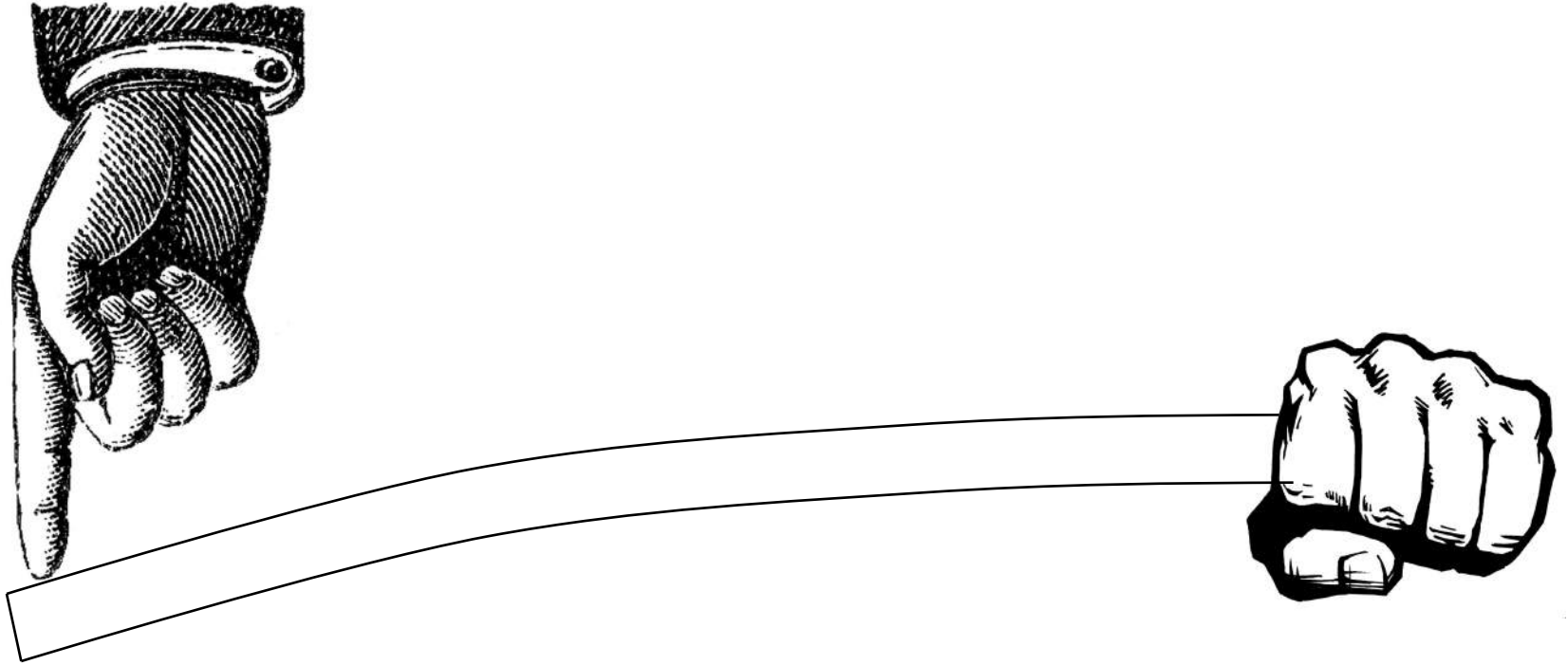
Brake press



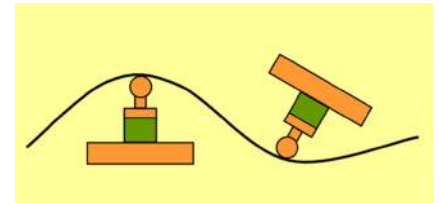
Finger brake



# What shape have we created?

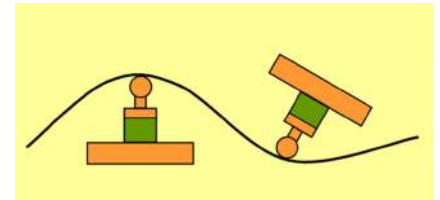
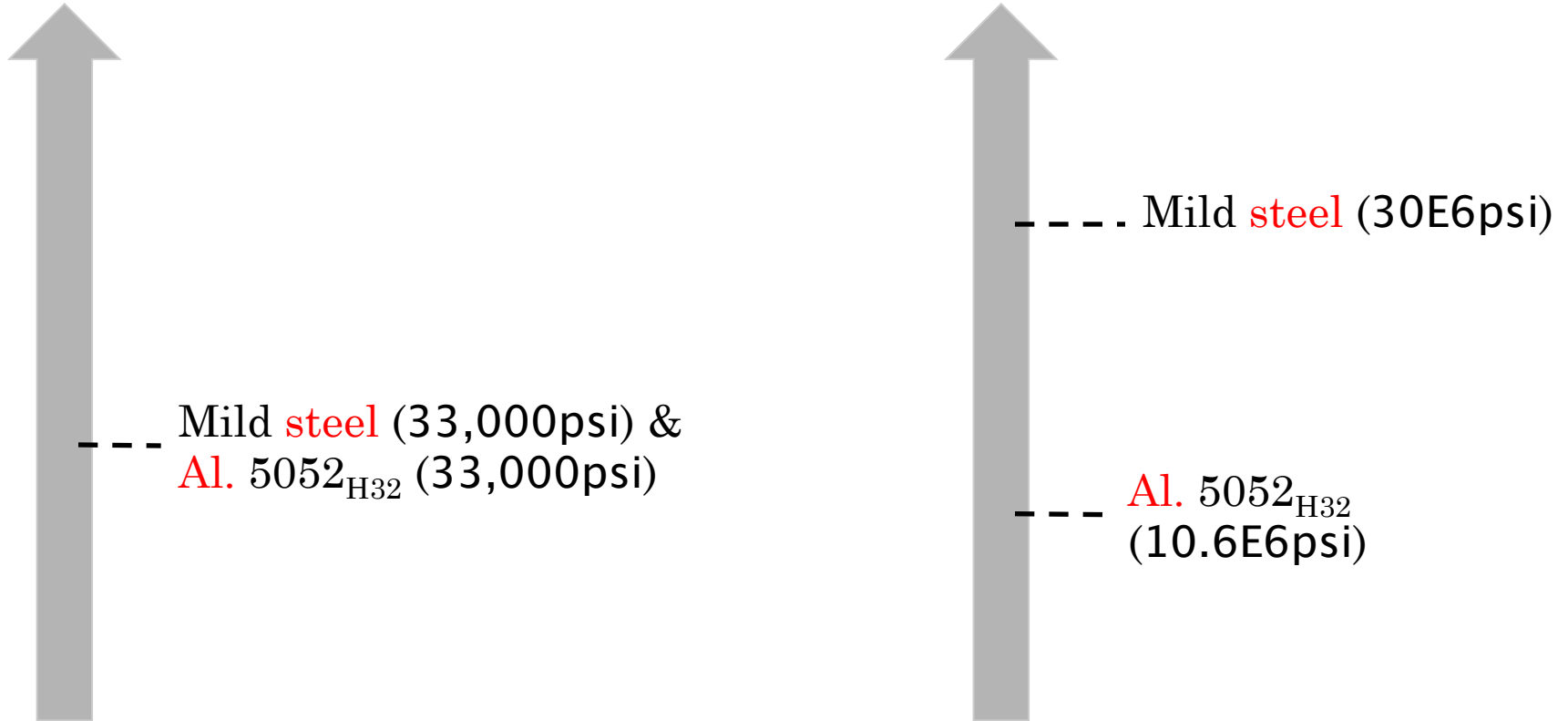


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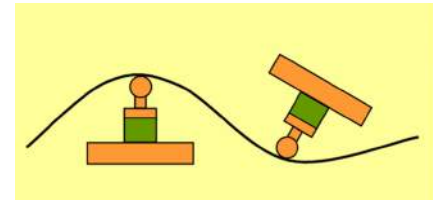
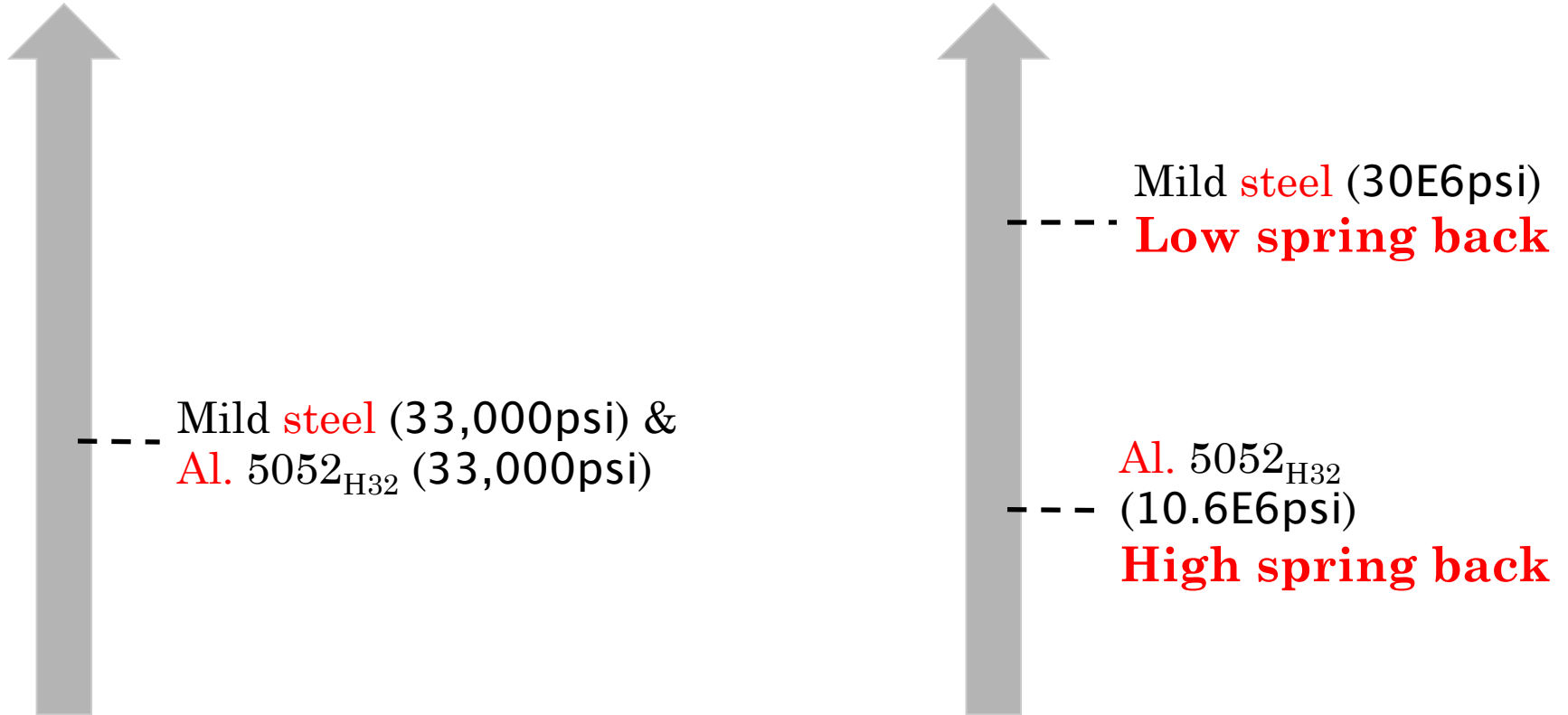
# Steel versus aluminum...

## Strength ( $\sigma_y$ ) versus Stiffness (E)



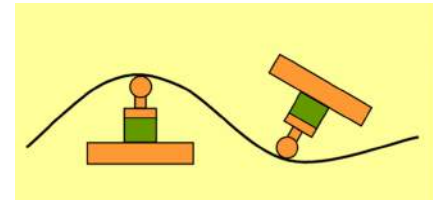
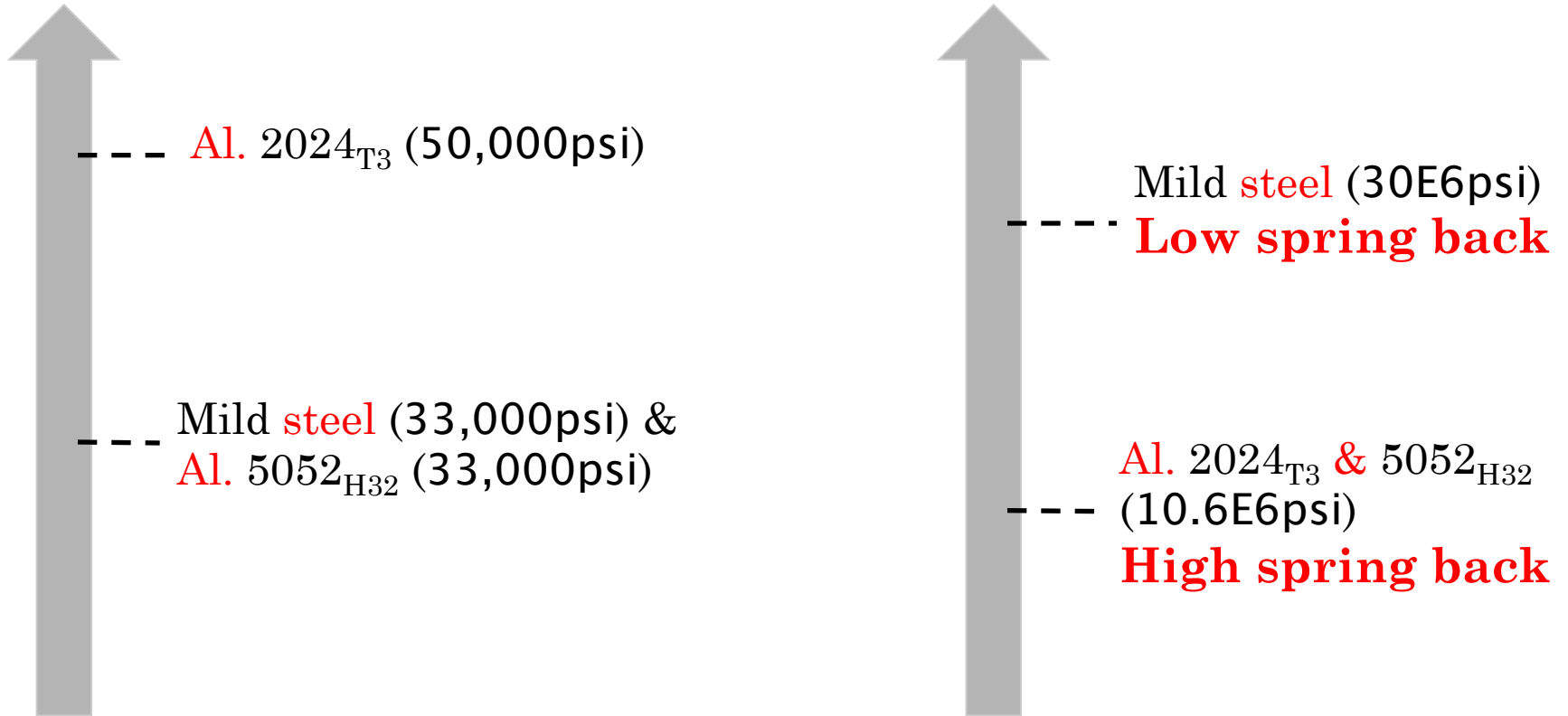
# Steel versus aluminum...

## Strength ( $\sigma_y$ ) versus Stiffness (E)



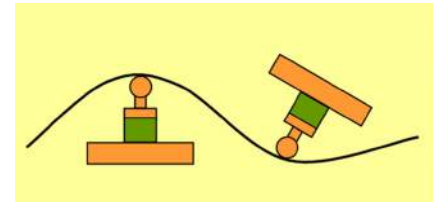
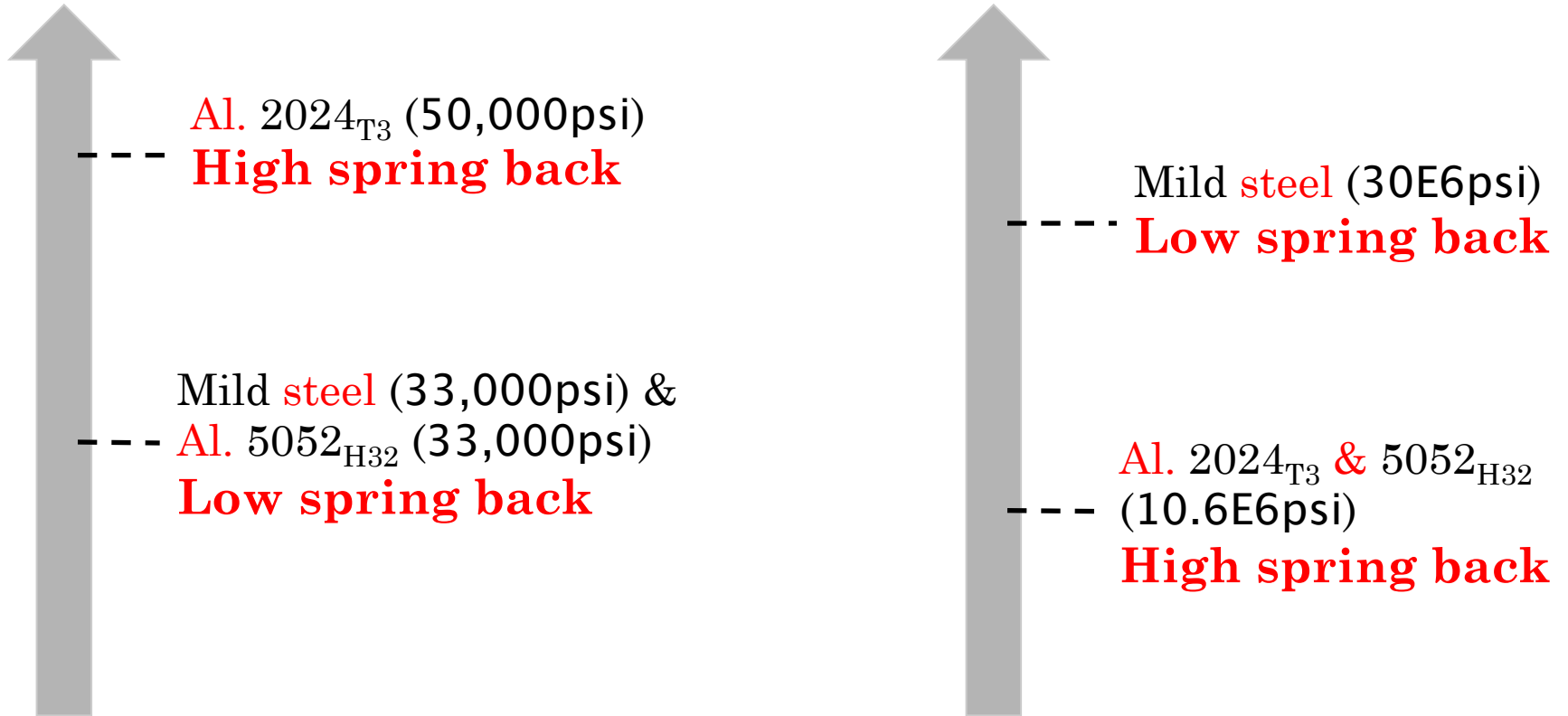
# Steel versus aluminum...

## Strength ( $\sigma_y$ ) versus Stiffness (E)



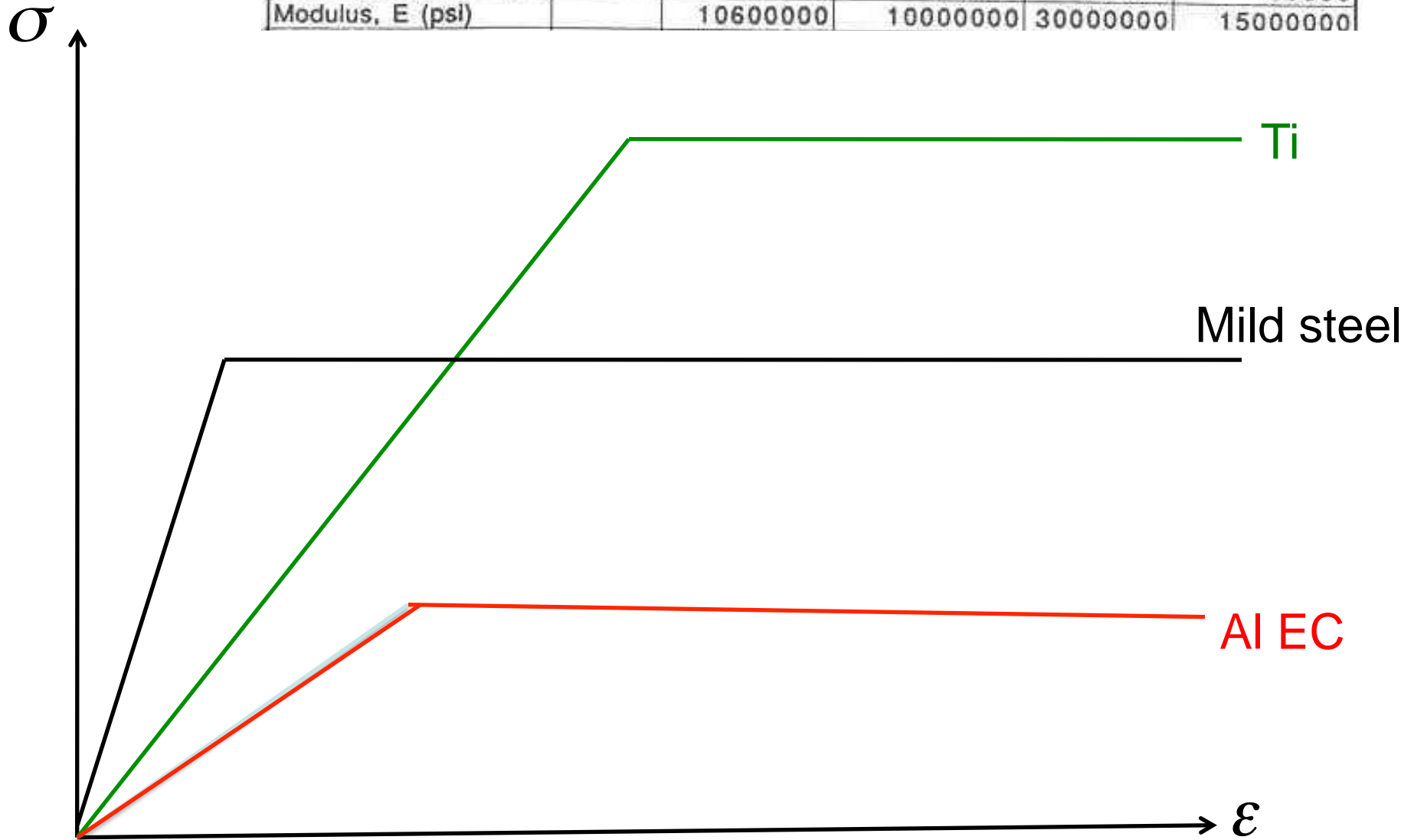
# Steel versus aluminum...

## Strength ( $\sigma_y$ ) versus Stiffness (E)

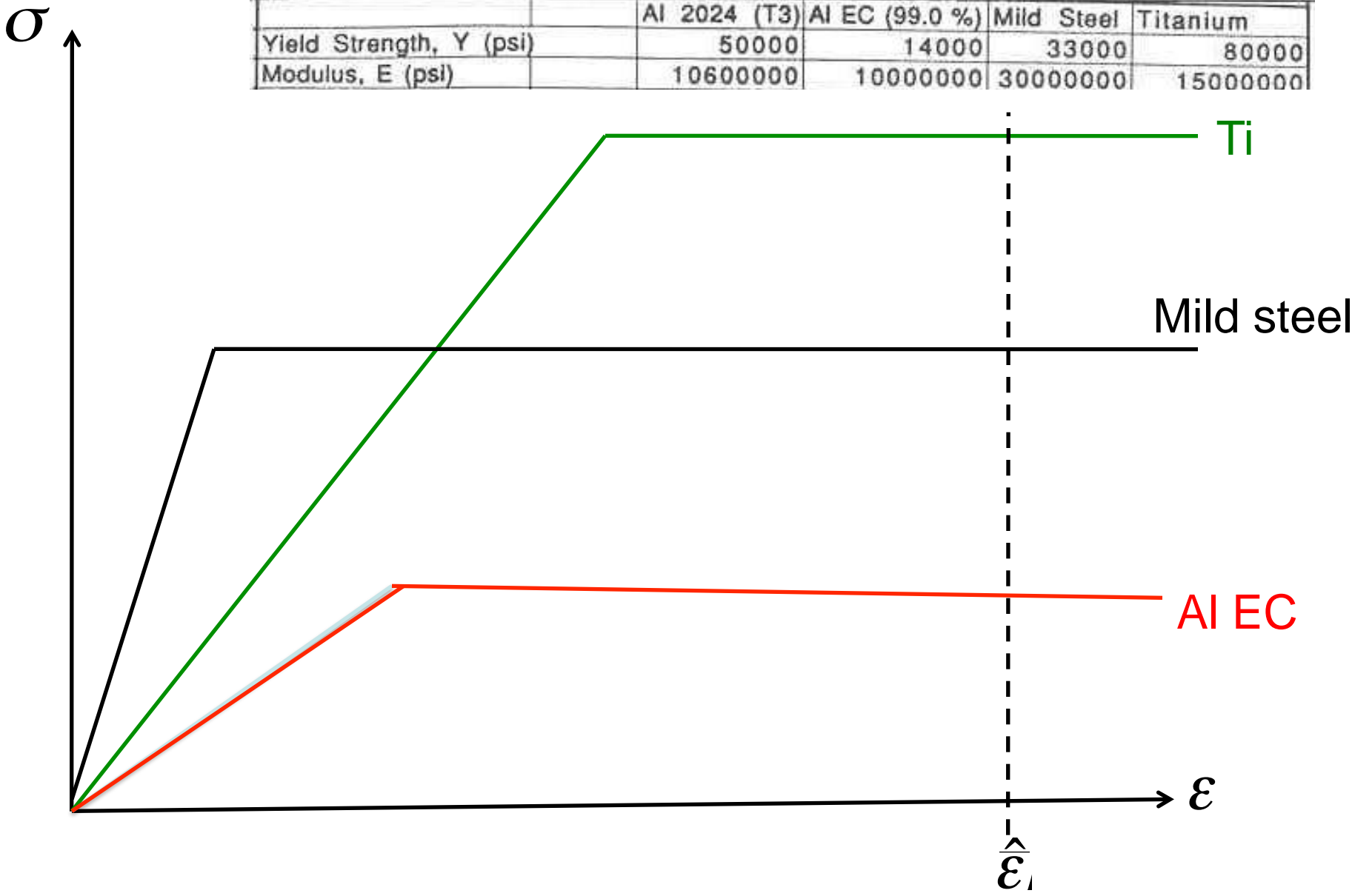




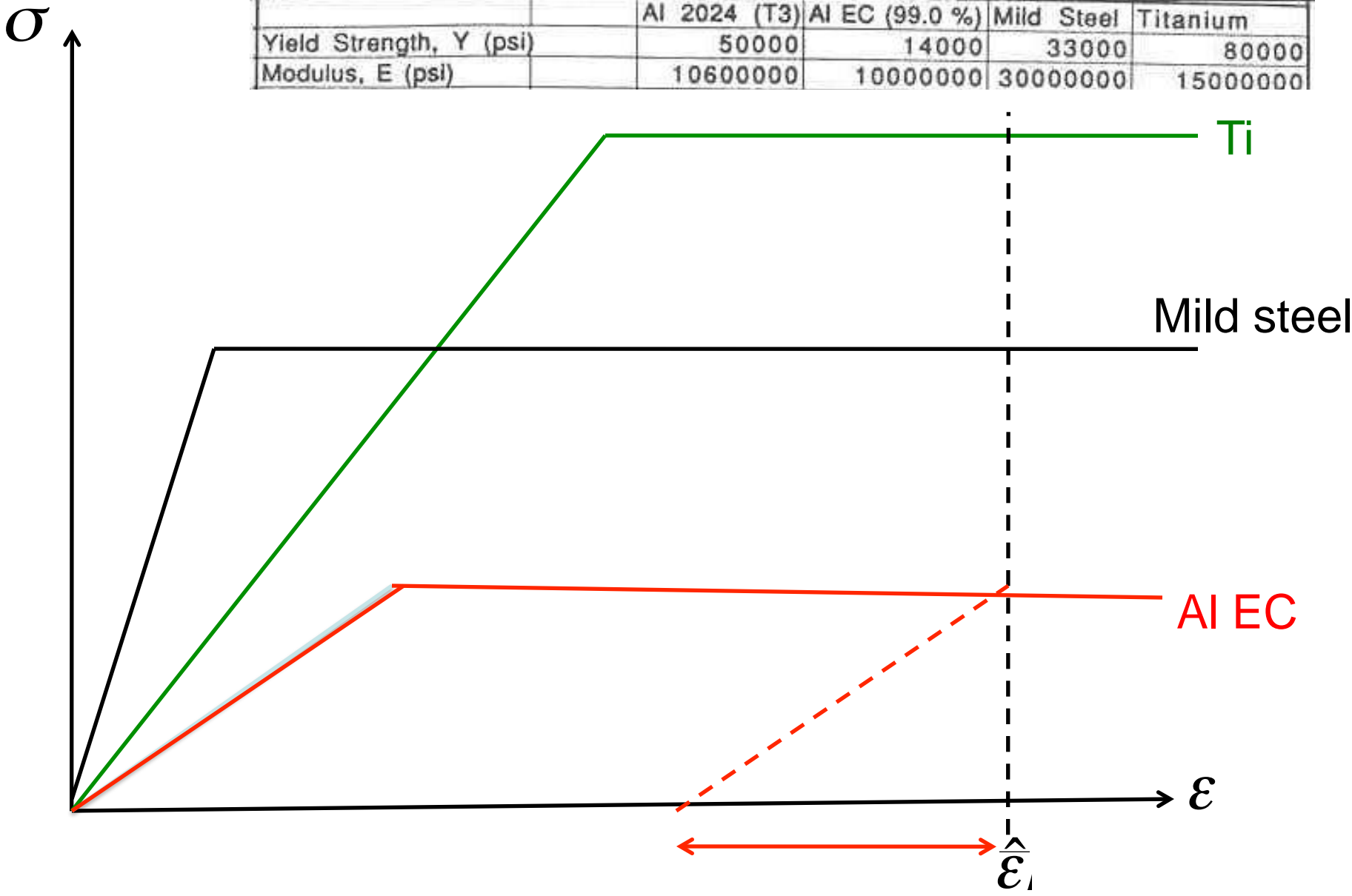
	Al 2024 (T3)	Al EC (99.0 %)	Mild Steel	Titanium
Yield Strength, Y (psi)	50000	14000	33000	80000
Modulus, E (psi)	10600000	10000000	30000000	15000000



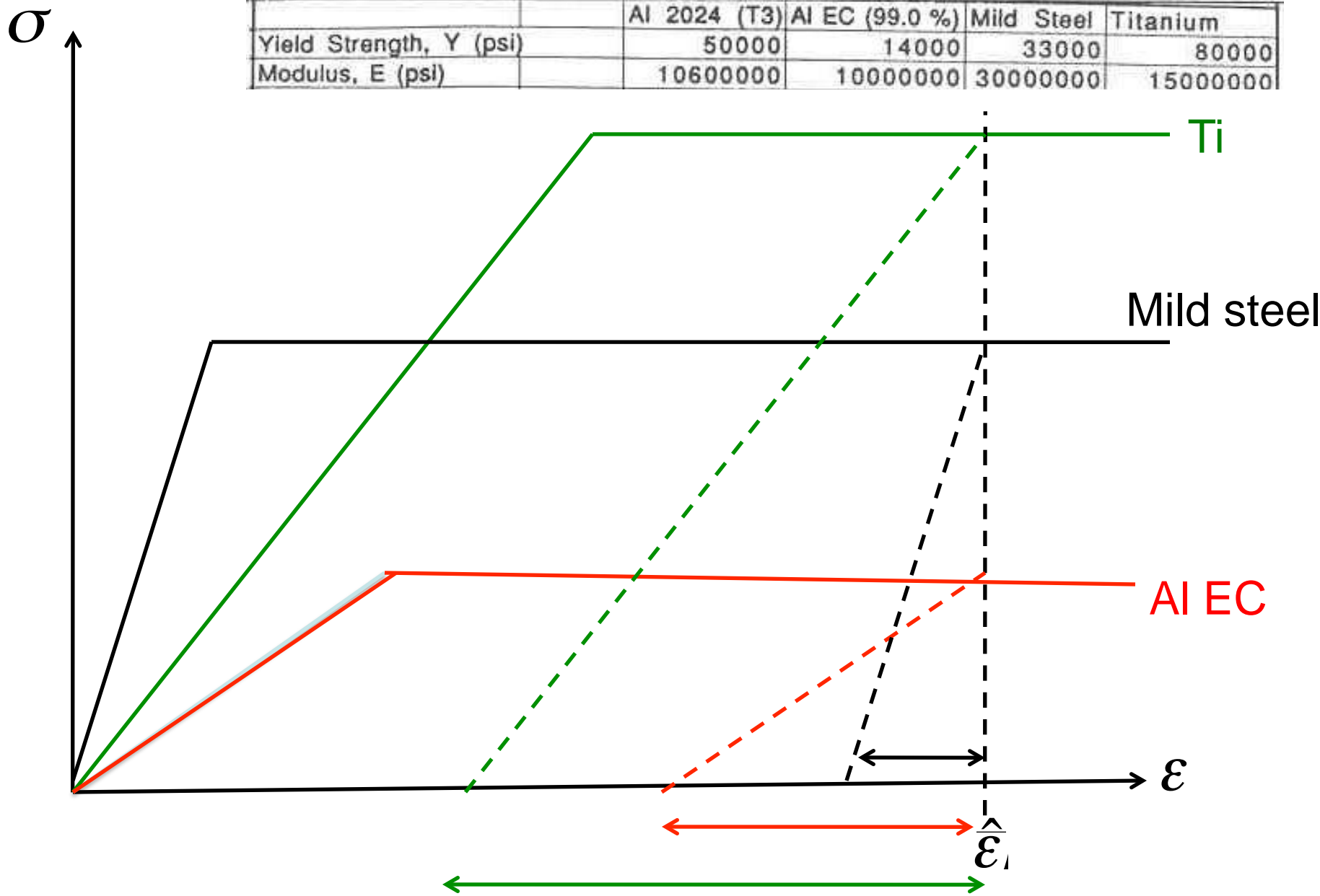
	Al 2024 (T3)	Al EC (99.0 %)	Mild Steel	Titanium
Yield Strength, Y (psi)	50000	14000	33000	80000
Modulus, E (psi)	10600000	10000000	30000000	15000000



	Al 2024 (T3)	Al EC (99.0 %)	Mild Steel	Titanium
Yield Strength, Y (psi)	50000	14000	33000	80000
Modulus, E (psi)	10600000	10000000	30000000	15000000

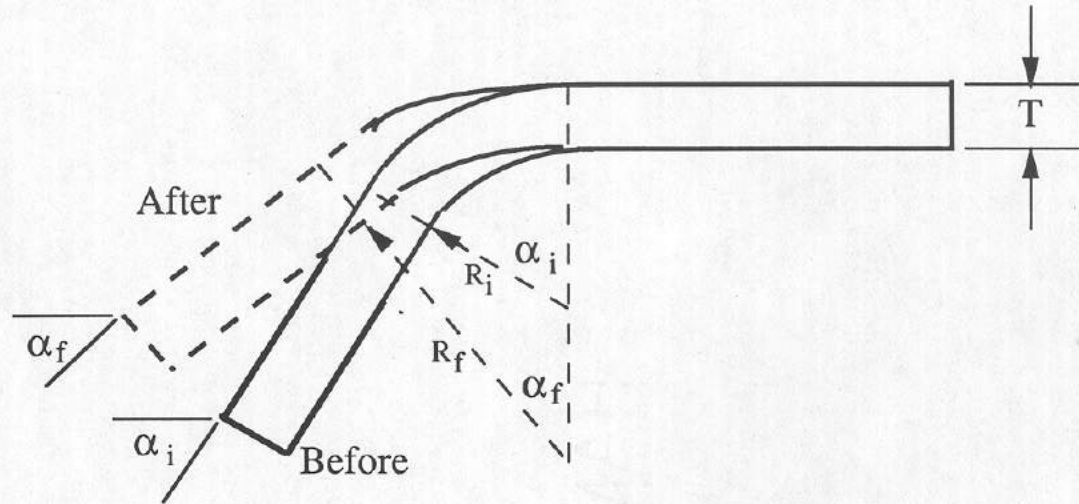


	Al 2024 (T3)	Al EC (99.0 %)	Mild Steel	Titanium
Yield Strength, Y (psi)	50000	14000	33000	80000
Modulus, E (psi)	10600000	10000000	30000000	15000000

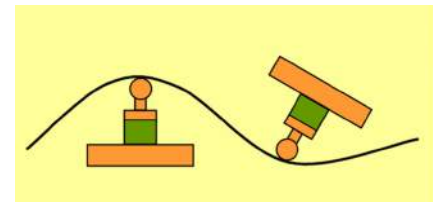


# Springback

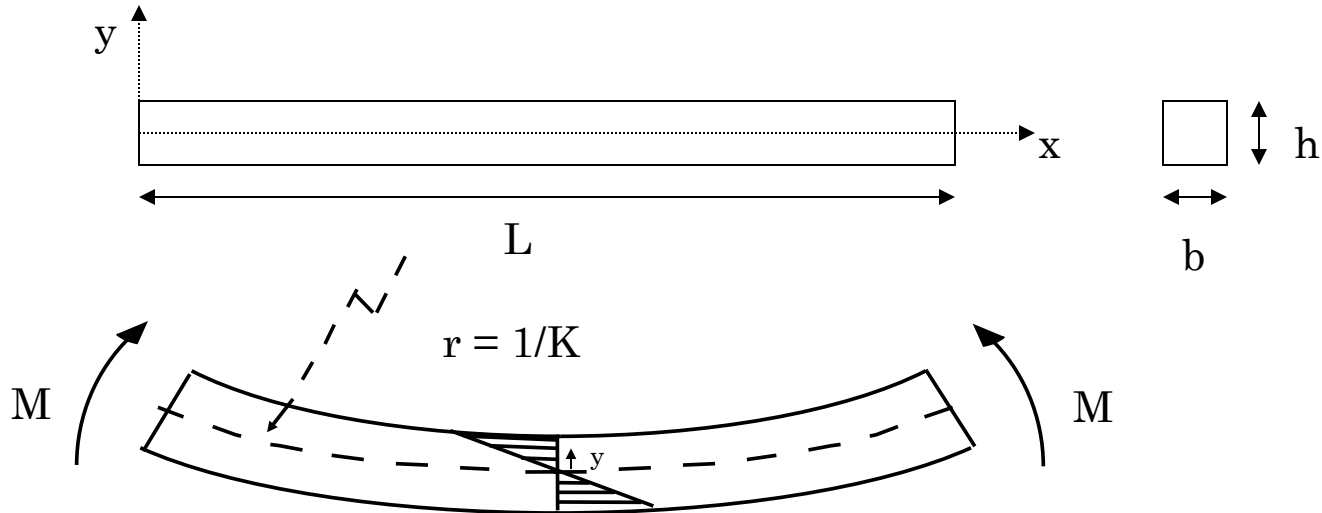
note R in the figure below is mislabeled, should go to the centerline of the sheet



Springback: 
$$\frac{R_i}{R_f} = 4 \left( \frac{R_i Y}{E T} \right)^3 - 3 \left( \frac{R_i Y}{E T} \right) + 1$$



# Elastic Springback Analysis

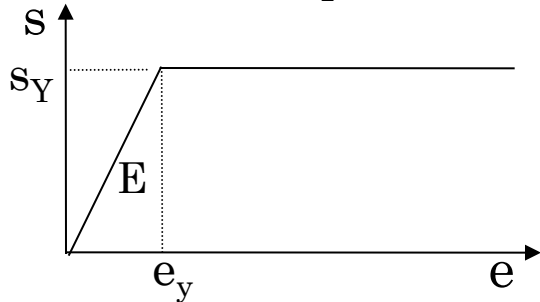


1. Assume plane sections remain plane:  

$$e_y = -y/r$$

(1)

2. Assume elastic-plastic behavior for material



$$\sigma = E e$$

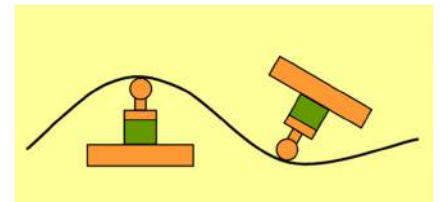
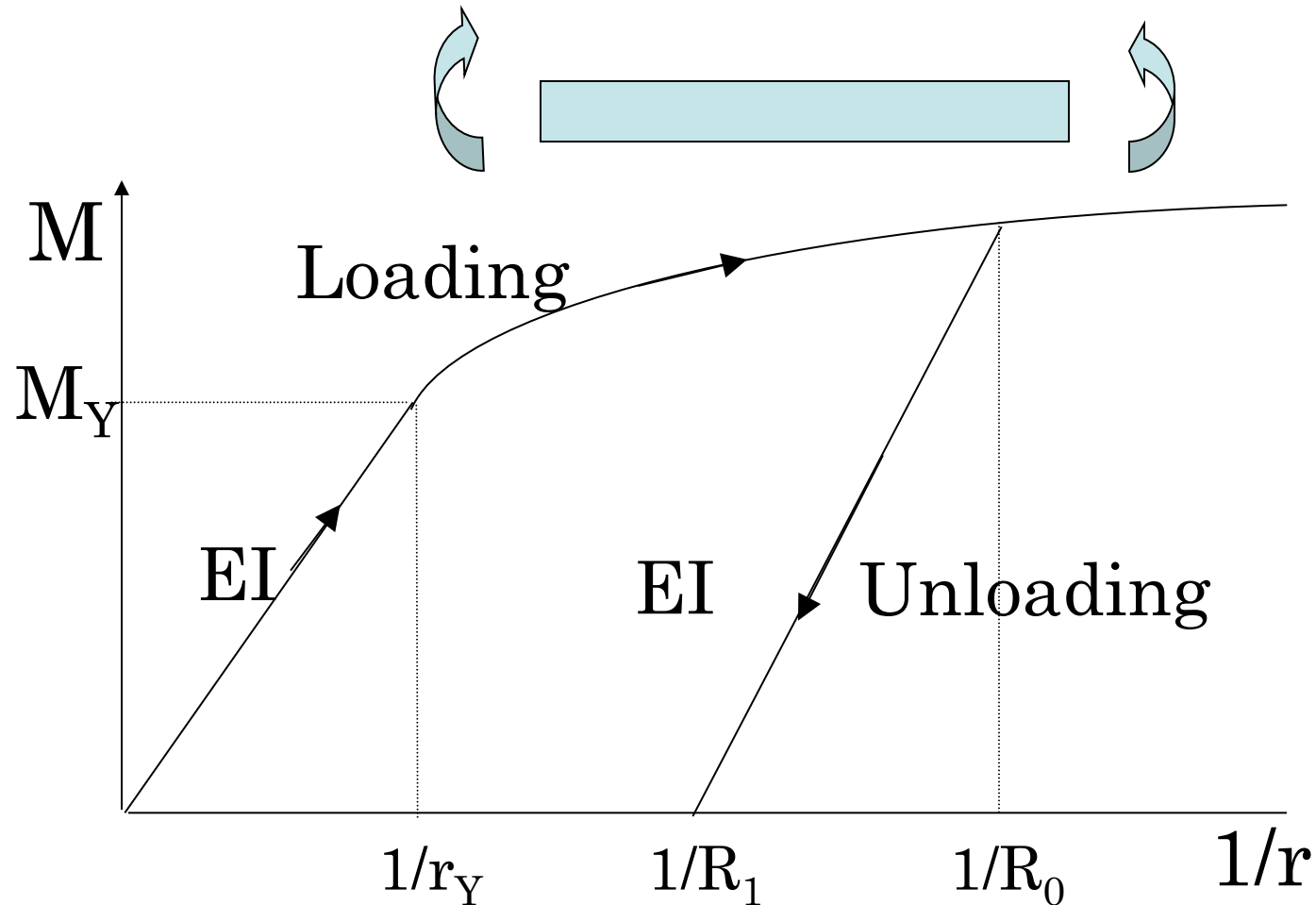
$$e < e_y$$

$$\sigma = \sigma_Y$$

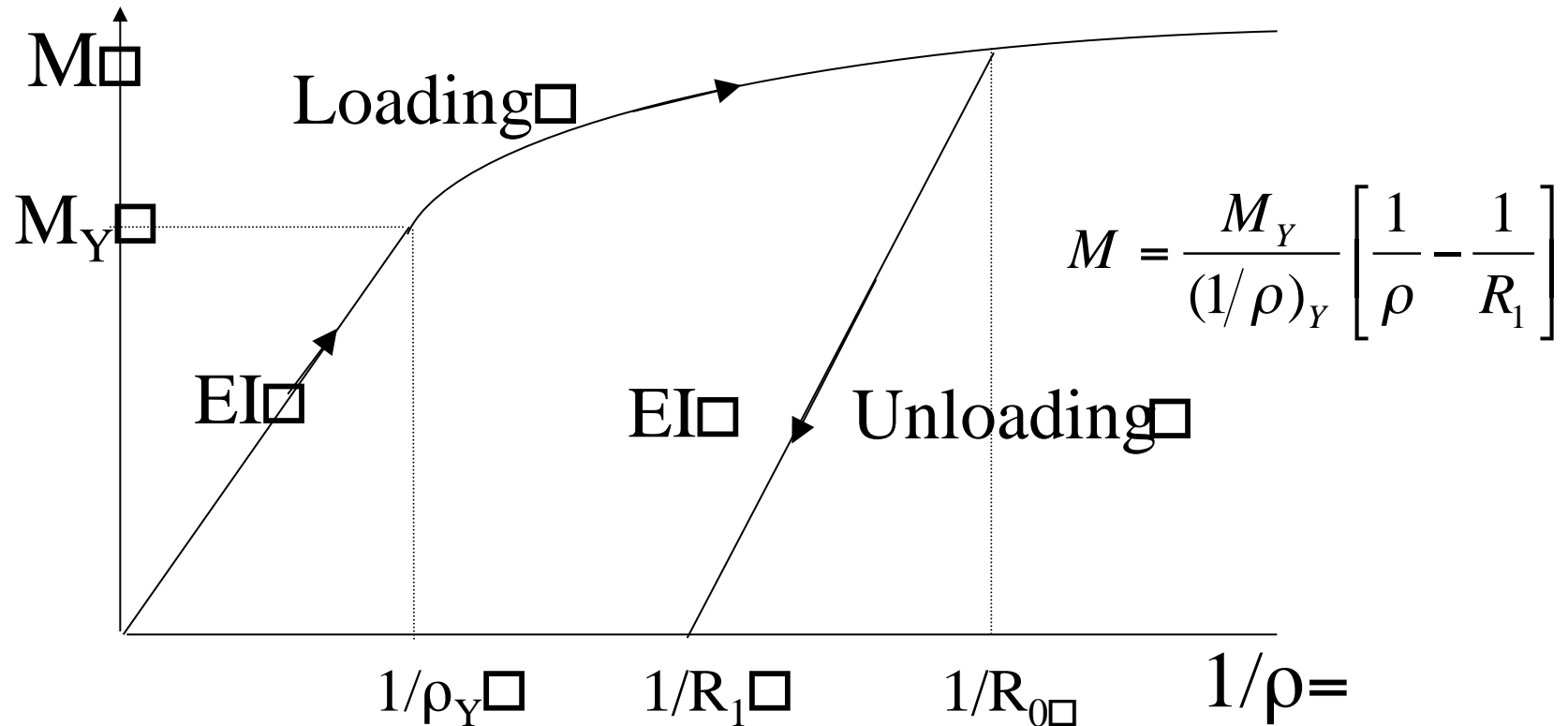
$$e \geq e_y$$



# Bending Moment – Curvature



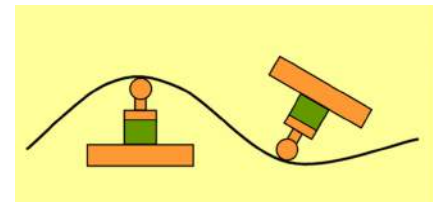
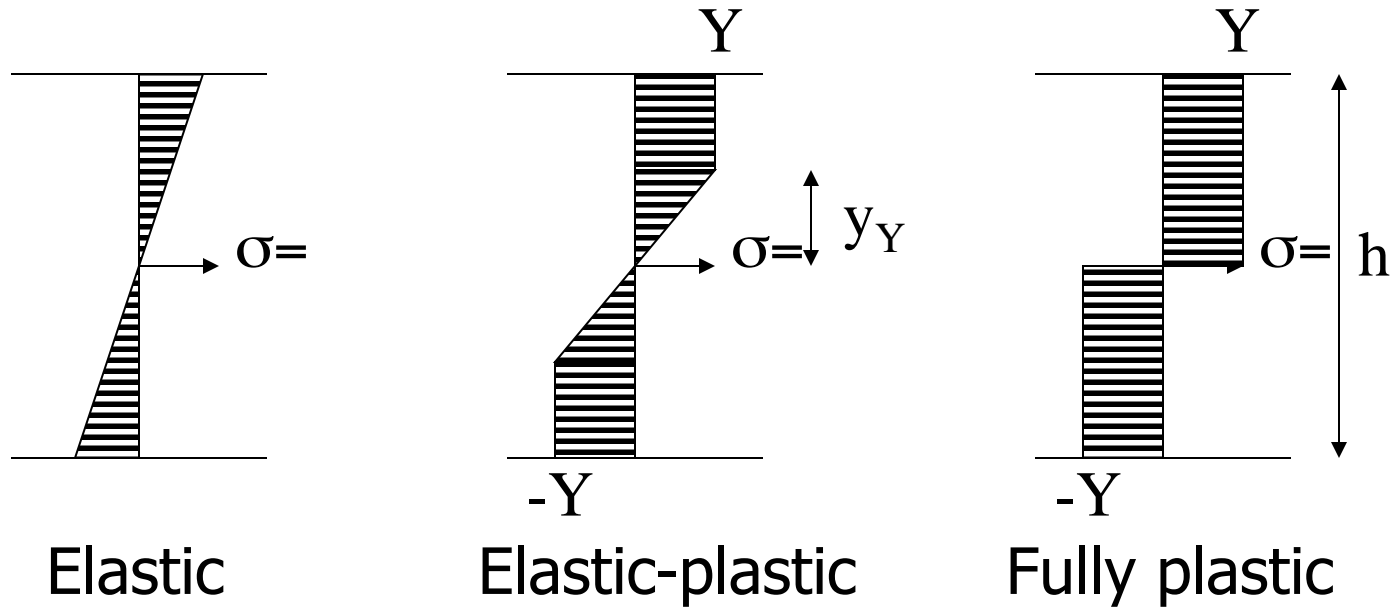
3. We want to construct the following  
Bending Moment “M” vs. curvature “1/ρ” curve



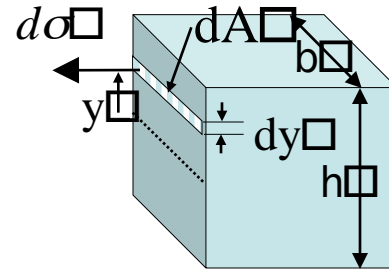
Springback is measured as  
Permanent set is

$$\frac{1/R_0 - 1/R_1}{1/R_1} \quad (2)$$

## 4. Stress distribution through the thickness of the beam



$$5. M = \int_A \sigma y dA$$



Elastic region

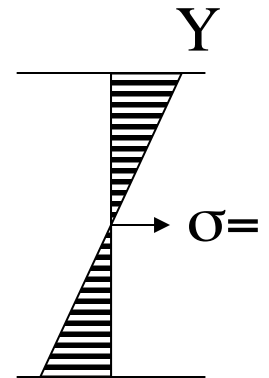
$$M = \int \sigma y dA = -E \int \frac{y^2}{\rho} dA = -\frac{EI}{\rho} \quad (3)$$

At the onset of plastic behavior

$$\sigma = -y/\rho E = -h/2\rho E = -Y \quad (4)$$

This occurs at

$$1/\rho = 2Y/hE = 1/\rho_Y \quad (5)$$

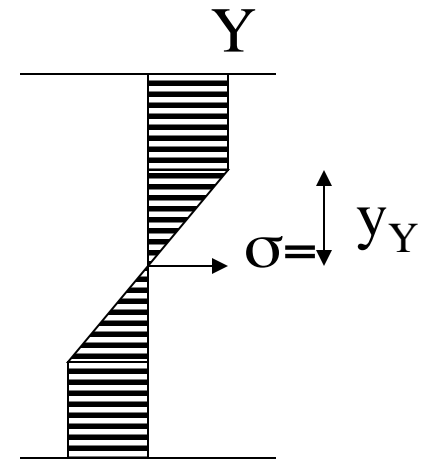


Substitution into eqn (3) gives us the moment at on-set of yield,  $M_Y$

$$M_Y = -EI/\rho_Y = EI 2Y/hE = 2IY/h \quad (6)$$

After this point, the  $M$  vs  $1/r$  curve starts to “bend over.”  
Note from  $M=0$  to  $M=M_Y$  the curve is linear.

In the elastic – plastic region



$$M = \int \sigma y b dy = 2 \int_{y_Y}^{h/2} Y b y dy \square 2 \int_0^{y_Y} \frac{y}{y_Y} Y b y dy$$

$$= 2 Y b \frac{y^2}{2} \Big|_{y_Y}^{h/2} \square 2 \frac{Y}{y_Y} b \frac{y^3}{3} \Big|_0^{y_Y}$$

$$= Y b \left( \frac{h^2}{4} - y_Y^2 \right) \square \frac{2}{3} y_Y^2 Y b$$

$$M = \frac{b h^2}{4} Y \left[ 1 - \frac{1}{3} \left( \frac{y_Y}{h/2} \right)^2 \right] \quad (7)$$

Note at  $y_Y = h/2$ , you get on-set at yield,  $M = M_Y$

And at  $y_Y = 0$ , you get fully plastic moment,  $M = 3/2 M_Y$

To write this in terms of  $M$  vs  $1/\rho$  rather than  $M$  vs  $y_Y$ , note that the yield curvature  $(1/\rho)_Y$  can be written as (see eqn (1))

$$\frac{1}{\rho_Y} = \frac{\varepsilon_Y}{h/2} \quad (8)$$

Where  $\varepsilon_Y$  is the strain at yield. Also since the strain at  $y_Y$  is  $-\varepsilon_Y$ , we can write

$$\frac{1}{\rho} = \frac{\varepsilon_Y}{y_Y} \quad (9)$$

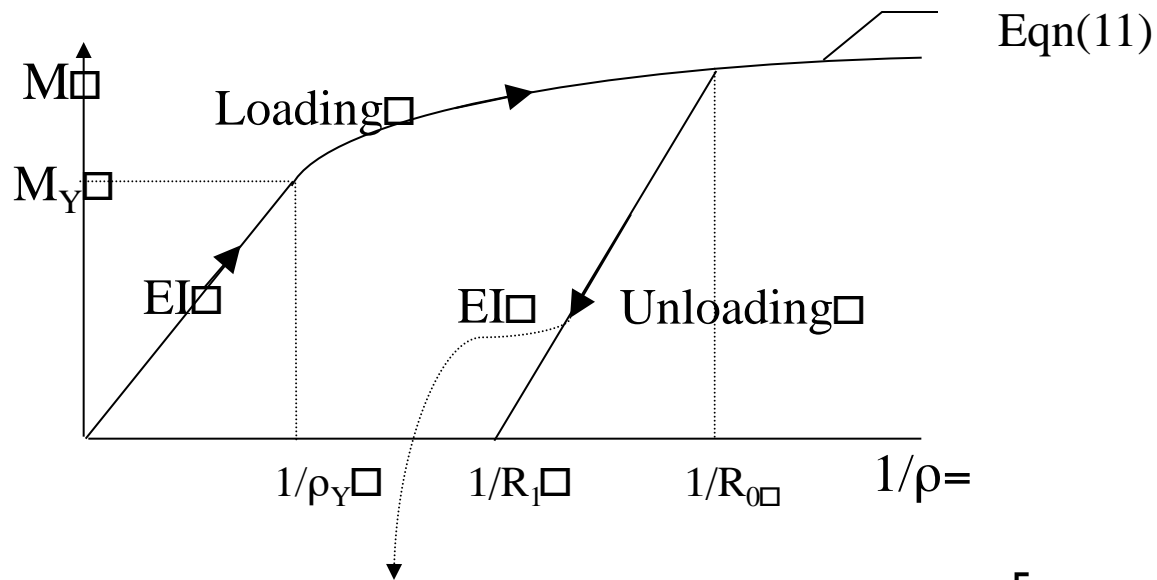
Combining (8) and (9) gives

$$\frac{y_Y}{h/2} = \frac{(1/\rho)_Y}{1/\rho} \quad (10)$$



Substitution into (7) gives the result we seek:

$$M = \frac{3}{2} M_Y \left[ 1 - \frac{1}{3} \left( \frac{(1/\rho)_Y}{1/\rho} \right)^2 \right] \quad (11)$$



Elastic unloading curve

$$M = \frac{M_Y}{(1/\rho)_Y} \left[ \frac{1}{\rho} - \frac{1}{R_1} \right] \quad (12)$$

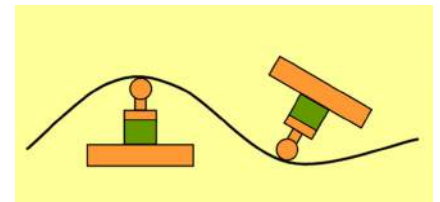
Now, eqn' s (11) and (12) intersect at  $1/\rho = 1/R_0$

Hence,

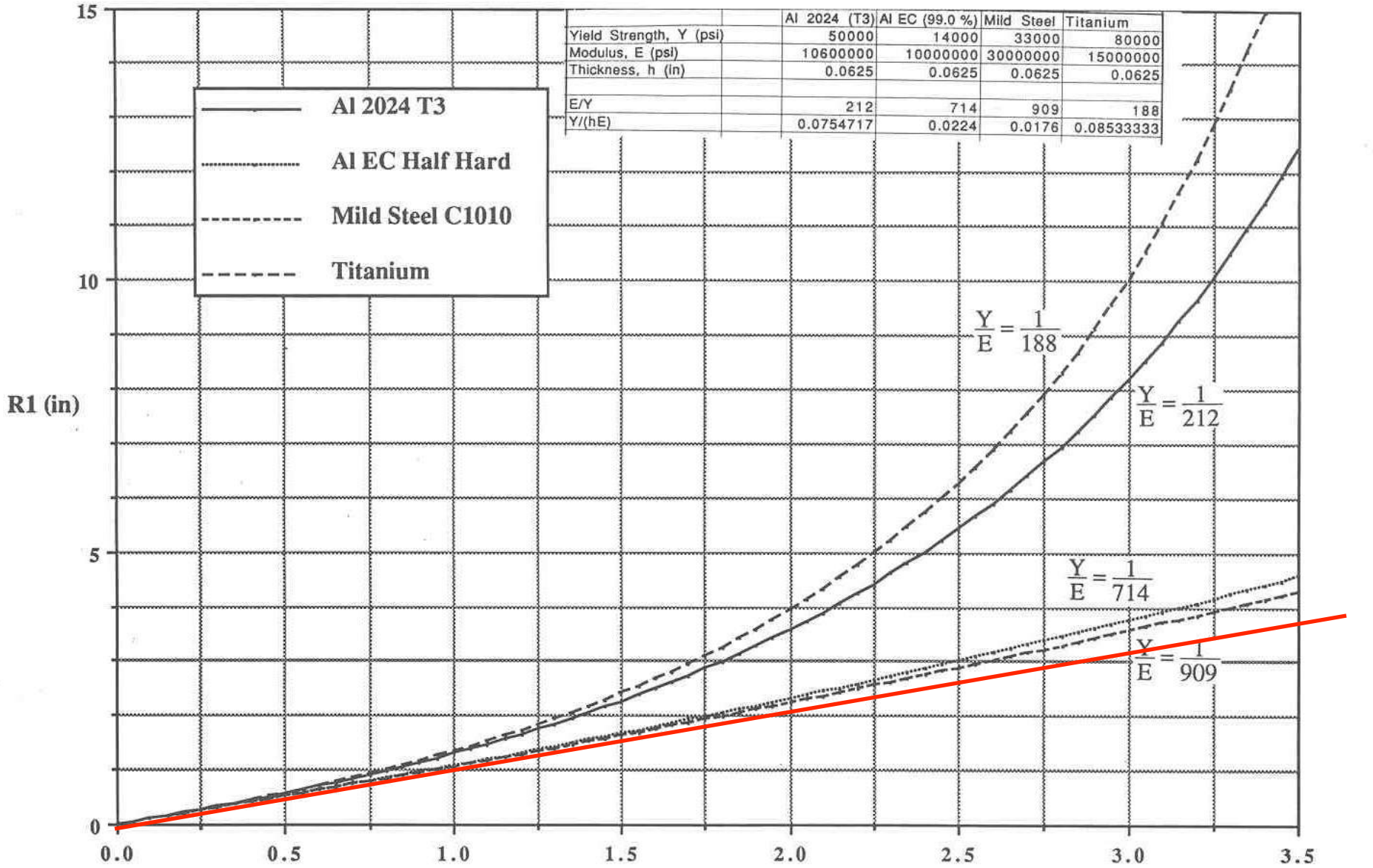
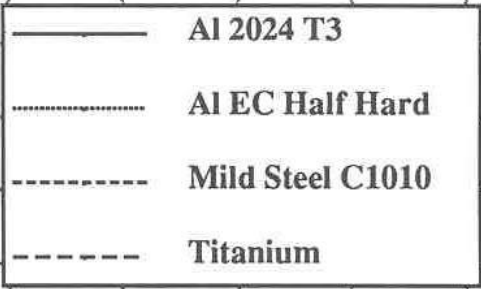
$$\frac{M_Y}{(1/\rho)_Y} \left[ \frac{1}{R_0} - \frac{1}{R_1} \right] = \frac{3}{2} M_Y \left[ 1 - \frac{1}{3} \left( \frac{(1/\rho)_Y}{1/R_0} \right)^2 \right]$$

Rewriting and using  $(1/\rho)_Y = 2Y/hE$  (from a few slides back), we get

$$\left[ \frac{1}{R_0} - \frac{1}{R_1} \right] = 3 \frac{Y}{hE} - 4R_0^2 \left( \frac{Y}{hE} \right)^3 \quad (13)$$



	Al 2024 (T3)	Al EC (99.0 %)	Mild Steel	Titanium
Yield Strength, Y (psi)	50000	14000	33000	80000
Modulus, E (psi)	10600000	10000000	30000000	15000000
Thickness, h (in)	0.0625	0.0625	0.0625	0.0625
E/Y	212	714	909	188
Y/(hE)	0.0754717	0.0224	0.0176	0.08533333



$$\frac{Y}{E} = \frac{1}{188}$$

$$\frac{Y}{E} = \frac{1}{212}$$

$$\frac{Y}{E} = \frac{1}{714}$$

$$\frac{Y}{E} = \frac{1}{909}$$

R0=R1

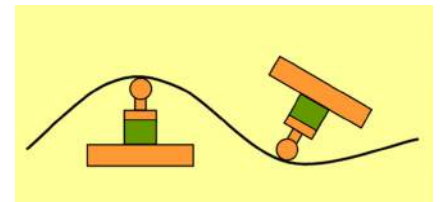
Ro (in)

Thickness, h = 0.0625 in.

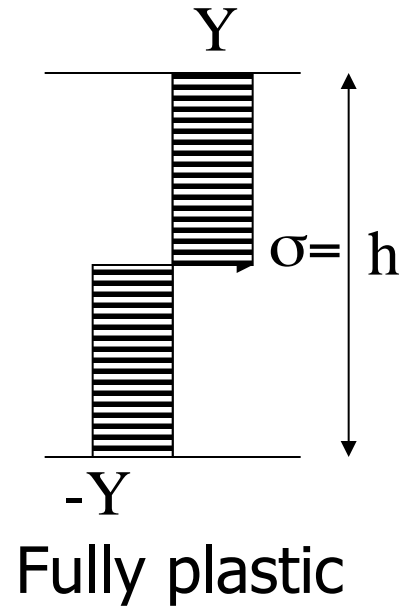
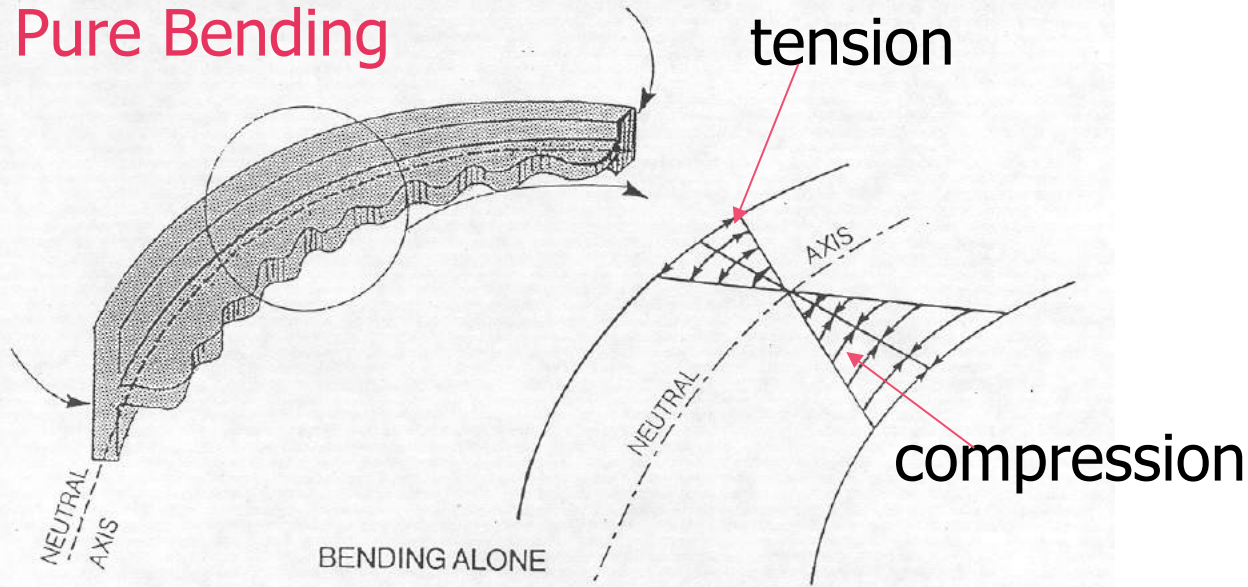
$$\frac{1}{R_0} - \frac{1}{R_1} = 3 \frac{Y}{hE} - 4R_0^2 \left( \frac{Y}{hE} \right)^3$$

# Methods to reduce springback

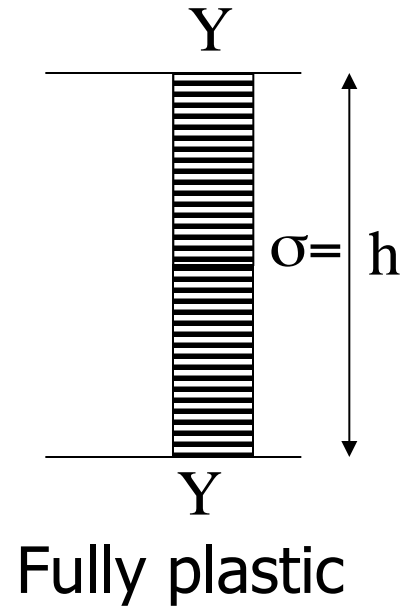
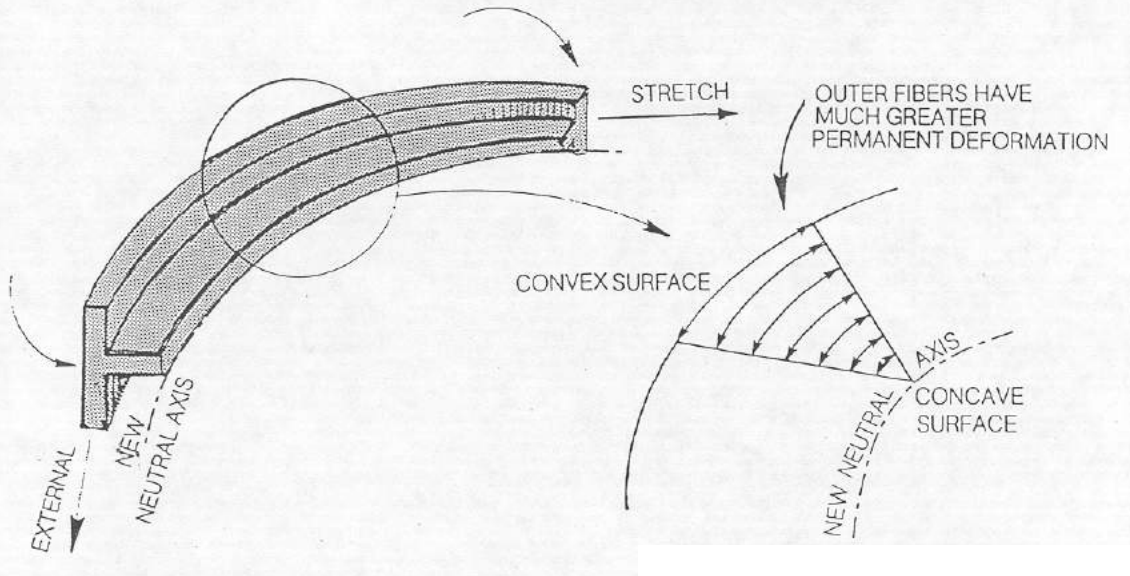
- Smaller Y/E
- Larger thickness
- Over-bending
- Stretch forming
- “coining” or bottoming the punch



# Pure Bending

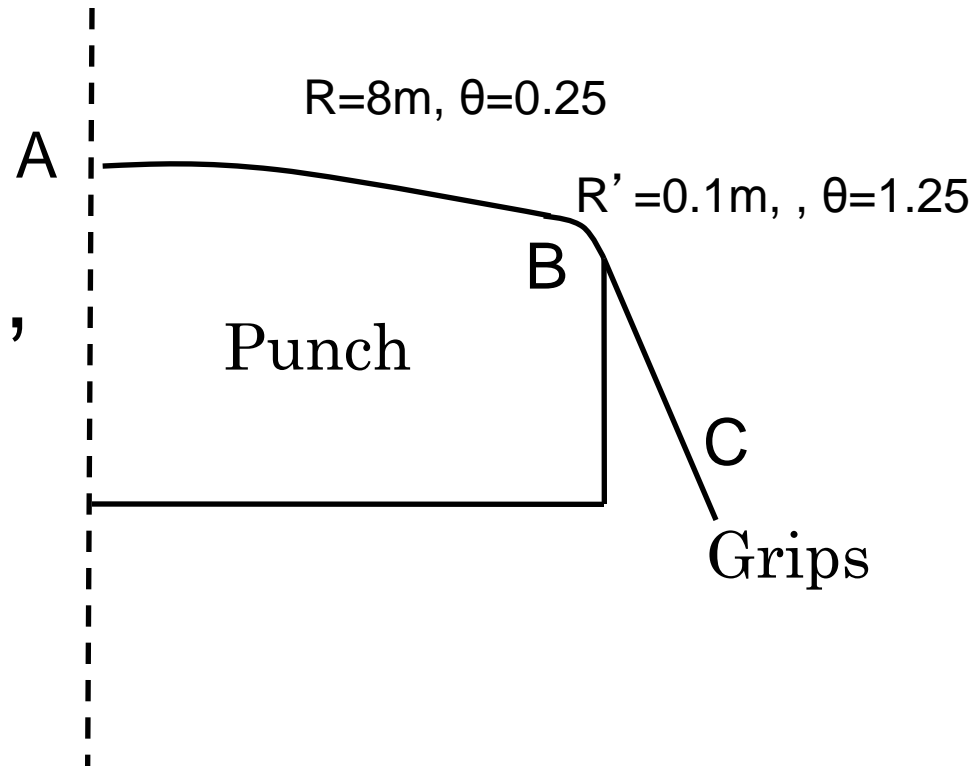


# Bending & Stretching

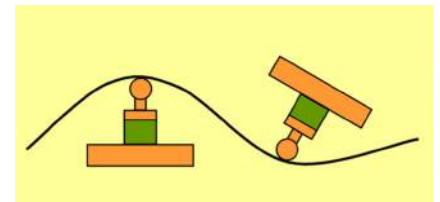


# Stretch forming: can we achieve a strain of 0.035 at A?

Sheet thickness 1mm,  
 $\mu=0.1$   
Material:  
 $\sigma=520\varepsilon^{0.18}\text{MPa}$



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# Can we achieve a strain of 0.035 at A?

Sheet thickness 1mm,

$\mu=0.1$

Material:  $\sigma=520\epsilon^{0.18}\text{MPa}$

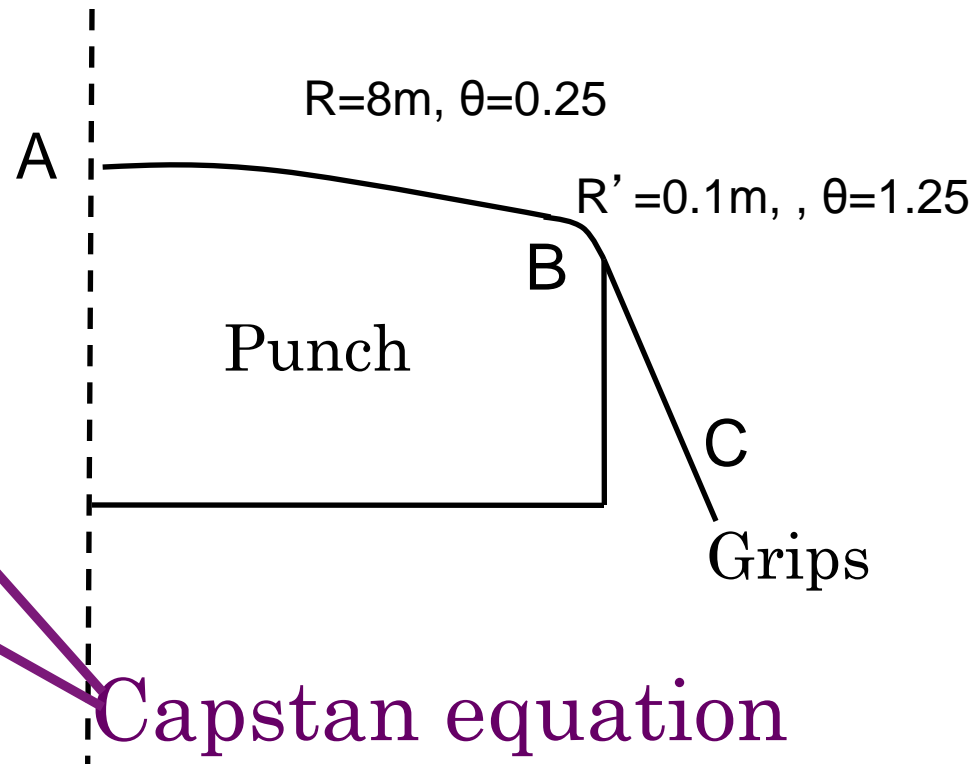
$$F_A = 0.001 * 520 * (0.035)^{0.18} = 284 \text{ kN/m}$$

$$F_B = F_A * \exp(0.1 * 0.25) = 292 \text{ kN/m}$$

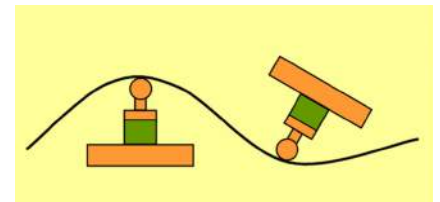
$$F_C = F_B * \exp(0.1 * 1.05) = \mathbf{323 \text{ kN/m}}$$

Max allowable force

$$= 0.001 * 520 * (0.18)^{0.18} = \mathbf{381 \text{ kN/m}}$$



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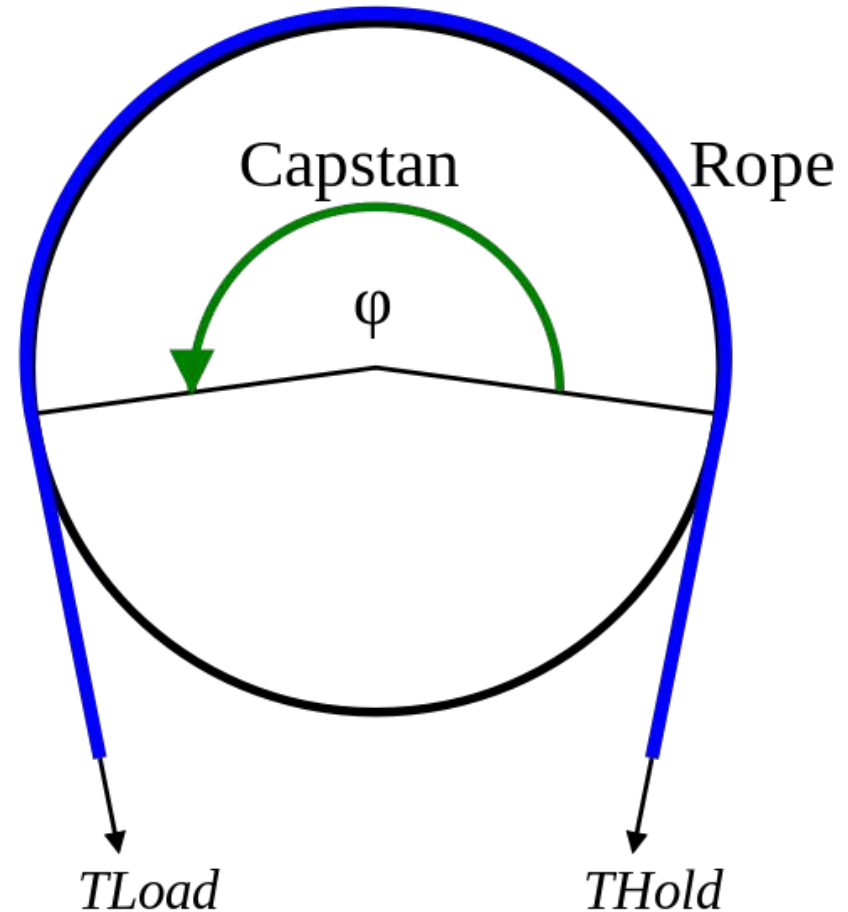


# Friction and the capstan equation

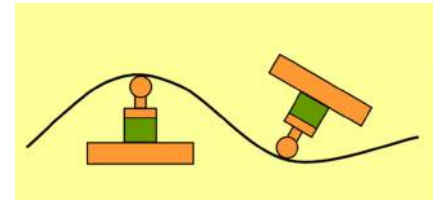
Typical stamping lubricants:

- Oil-based lubricants
- Aqueous lubricants
- Soaps and greases
- Solid films

$$T_{load} = T_{hold} \times \exp(\mu\theta)$$



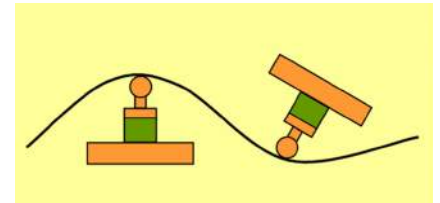
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# Research opportunities and challenges: reducing cost and environmental impacts



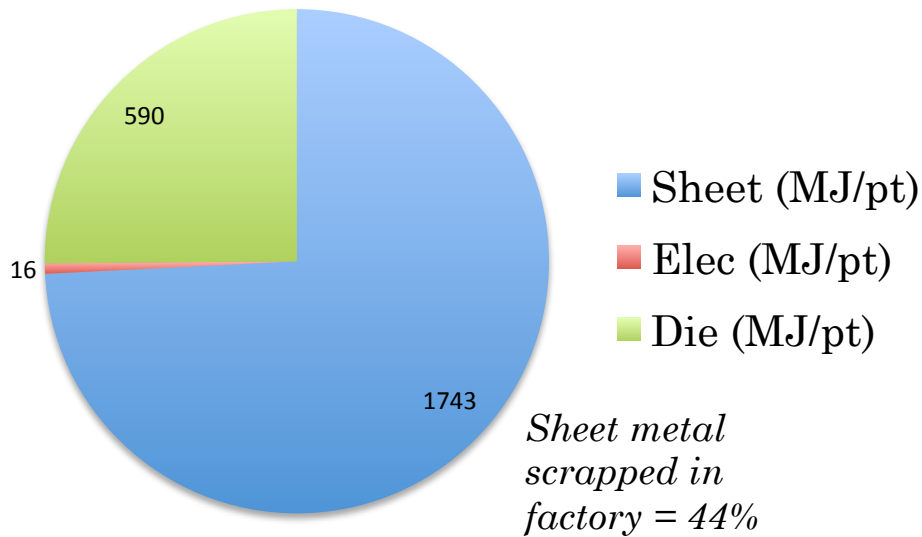
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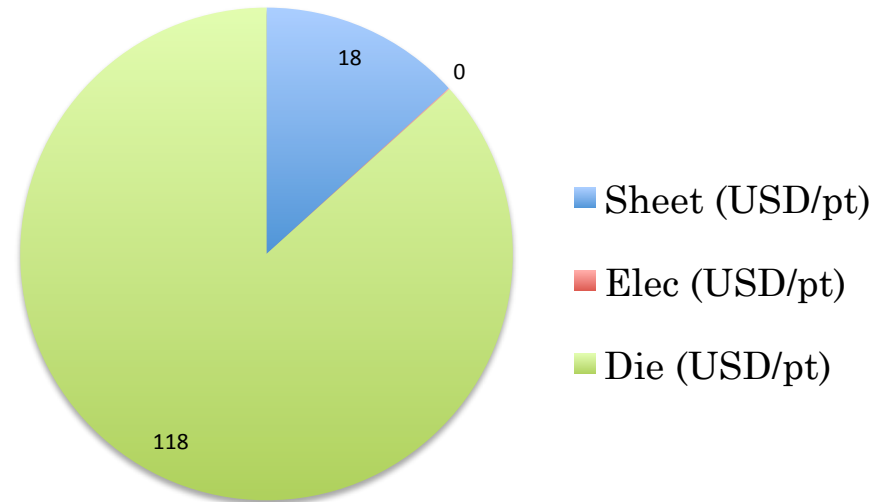
# Energy & cost: Stamping alum car hoods

- Final part = 5.4kgs
- Total number of parts made = 400
- Die material: cast and machined zinc alloy

Energy. 2.3GJ/pt. Stamping alum. car hoods. 5.4kgO/P. (400pts)



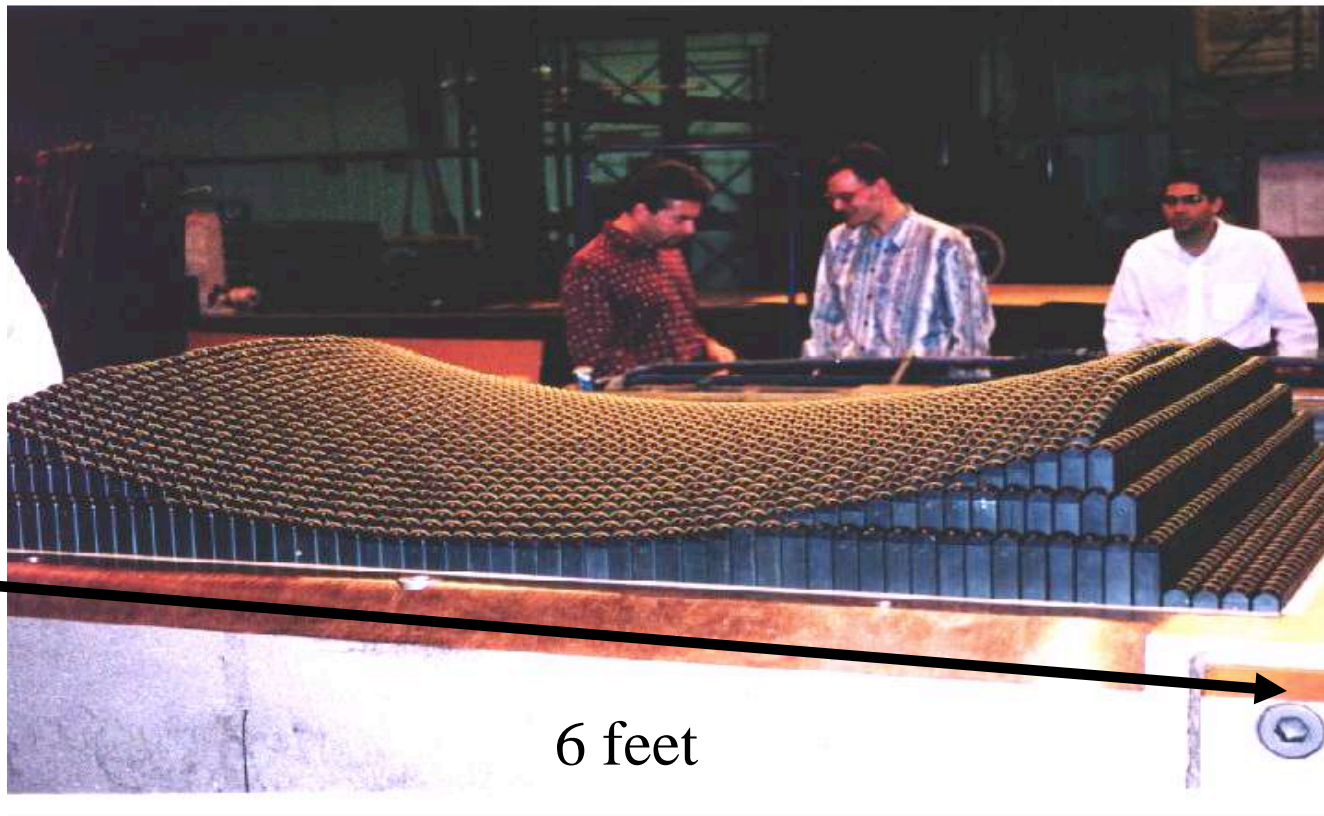
Cost. 136USD/pt



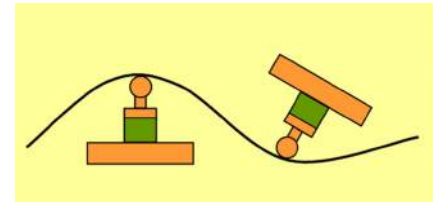
Source: Unpublished work: Cooper, Rossie, Gutowski (2015)

Excludes equipment depreciation and labor during forming

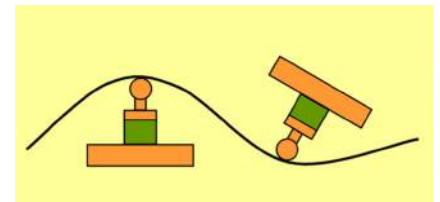
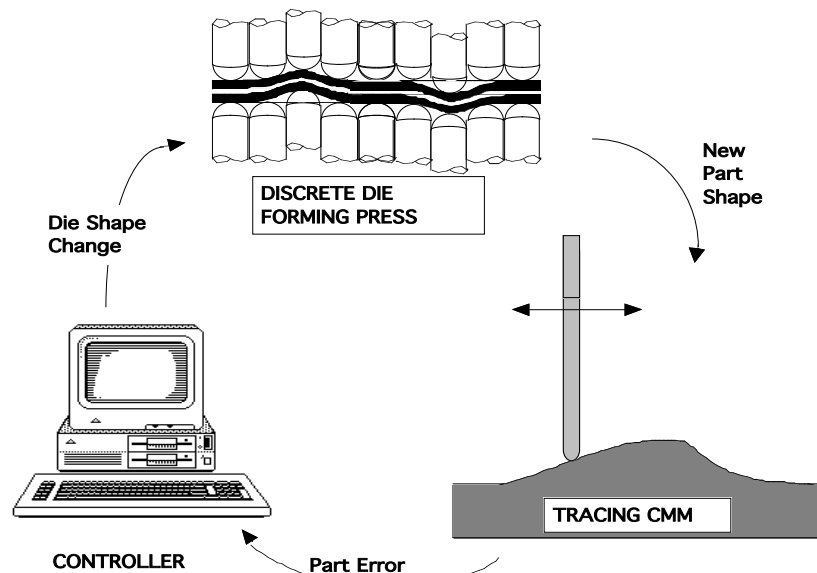
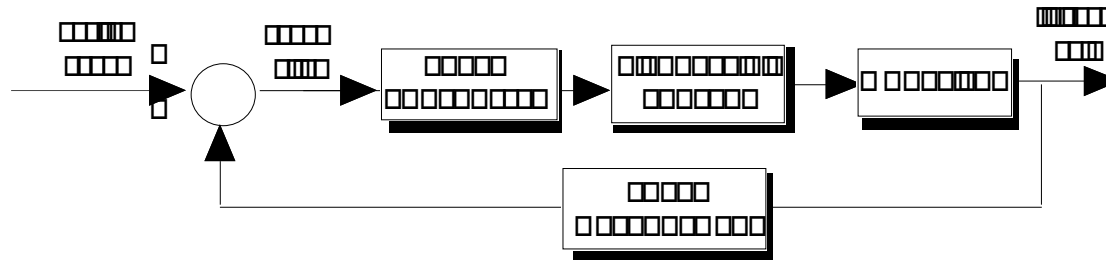
# 60 Ton Discrete Die Press (LMP - Hardt)



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# The Shape Control Concept

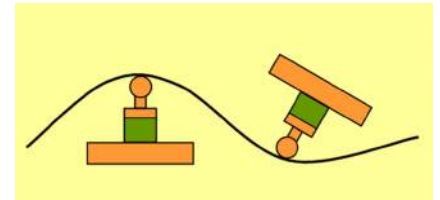




# Stretch Forming with Reconfigurable Tool @ Northrop Grumman



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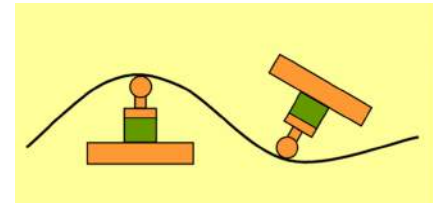




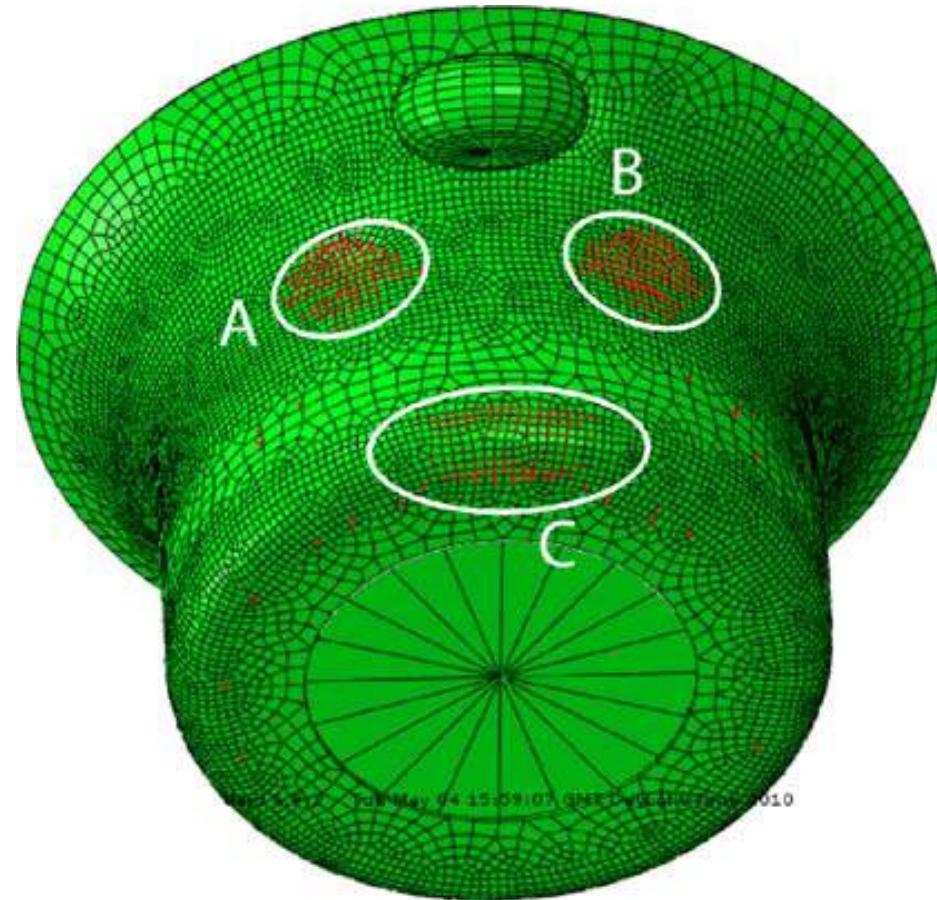
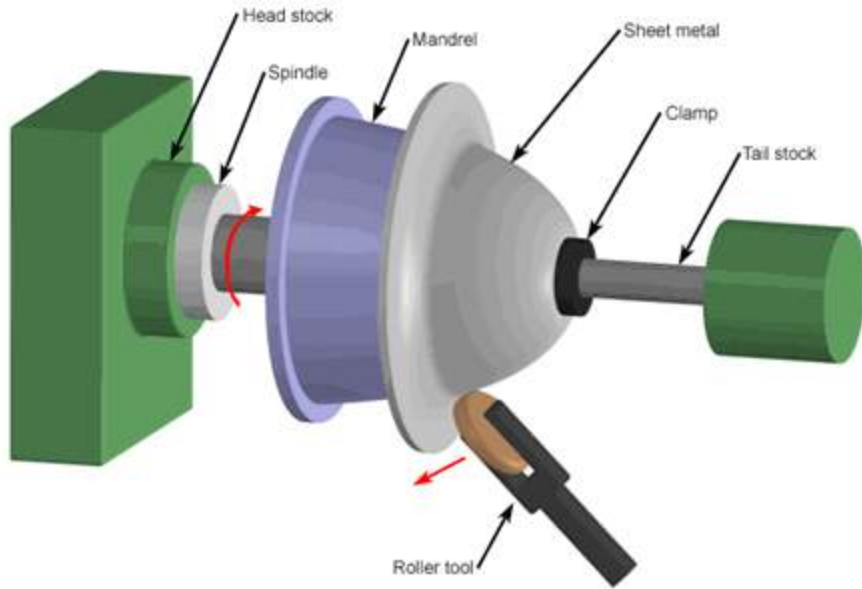
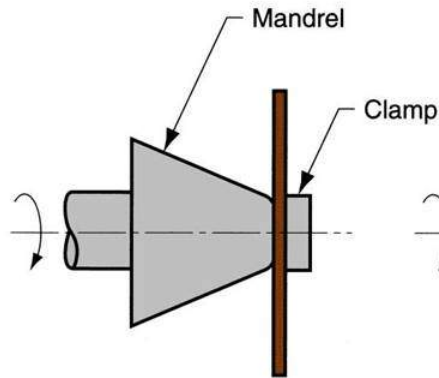
# Flexible Forming at Ford



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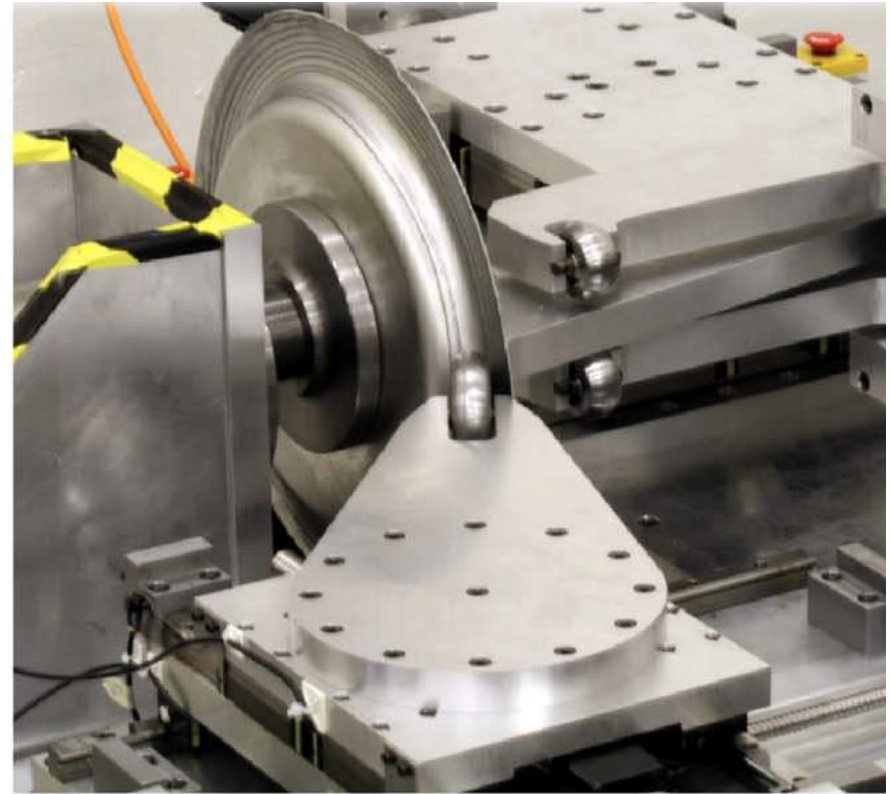
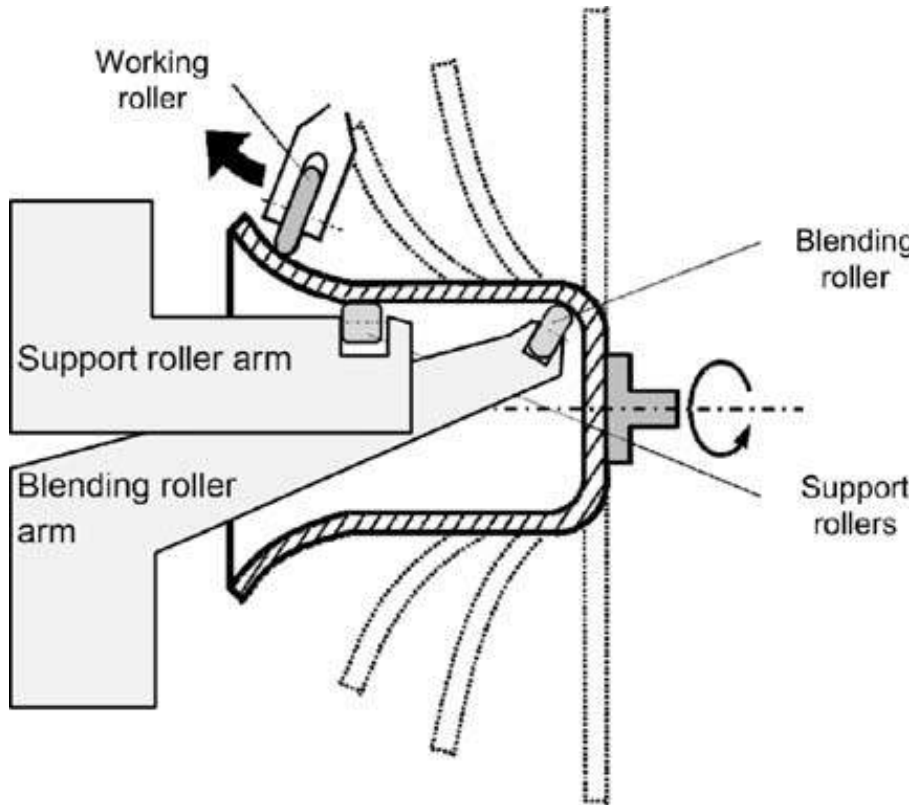


# Conventional Spinning



Copyright © 2009 CustomPartNet

# Flexible Spinning



**(b) Machine in operation**



Circular cup



Elliptical cup



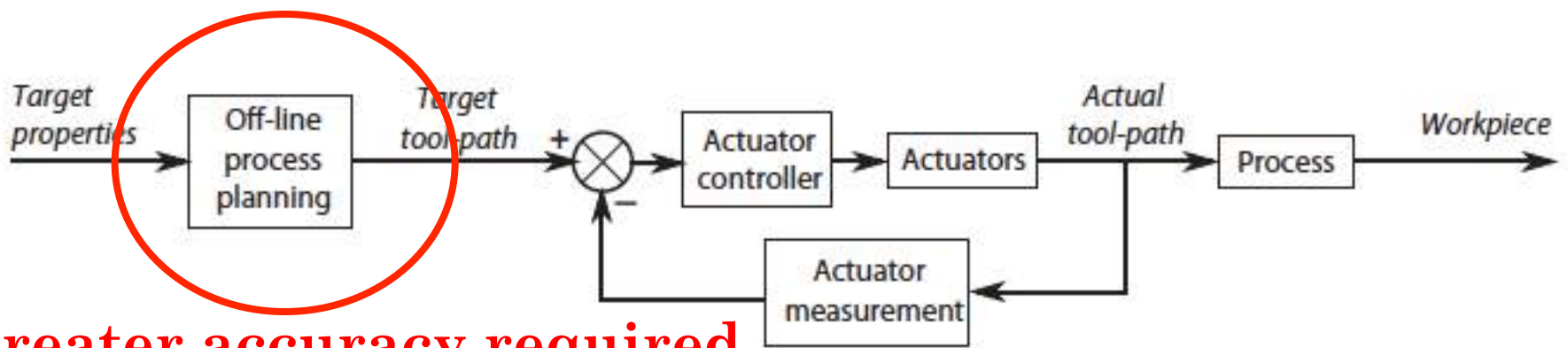
Rectangular cup



Kidney bean

Music, O., & Allwood, J. M. (2011). Flexible asymmetric spinning. *CIRP Annals - Manufacturing Technology*, 60(1), 319–322. doi:10.1016/j.cirp.2011.03.136





**Greater accuracy required**

Fig. 1. A system diagram for open-loop control of metal forming.

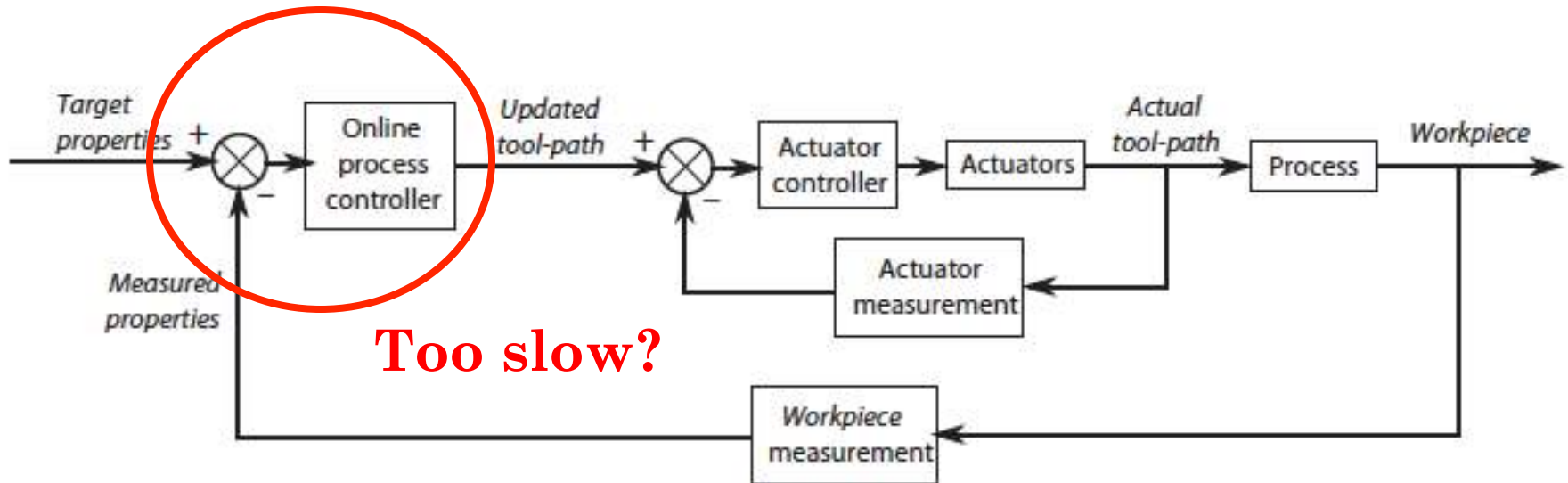
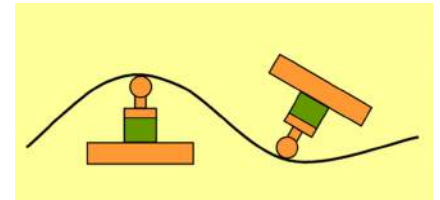


Fig. 2. A system diagram for closed-loop control of metal forming.



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Polyblank, J. a., Allwood, J. M., & Duncan, S. R. (2014). Closed-loop control of product properties in metal forming: A review and prospectus. *Journal of Materials Processing Technology*, 214(11), 2333–2348. doi:10.1016/j.jmatprotec.2014.04.014



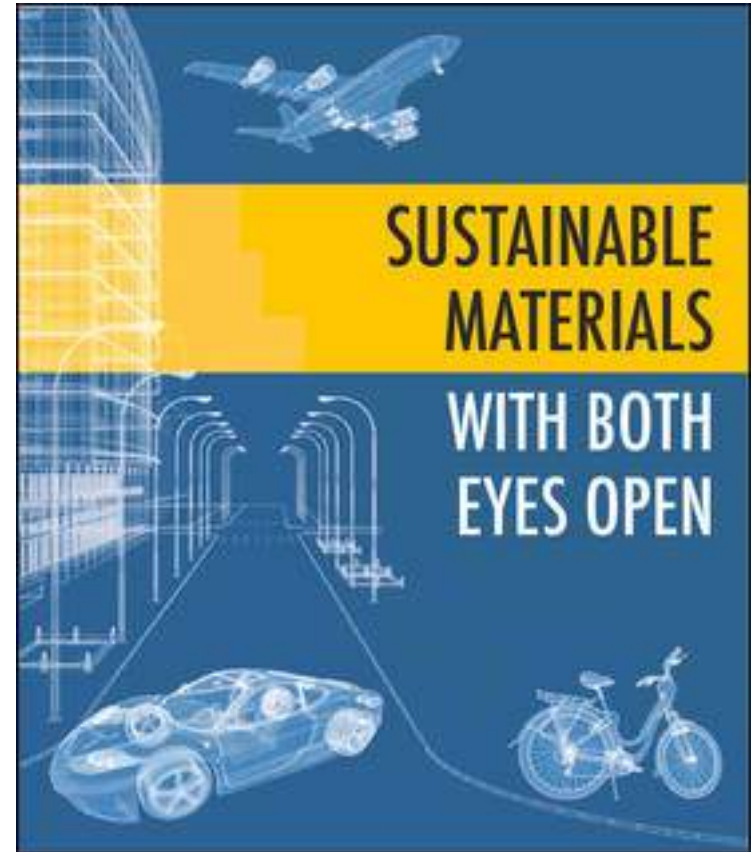
# Thank you

## Sheet metal forming in a low carbon future?

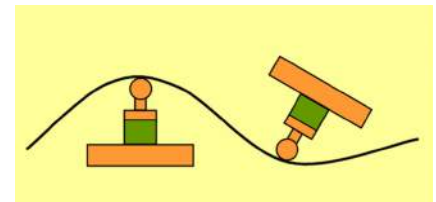
See the wonderful...

[http://  
www.withbotheyeyesopen.com](http://www.withbotheyeyesopen.com)

Allwood, J., Cullen, J., Carruth, M., Cooper, D., McBrien, M., Milford, R., ... Patel, A. (2012). *Sustainable Materials with Both Eyes Open*. Cambridge: UIT.



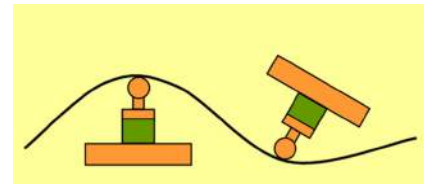
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# Extra slides – just for fun



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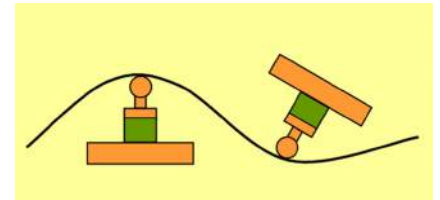
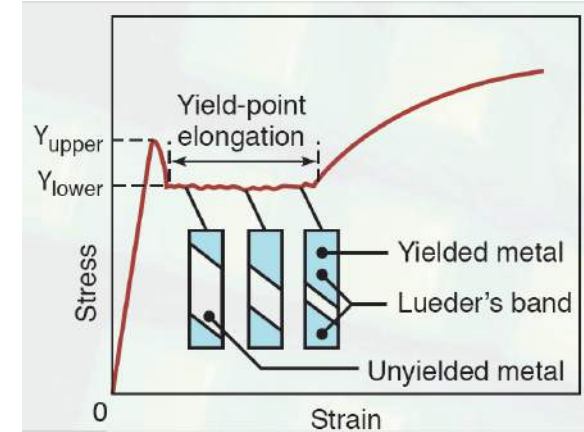


# Surface finish defects

- Orange peel effect

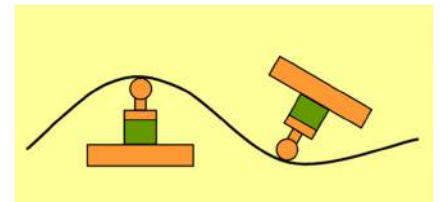
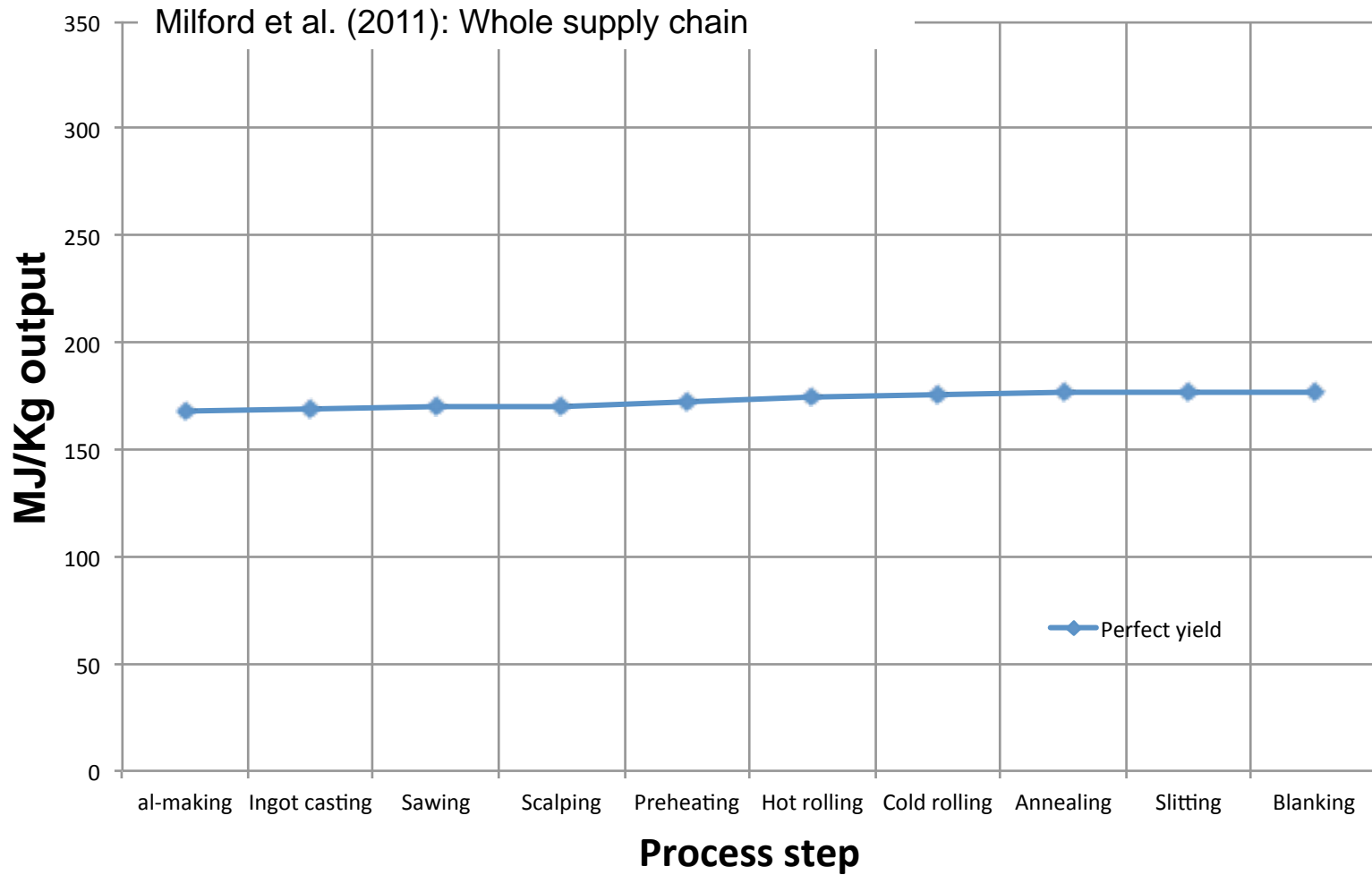


- Lüders bands

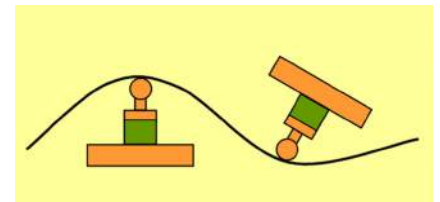
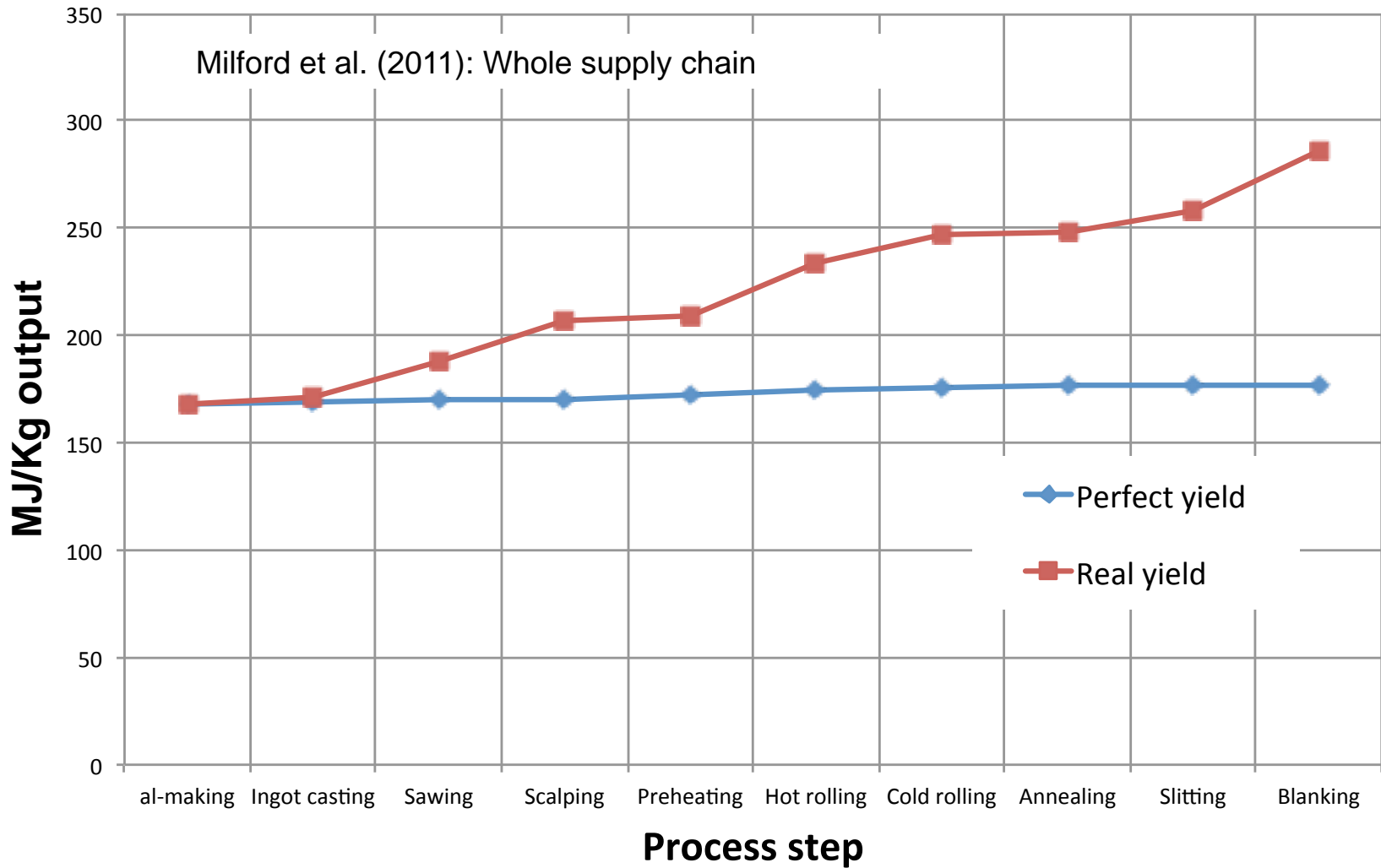




# Material embodied energy: Aluminum primary production



# Material embodied energy: Aluminum primary production



# Material embodied energy: Aluminum primary production

